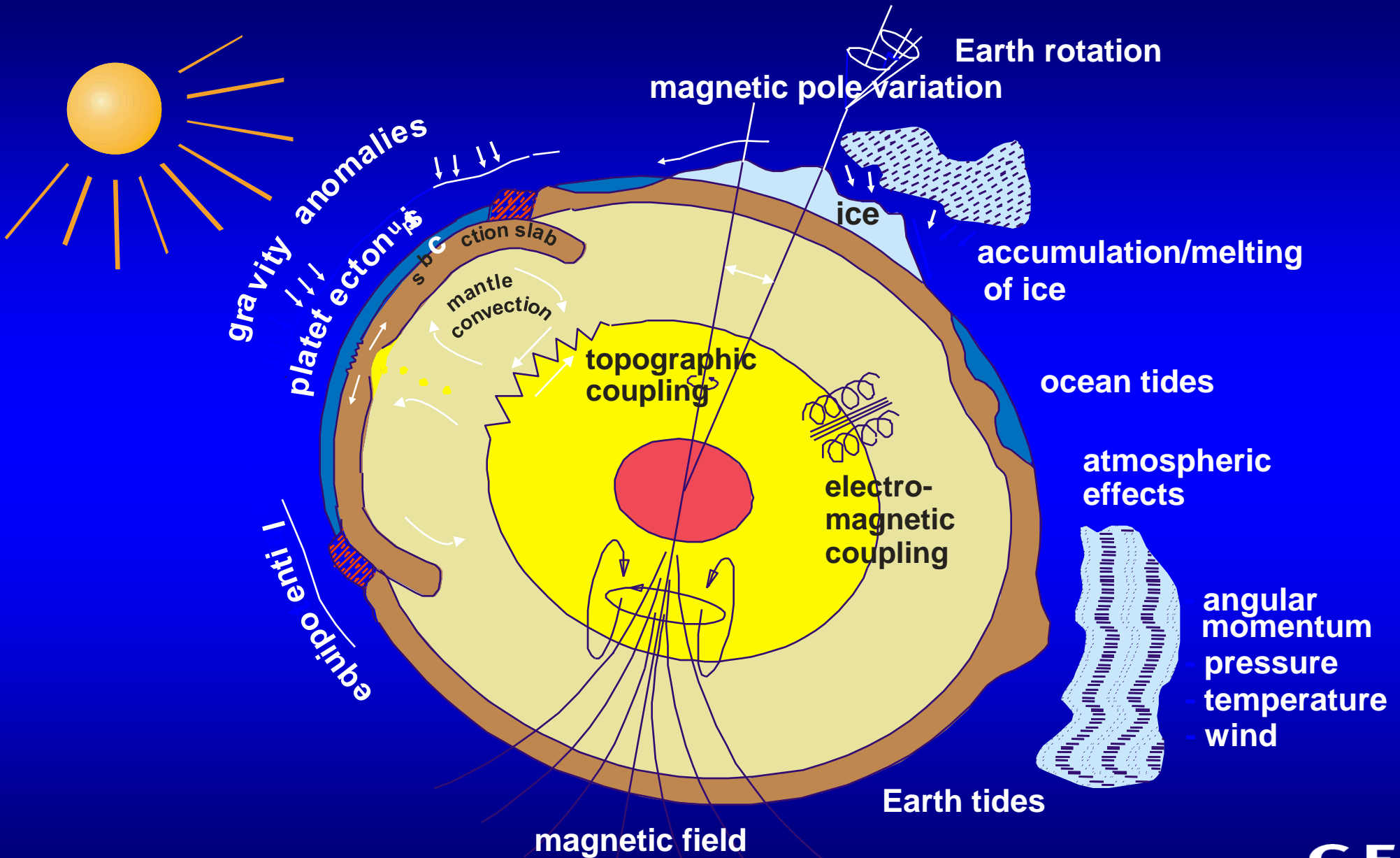


# ***The CHAMP-, GRACE- and GOCE- Geopotential Missions - and their contributions to environmental monitoring -***

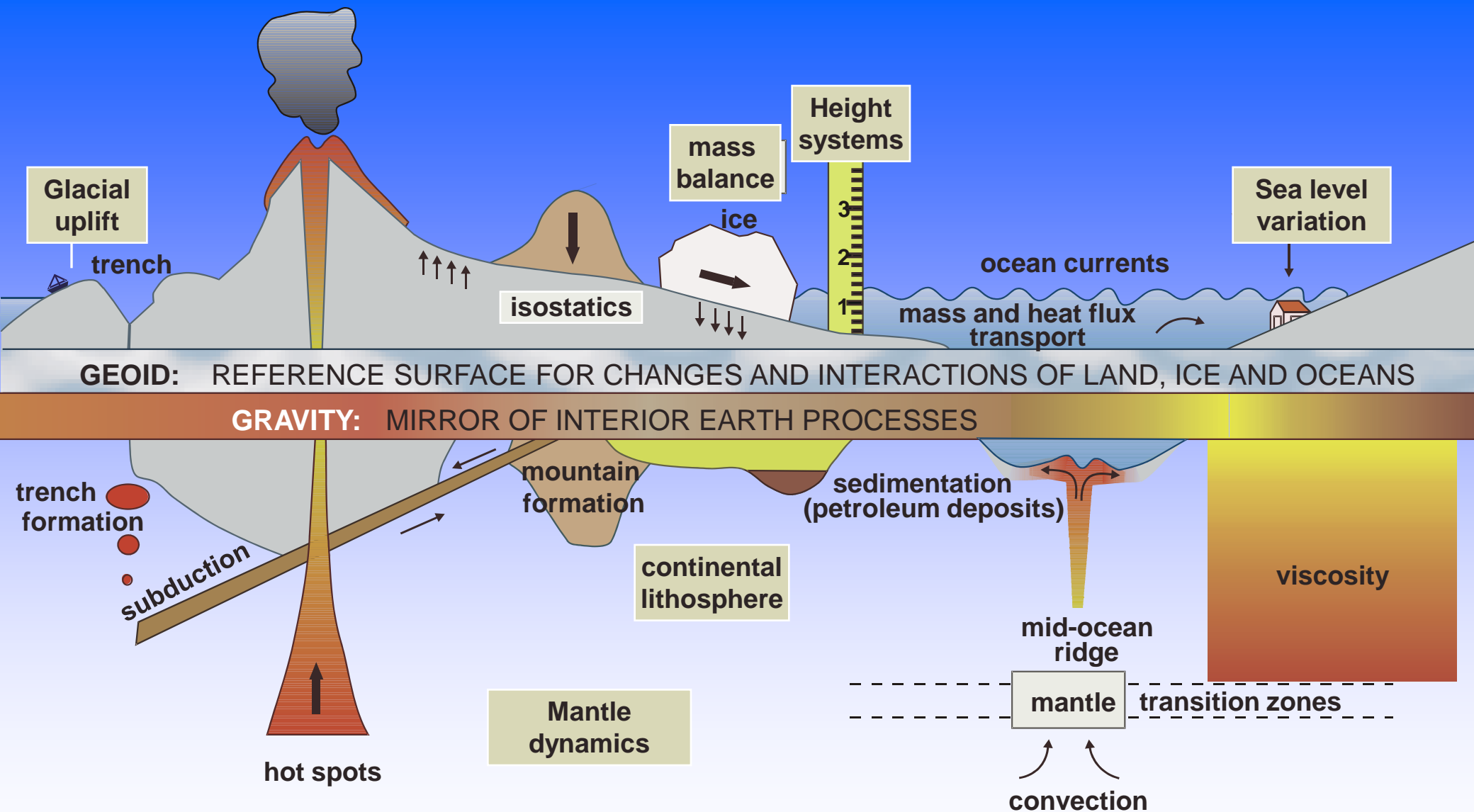
**Christoph Reigber <sup>1,2</sup> Frank Flechtner <sup>1</sup> Roman Galas <sup>3</sup>**

- 1 German Research Center for Geosciences GFZ**
- 2 University Potsdam**
- 3 Technical University Berlin**

# Geosystem: The Changing Earth



# Dual Role of Earth Gravity



*Various forces effect mass redistributions. Gravity variations are correlated to these redistributions & are thus important impact factors.*

# Gravity Field Representation & Functionals

$$T = \frac{GM}{R} \sum_{l=2}^L \sum_{m=0}^l \left( \frac{R}{r} \right)^{l+1} P_{lm}(\cos \vartheta) \left[ \Delta \hat{C}_{lm} \cos m\lambda + \Delta \hat{S}_{lm} \sin m\lambda \right]$$

RESIDUAL  
GRAVITATIONAL  
POTENTIAL  
in m<sup>2</sup>/s<sup>2</sup>

100 km resolution requires  $L = 20000 \text{ km} / 100 \text{ km} = 200$

**RESOLUTION**

$$N = R \sum_{l=2}^L \sum_{m=0}^l P_{lm}(\cos \vartheta) \left[ \Delta \hat{C}_{lm} \cos m\lambda + \Delta \hat{S}_{lm} \sin m\lambda \right]$$

GEOID HEIGHTS  
in meter

$$\Delta g = \gamma \sum_{l=2}^L (l-1) \sum_{m=0}^l P_{lm}(\cos \vartheta) \left[ \Delta \hat{C}_{lm} \cos m\lambda + \Delta \hat{S}_{lm} \sin m\lambda \right]$$

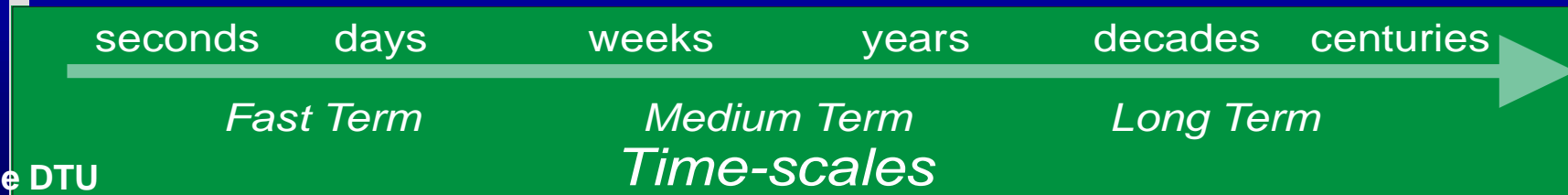
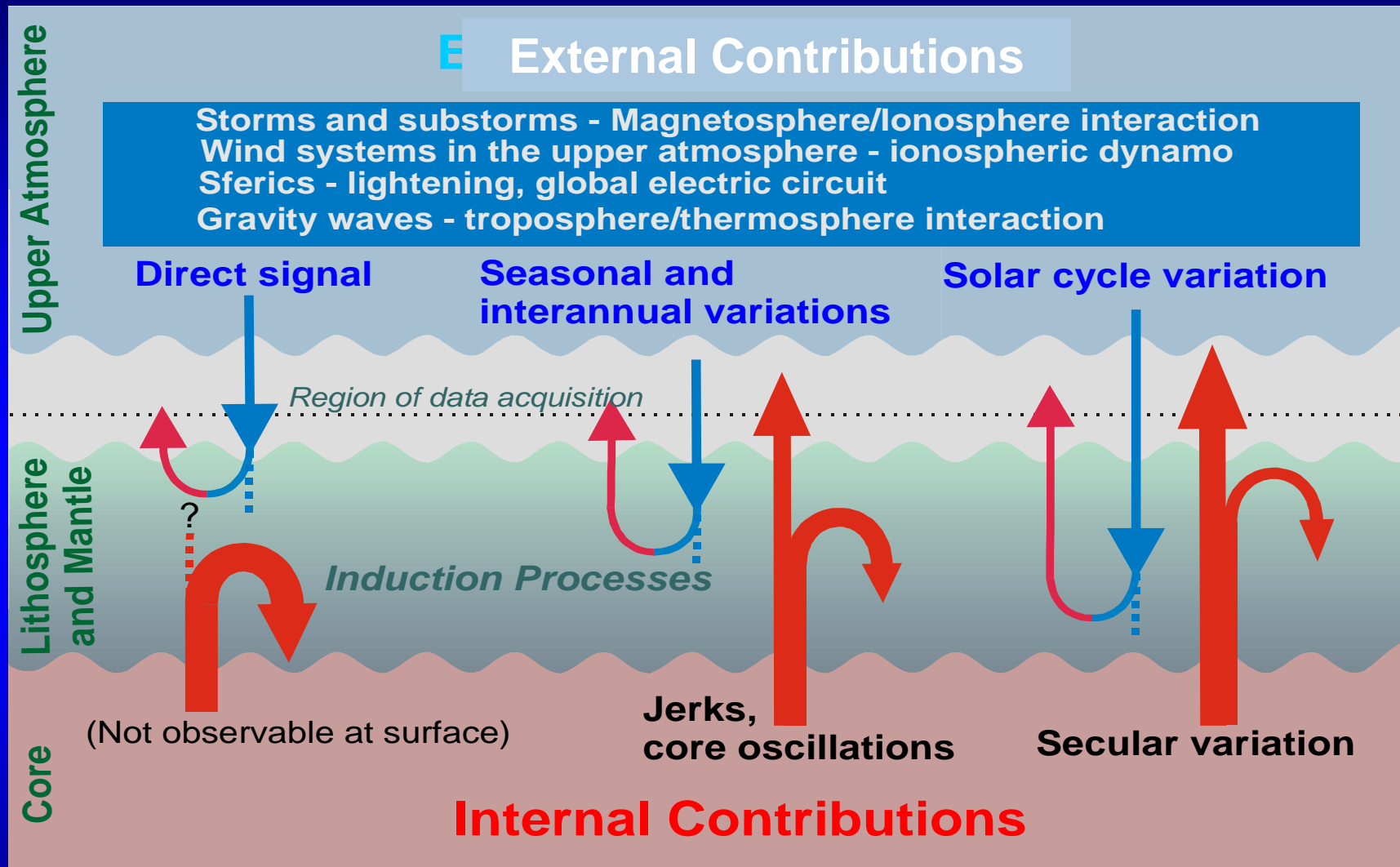
FREE AIR GRAVITY  
ANOMALIES  
in mgal ( 10<sup>-5</sup> m/s<sup>2</sup>)

$$c_l = \sum_{m=0}^l \left[ C_{lm}^2 + S_{lm}^2 \right]$$

Power Spectral Density (PSD)  
Average Signal Strength



# Dual Role of Magnetic Field



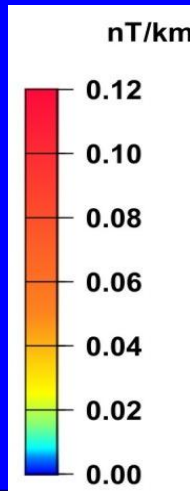
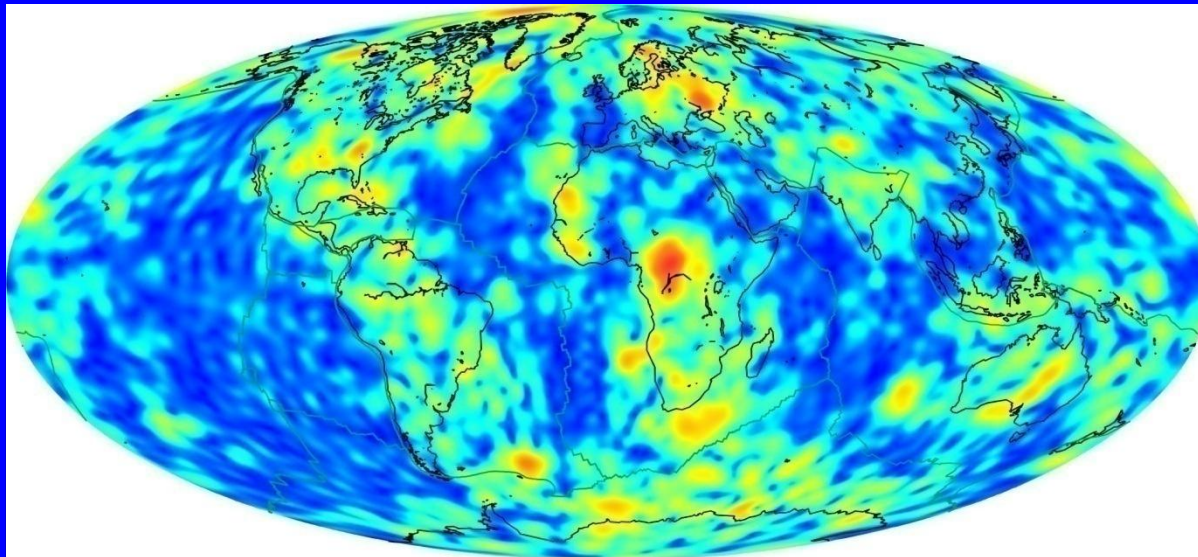
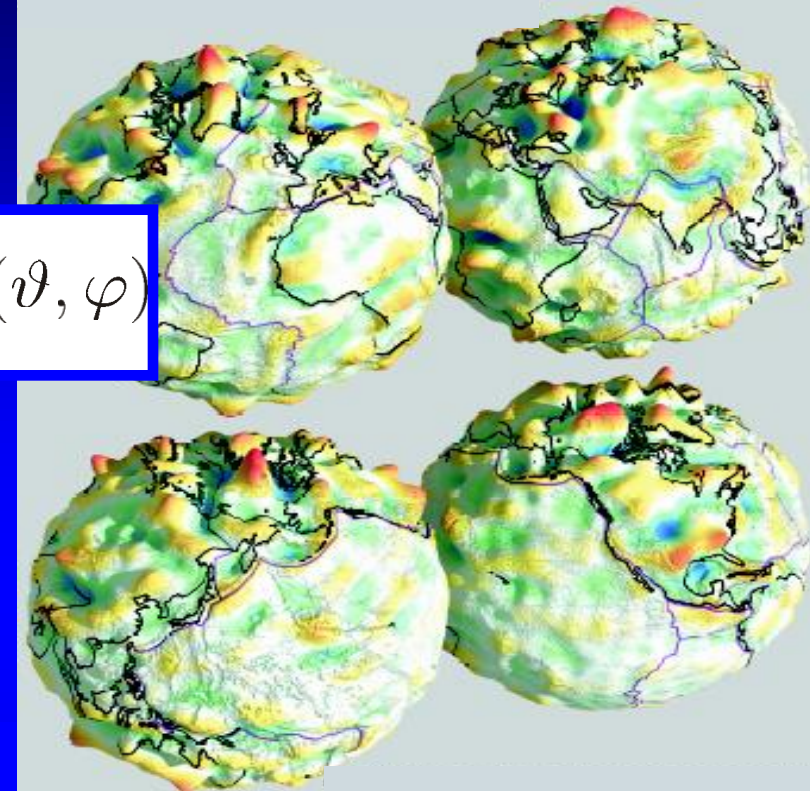
# Magnetics Represented by Crustal Magnetic Anomalies

$$\Psi(r, \vartheta, \varphi) =$$

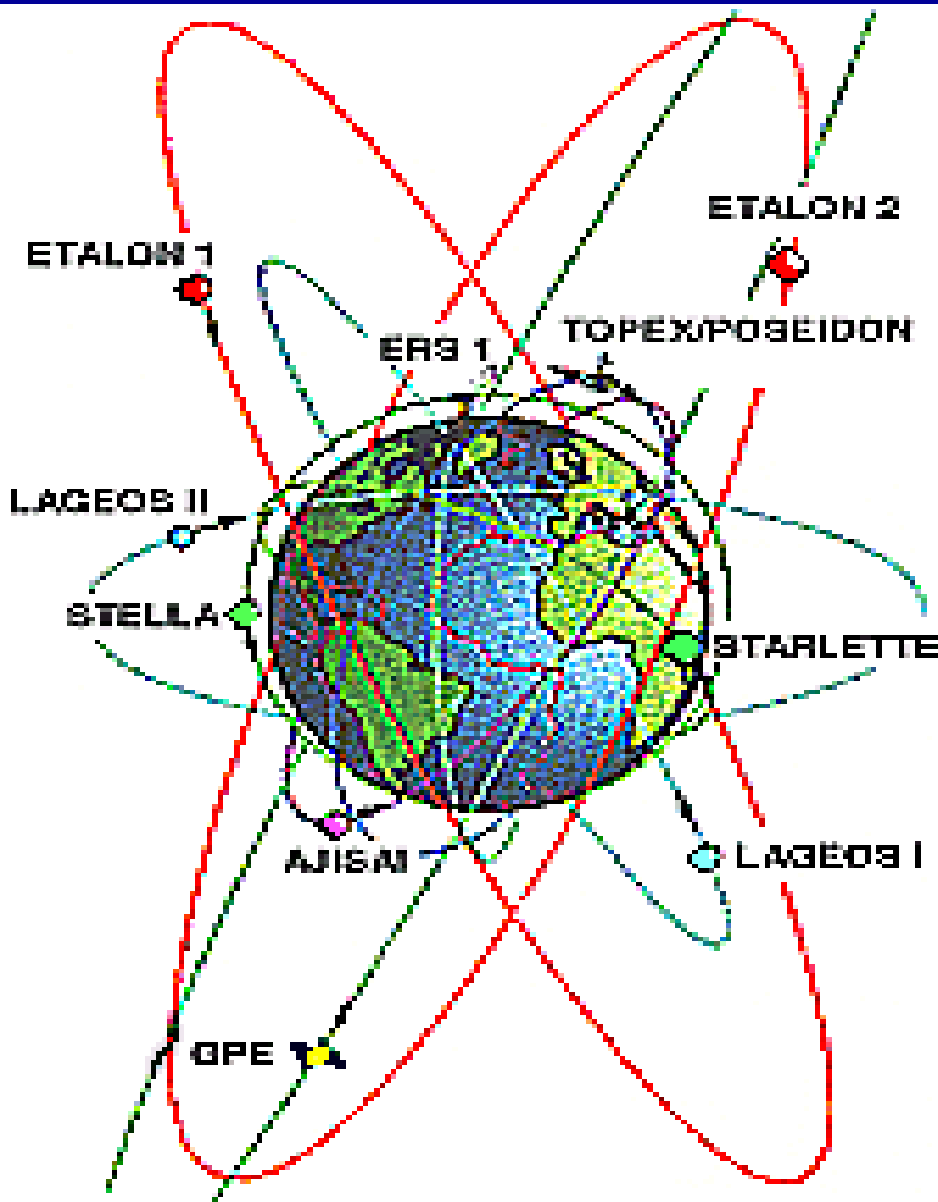
$$a \sum_{\ell=1}^{\infty} \sum_{m=-\ell}^{\ell} \left[ g_{\ell}^m \left( \frac{a}{r} \right)^{\ell+1} + k_{\ell}^m \left( \frac{r}{a} \right)^{\ell} \right] Y_{\ell}^m(\vartheta, \varphi)$$

internal  
sources

external  
sources



# Gravity Field Determination through Analysis of Satellite Orbit Perturbations



## Classical Approach:

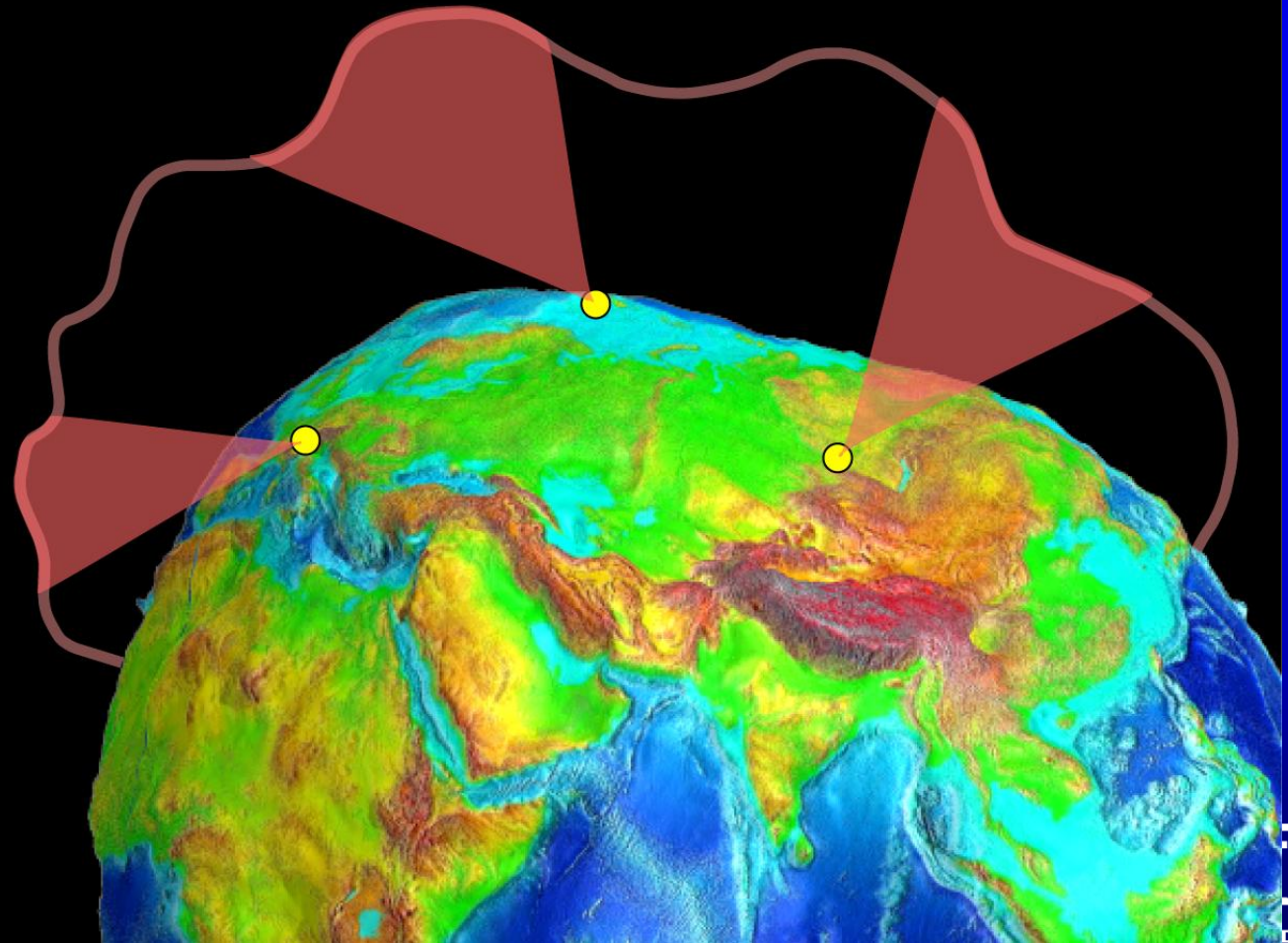
- Ground tracking of orbital segments of various near Earth satellites over many years
- Solution of motion equations for all these objects and common adjustment of orbit residuals
- Modelling of surface forces, such as air drag, solar radiation pressure etc.



# ***Typical Ground-Based Coverage***

*(before GPS Space Receiver became available)*

- One dimension at a time
- Large coverage gaps
- Very critical for LEOs





# ***DOI Process for Gravity Field Recovery***

$$\Delta r_i^T = [\Delta x_i, \Delta y_i, \Delta z_i] = f(\Delta C_{lm}, \Delta S_{lm})$$

Pseudo-observations

$$A = \left\{ \frac{\partial (\Delta x_i, \Delta y_i, \Delta z_i)}{\partial (\Delta C_{lm}, \Delta S_{lm})} \right\}$$

Design matrix from partials

$$\{\Delta x_i, \Delta y_i, \Delta z_i\}$$

Observation residuals

**LSA**

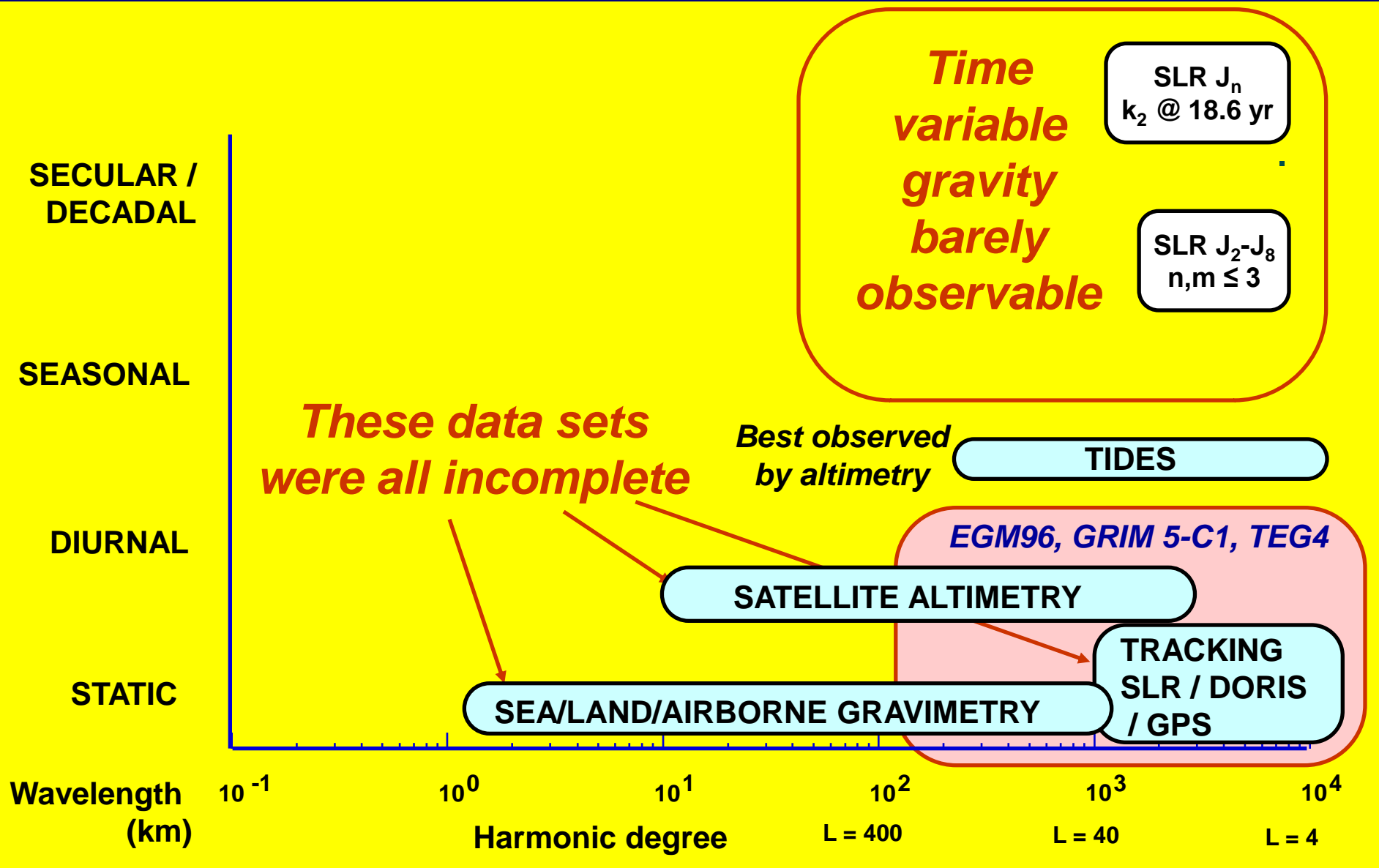
$$\{\Delta \hat{C}_{lm}, \Delta \hat{S}_{lm}\}$$

Res. harmonic coeff.

# ***History of Satellite Gravity Field Modelling***

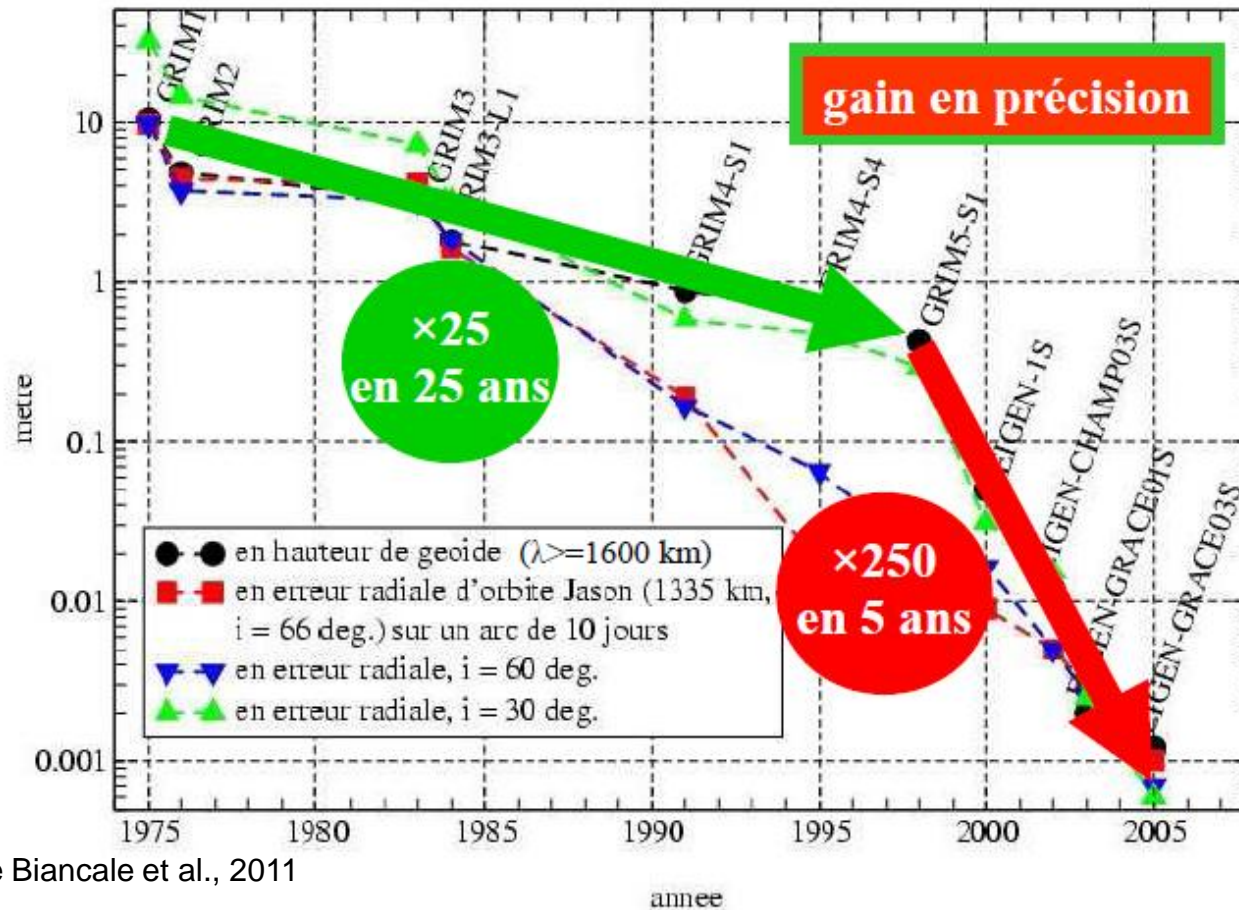
<b>Model</b>	<b>Institution</b>	<b>Period</b>
<b><i>SE I-III</i></b>	Smithonian Astrophysical Observatory	1966 - 1973
<b><i>GEM 1-10/T1-3 EGM 96</i></b>	Goddard Space Flight Center/ Ohio State University	1972 - 1996
<b><i>GRIM 1-5</i></b>	DGFI/GFZ Potsdam/GRGS Toulouse	1976 - 1998
<b><i>TEG 1-4</i></b>	CSR /UTEX Austin	1988 - 1999

# Situation Gravity Field Determination in 2000



# GFZ/GRGS gravity model improvements over time (GRIM and EIGEN models)

(par différence au modèle EIGEN-GL04S)



Source Biancale et al., 2011



# New Observation Techniques

## Satellite-to-Satellite Tracking

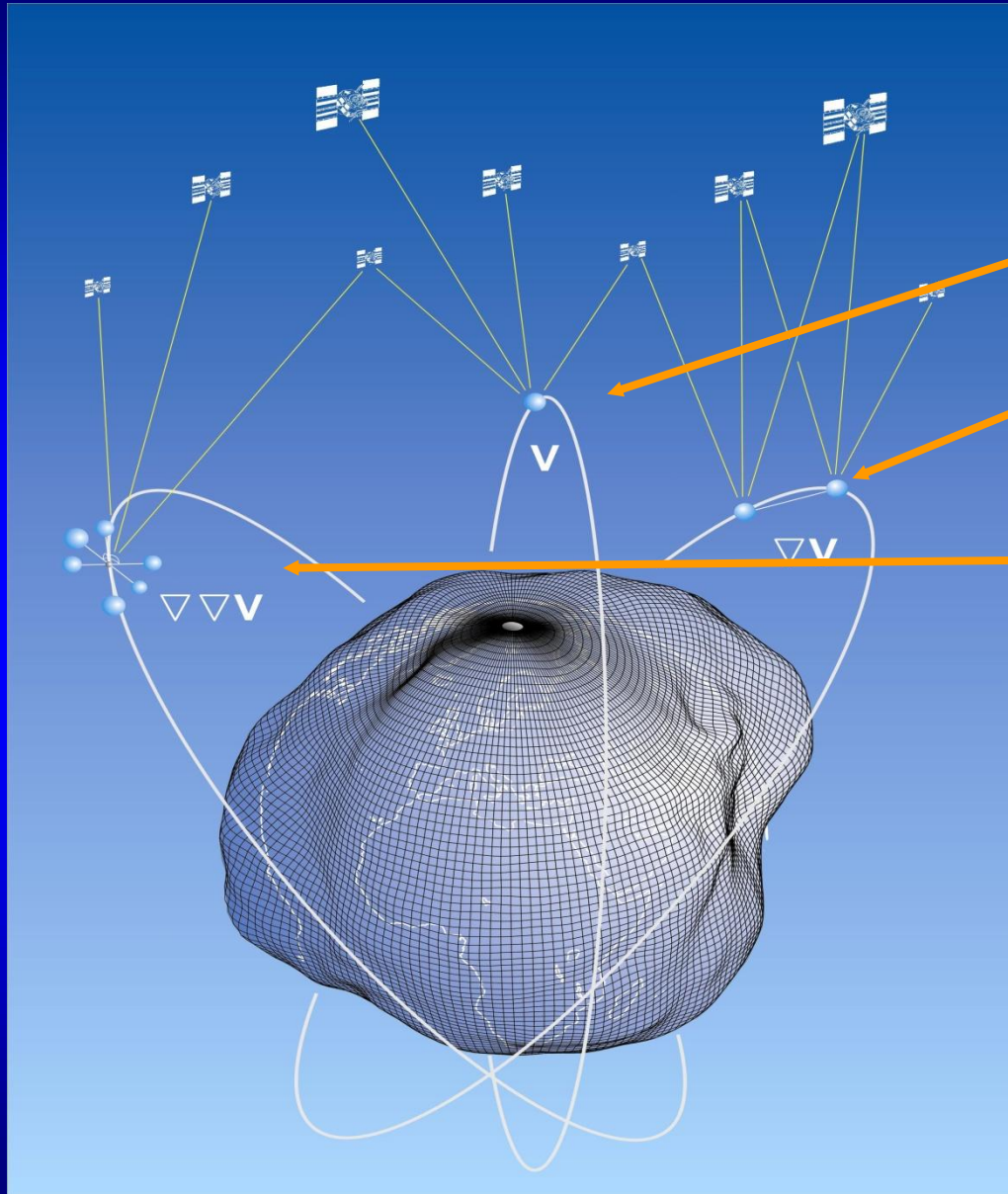
◆ *high- low*

Wolff, 1969

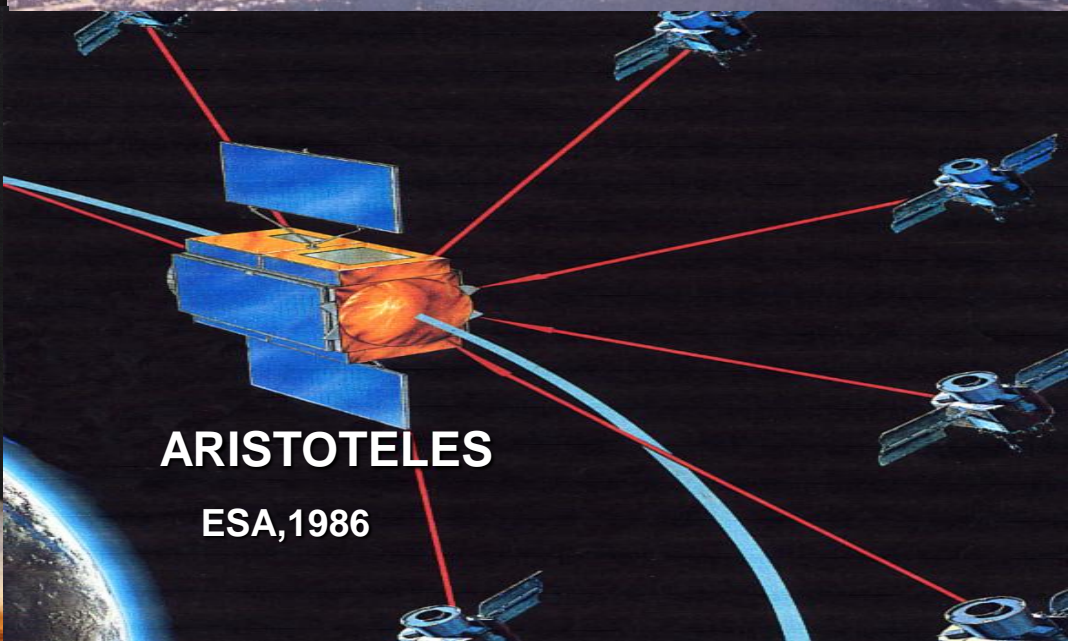
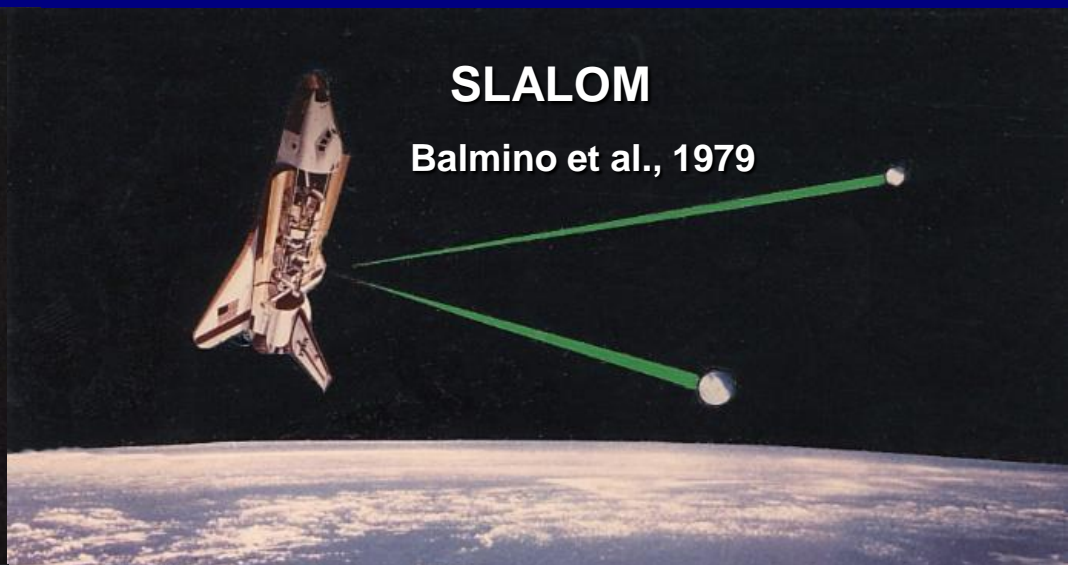
◆ *low- low*

## Satellite Gravity Gradiometry

**Williamstown 1969, NASA**  
**Elmau 1979, ESA**  
**Erice 1985, IAG**  
**Coolfont 1991, NASA**

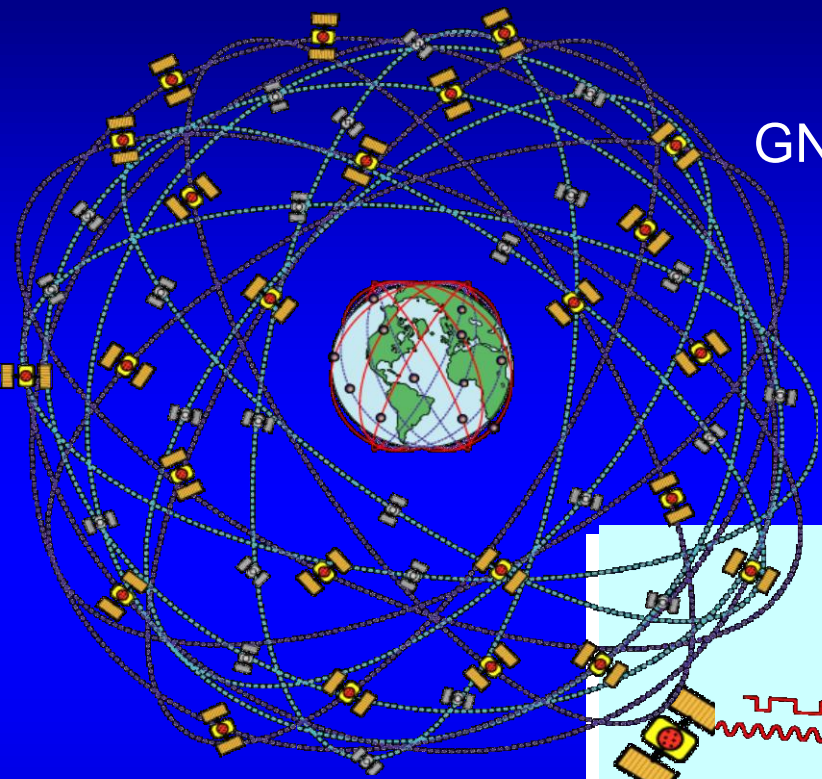


# Early SST& SGG Mission Proposals



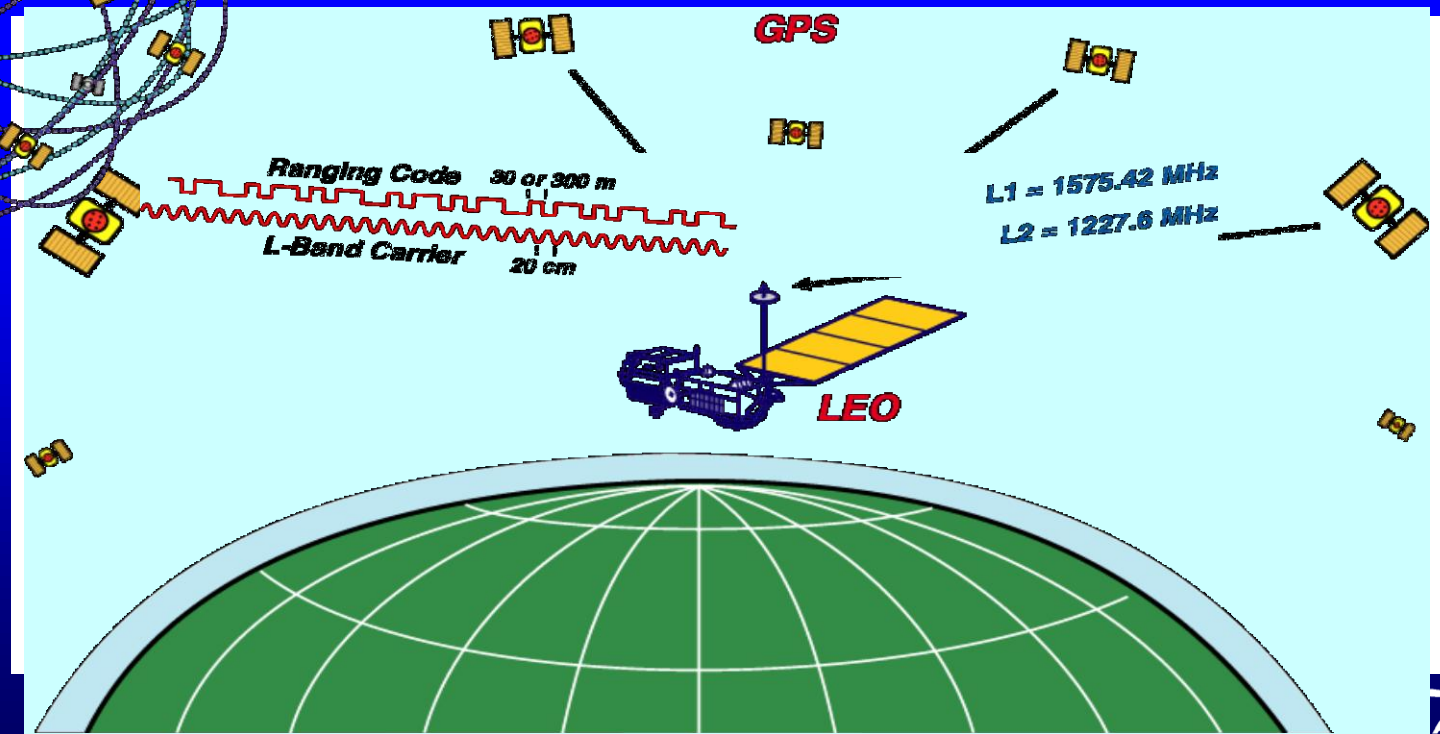


# Global Navigation Satellite System (GNSS)



GNSS constellation

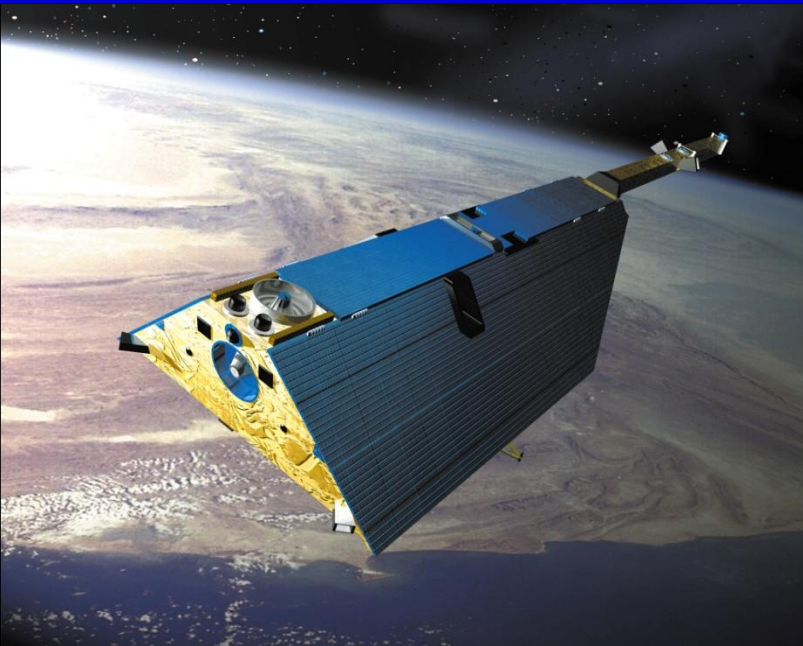
GPS Code & Trägerphasen-  
messungen im L-Band



# *The Breaking New Missions ....*

CHAMP (2000-2010)

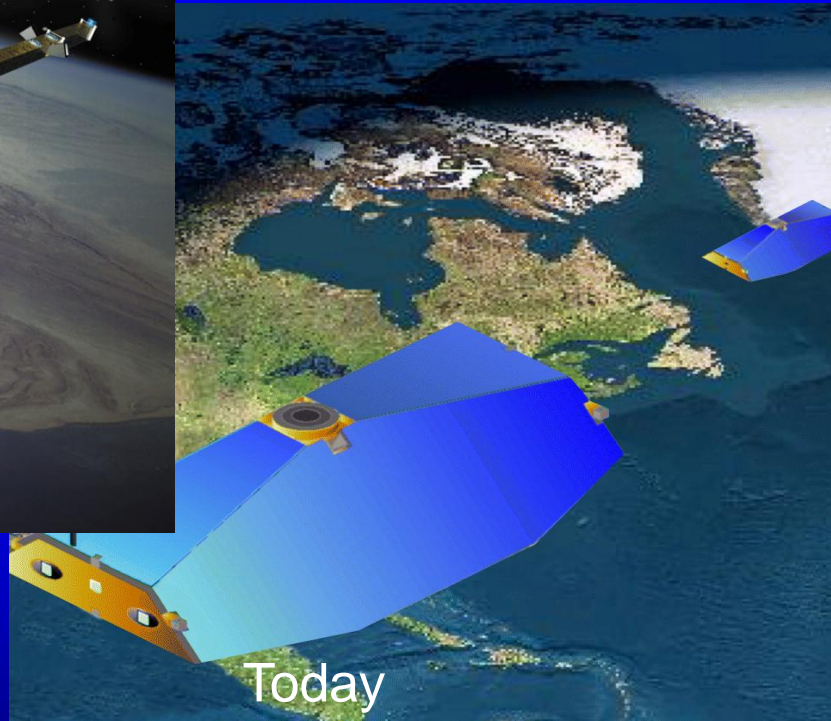
Germany



EoL: Sep.19, 2010  
3717 days in orbit  
58277 revolutions

GRACE (2002)

USA-Germany

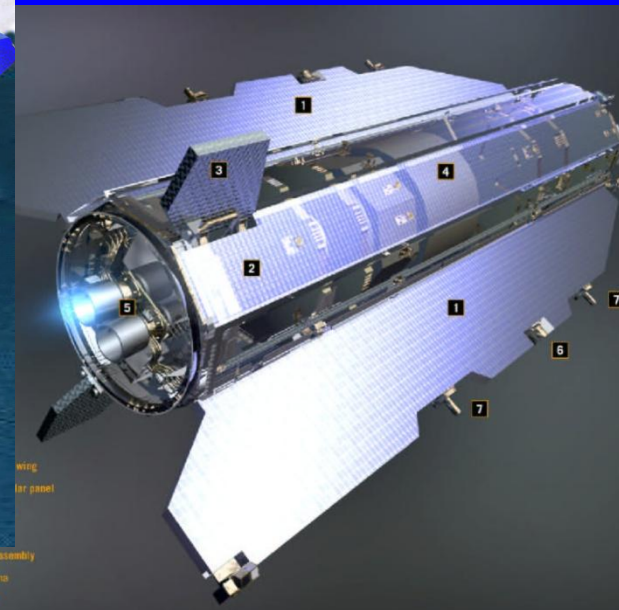


Today

3364 days in orbit  
51506 revolutions

GOCE (2009)

European



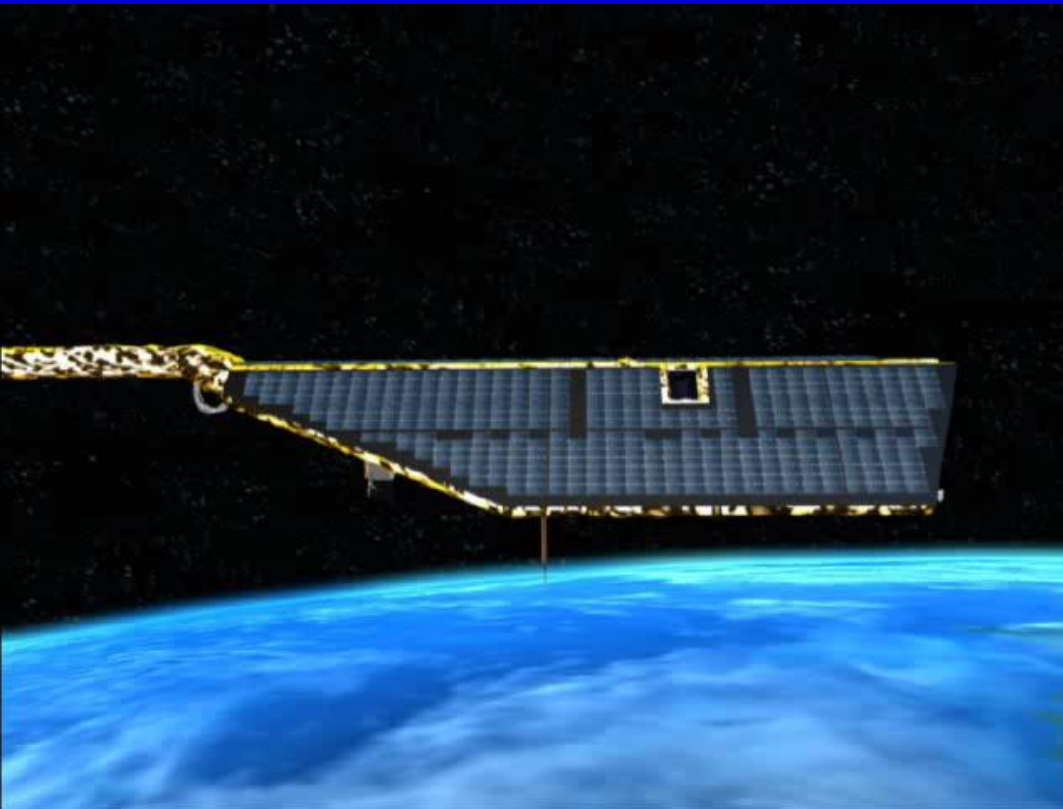


# CHAMP

Mission of GFZ Potsdam & DLR (Germany)  
with contributions of NASA (USA), CNES (France) and AFRL (USA)

**Launch:** Juli 2000      **Planned Mission duration:** 5 years

**Orbit:** nearly circular polar orbit ( $87^\circ$ ), 454 km altitude



## Mission goals:

- ◆ Gravity field mapping: medium resolution static & time-variable components
- ◆ Magnetic field mapping: main & crustal field and time-variable components
- ◆ Atmosphere & ionosphere sounding

## Payload:

- ◆ 3-D accelerometer ( $3 \times 10^{-9} \text{ m/s}^2$ ).
- ◆ 16-channel GPS receiver (high-low SST)
- ◆ Laser-retro-reflector
- ◆ Magnetometer
- ◆ Ion drift meter
- ◆ Star sensors

PI : Ch. Reigber / H. Lühr GFZ

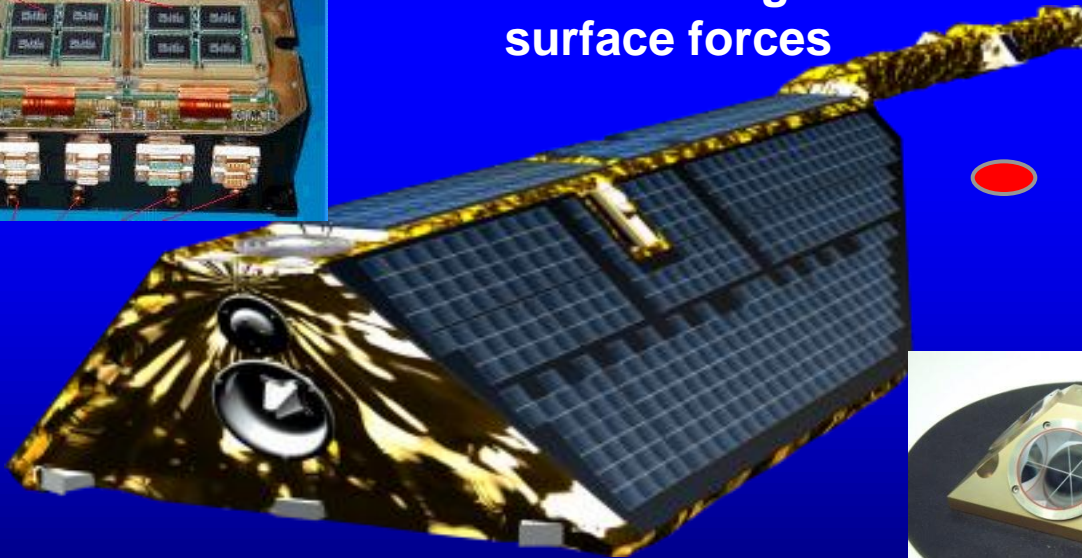
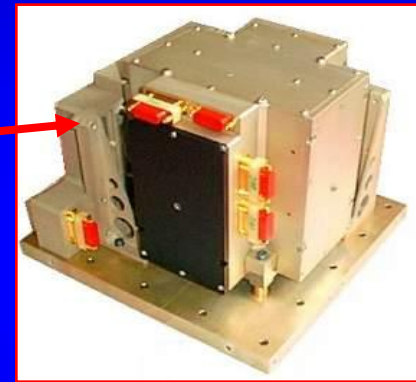
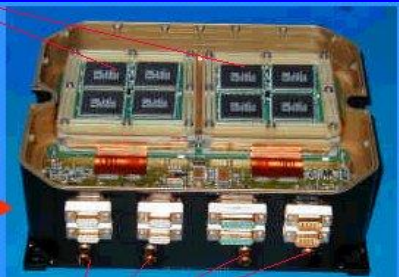
# ***CHAMP Mission Supports Fundamental Progress in Static Gravity Field Recovery due to:***

- continuous high-low GPS SST tracking at a very low altitudes (450 - 200 km )

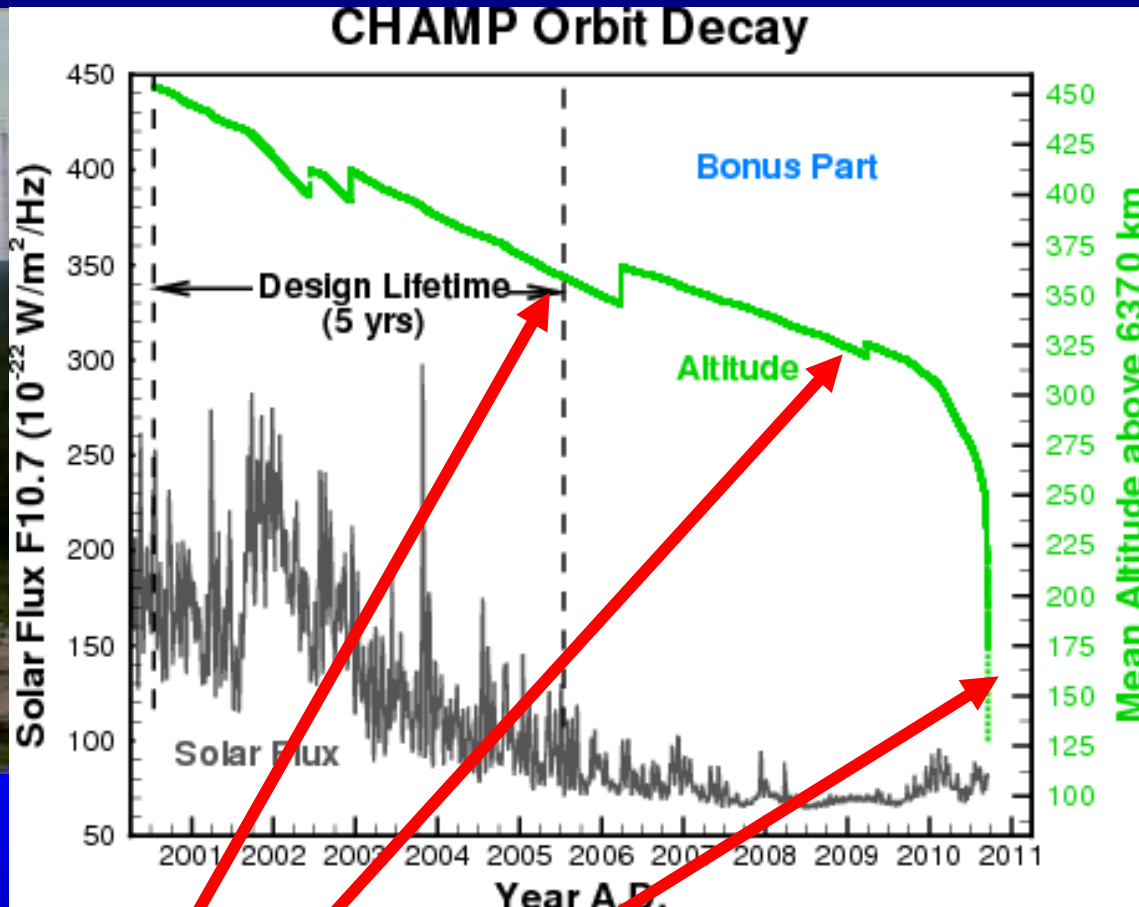
- Accelerometry for direct measurements of difficult to model non gravitational surface forces

- Polar orbit ( 87deg) for an almost complete coverage of the Earth

- Long mission duration to resolve temporal gravity field variations



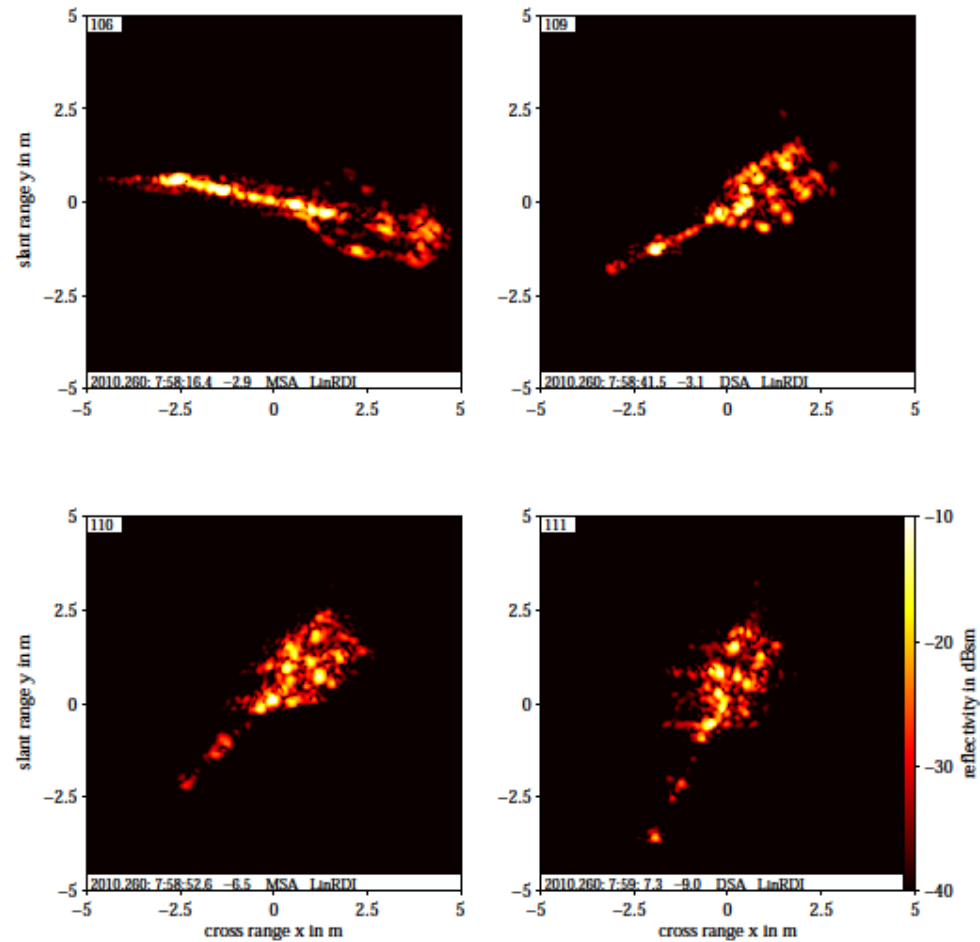
# CHAMP Orbit Profile



## Mission:

- Launch: July 15, 2000
- Planned mission duration: 5 years
- Four orbit change maneuvers (uplifts)
- Final mission duration: 10 years, 2 month
- End of mission: September 19, 2010

# ISAR imaging of object 26405



Object No: 26405

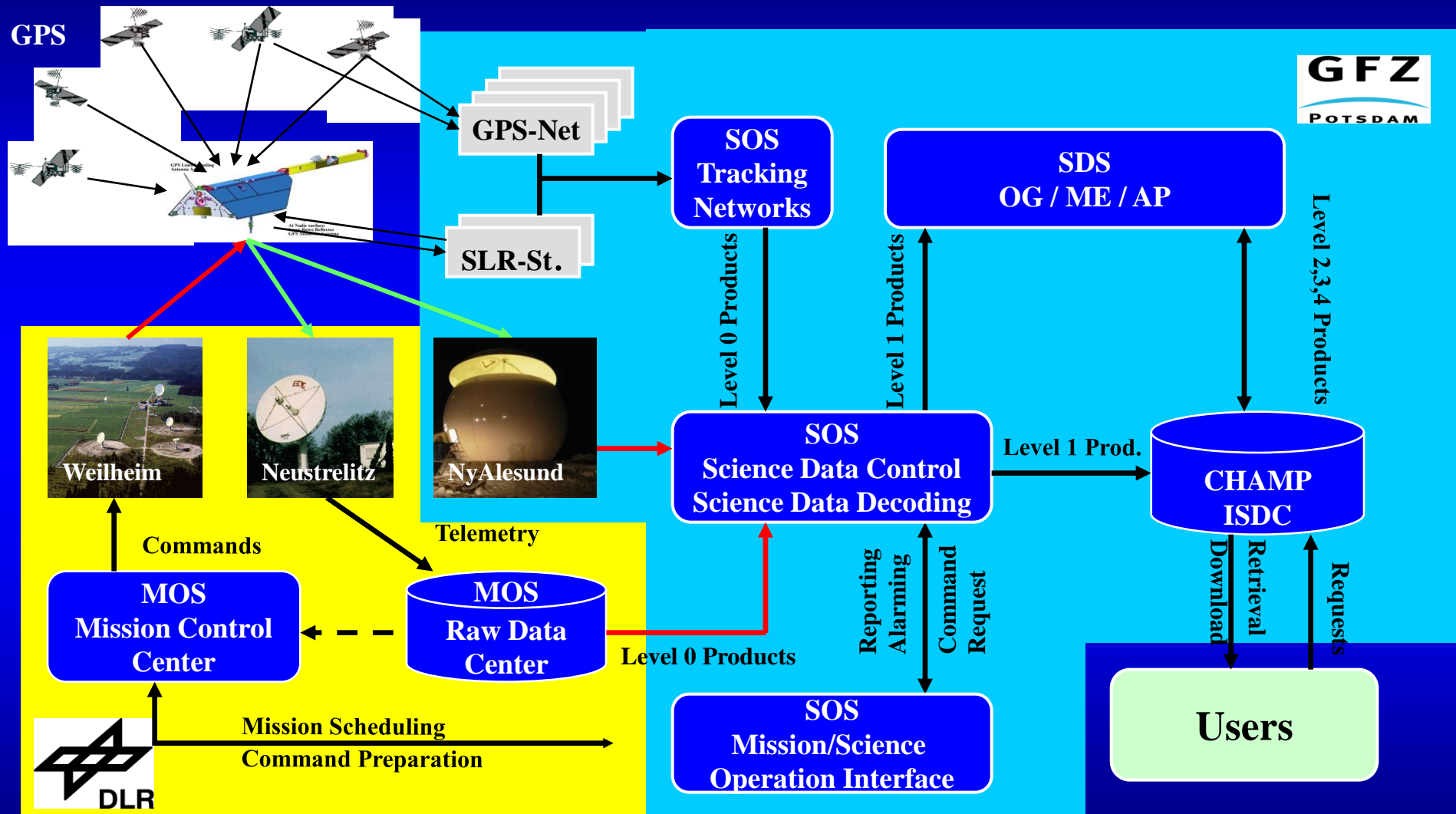
Date of observation: 17 09 (260) 2010, 07:56:00 (UTC)

Window: Hamming,  $\Delta\phi = 5.56^\circ$ ,  $\Delta t = 3.70$  s,  $\delta_x = 0.12$  m,  $\delta_y = 0.12$  m

Parameters: year, day, time (UTC), maxRCS/dBsm, autofocus, RDI method



# CHAMP Mission Space & Ground System

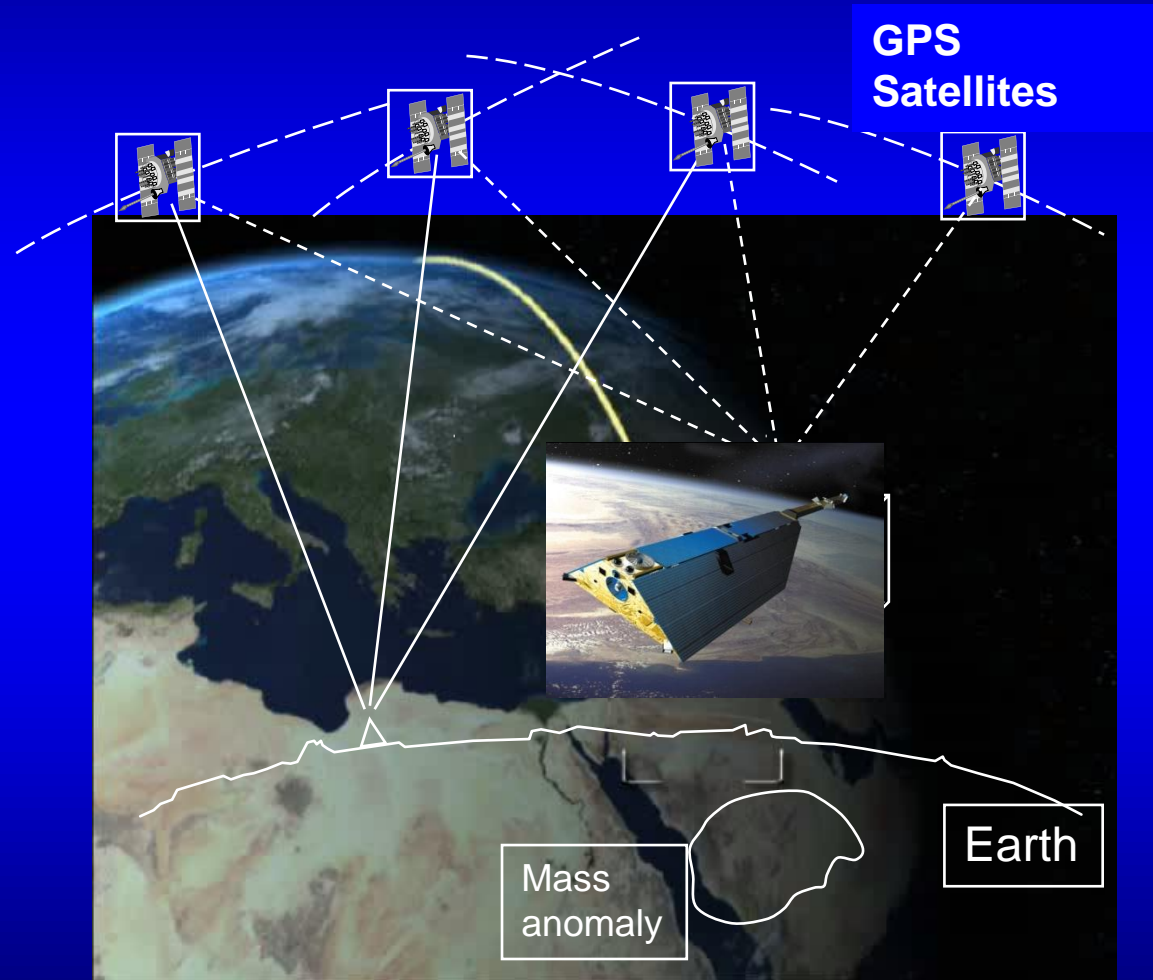
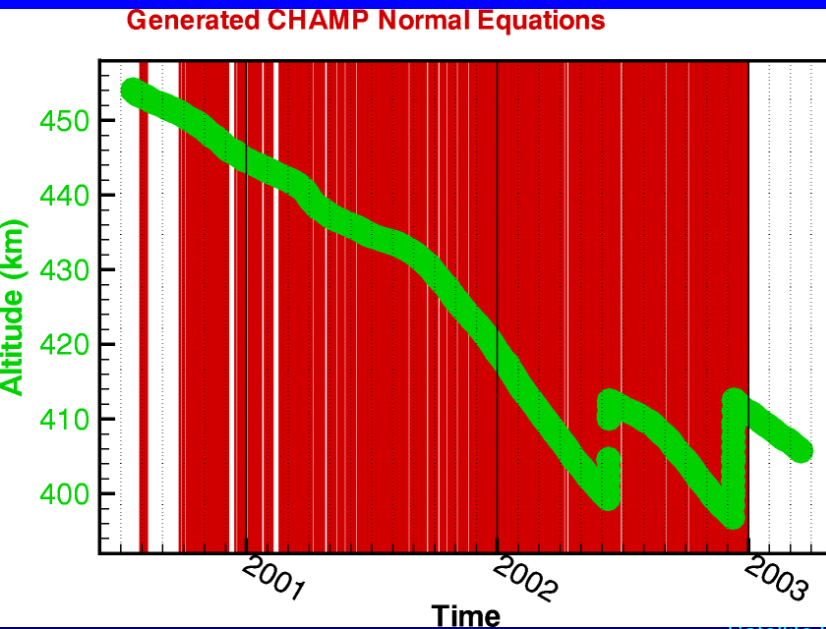


# The CHAMP Mission- Gravity Recovery Part

## Principle:

- GPS/CHAMP h-SST-tracking
- 3D-measurement of surface forces
- Measurement of  $\{x_i, \dot{x}_i\}$

## Computed Normal Equations:



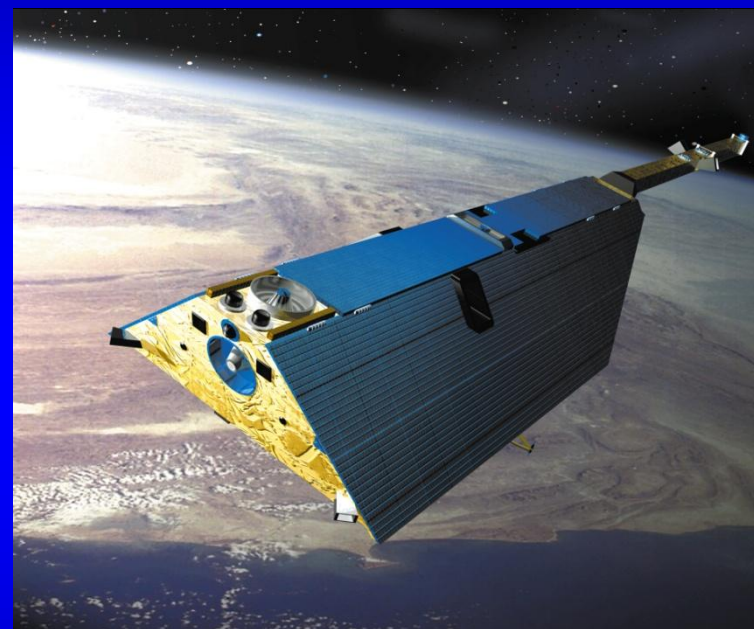
# CHAMP- Major Achievements

- Mission operation:

- Almost uninterrupted and smooth operation of the satellite and instrumentation over more than 10 years
- Almost complete recovery of all science data (>98%) with the receiving antennas in Neustrelitz, Weilheim and Ny Alesund
- Routine processing of all science data within the CHAMP Science Data System

- Science output:

- First continuous long-term occultation data set for GPS climatology
- First precision crustal magnetic anomalies field and field variability due to ocean water flow
- Best pre-GRACE gravity field from a single satellite data set
- Application of new gravity recovery approaches



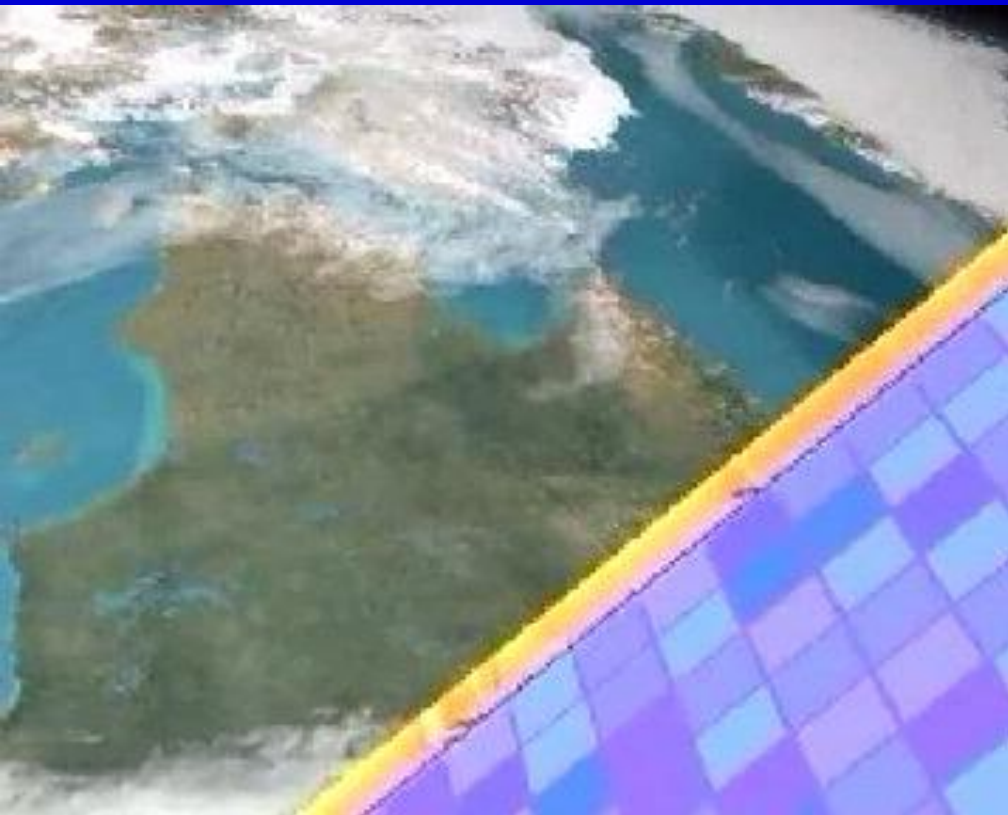
**Mission life: 200 % of design life**

# GRACE

Mission of NASA (USA), DLR (Germany), JPL(USA) , CSR/UTEX(USA)  
& GFZ Potsdam (Germany) within NASA's ESSP Programme

**Launch:** Mar 2002    **Planned Mission duration:** 5 yrs

**Orbit:** nearly circular polar orbit (89°), 500 km altitude



## Mission goals:

- ◆ High resolution, long- and medium-wave, mean & time- variable gravity field mapping
- ◆ Atmosphere sounding

## Payload:

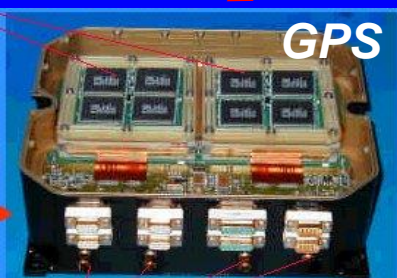
- ◆ Microwave distance ranging between the satellites ( $<5 \mu$ ), relative velocity ( $<0.5 \mu/s$ ) (low-low SST)
- ◆ 3-D accelerometer ( $10^{-10} \text{ m/s}^2$ )
- ◆ 24-channel GPS receiver (high-low SST)
- ◆ Laser-retro-reflector
- ◆ Star sensors

*PI: Byron D. Tapley / CSR Austin*

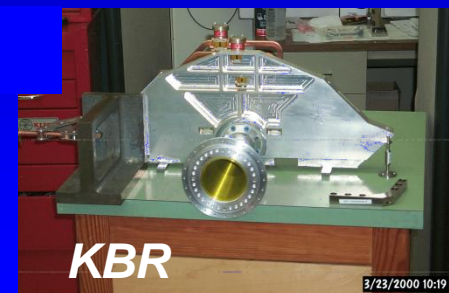


# ***GRACE Mission Supports Fundamental Progress in Static & Time Variable Gravity Field Recovery due to:***

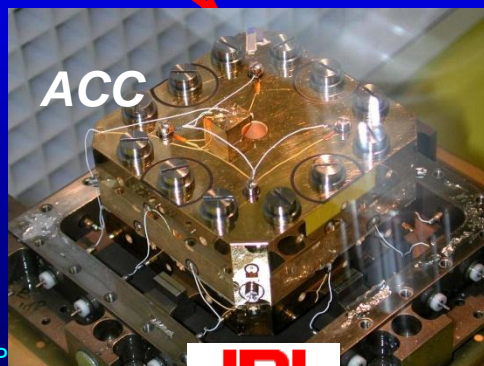
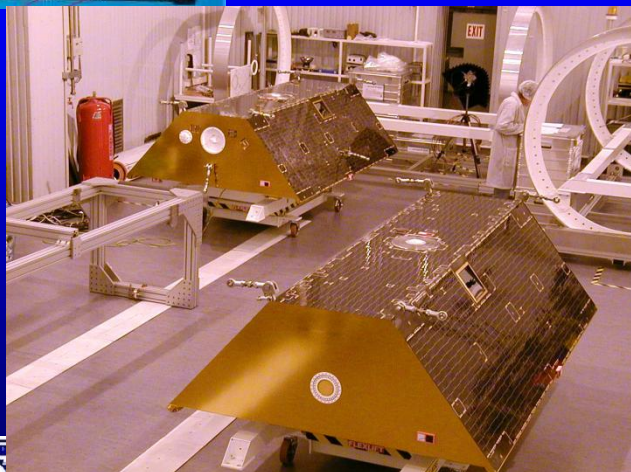
- Continuous high-low & low-low SST tracking at a low altitude (approx. 490 km)



- Super precise accelerometry for direct measurements of non gravitational surface forces

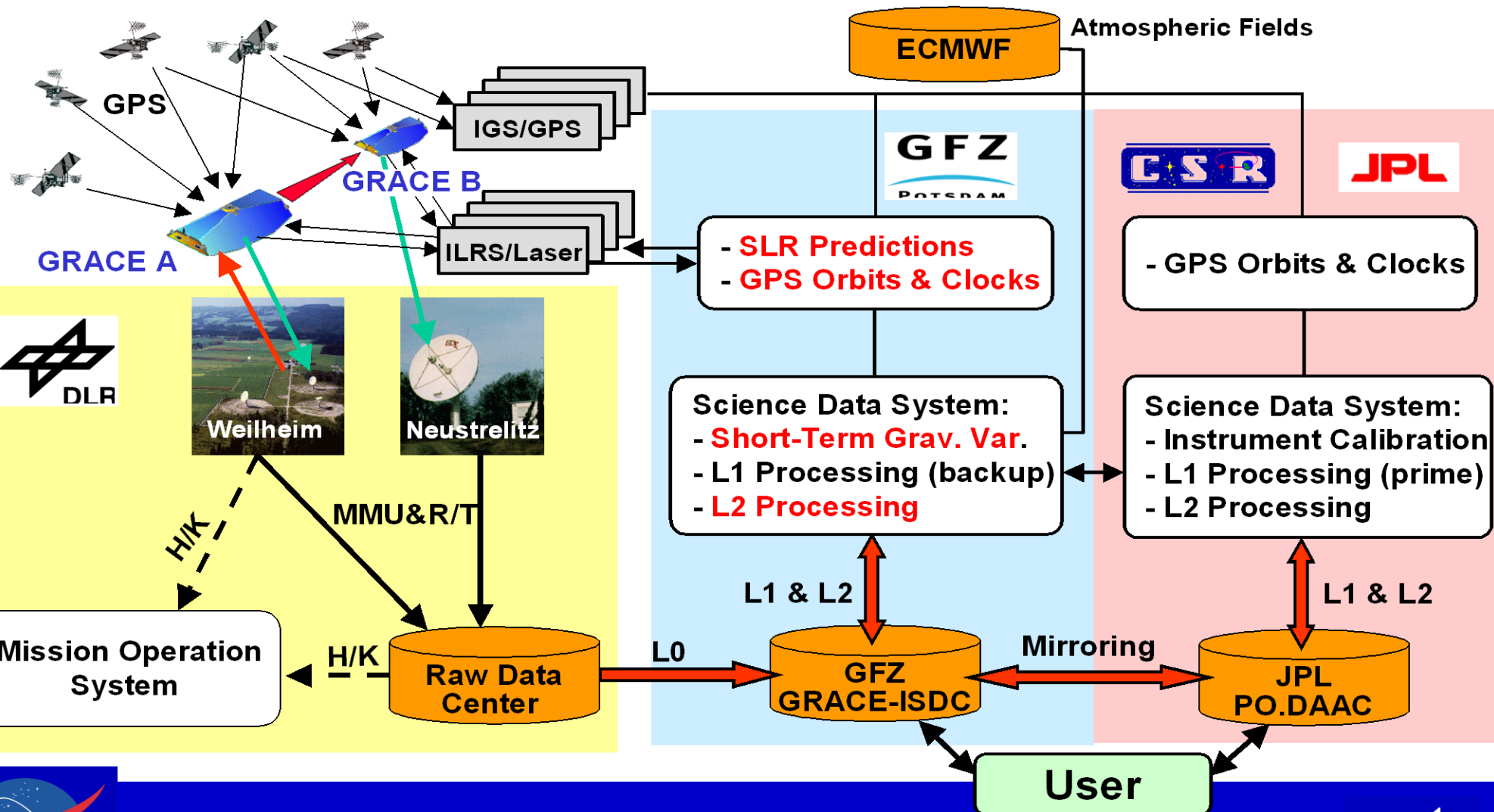


- Polar orbit ( 89 deg) for a complete coverage of the Earth



- Long mission duration to resolve temporal gravity field variations

# GRACE Ground Segment Operations

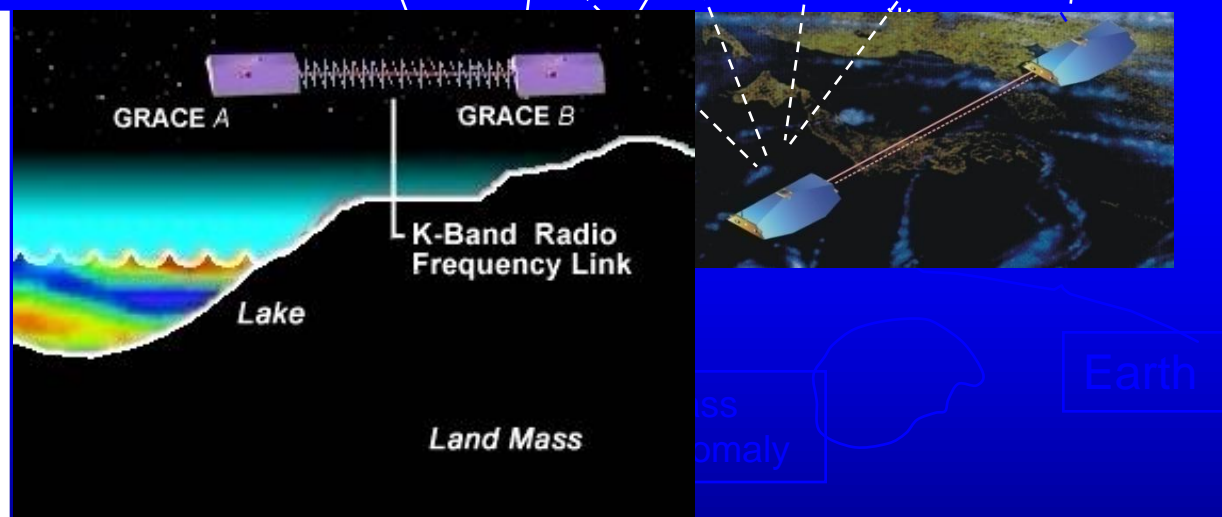
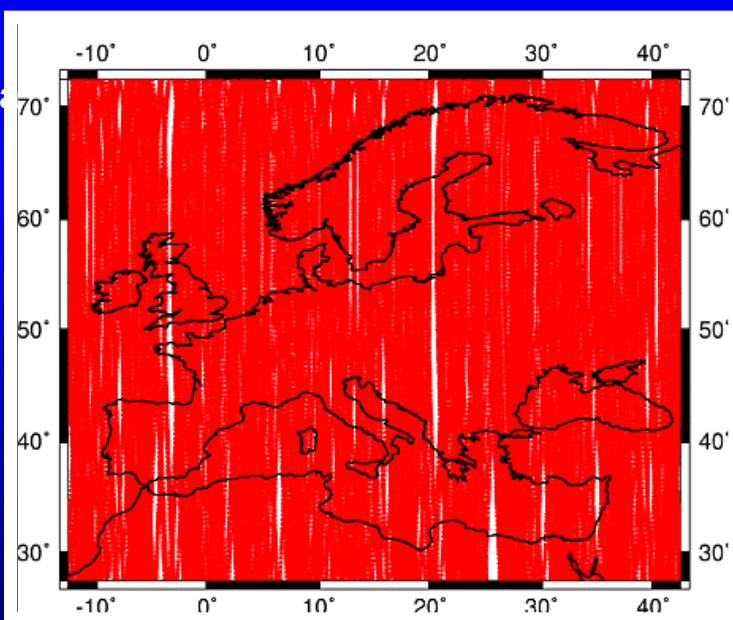


# GRACE Mission - Gravity Recovery Part

## Principle:

- GPS/GRACE hl-SST-tracking
- GRACE A/B ll-SST-tracking
- 3D-measurement of surface forces
- Measurement of  $\{x_i, \dot{x}_i\}$
- Measurement of  $\{\rho_i, \dot{\rho}_i\}$

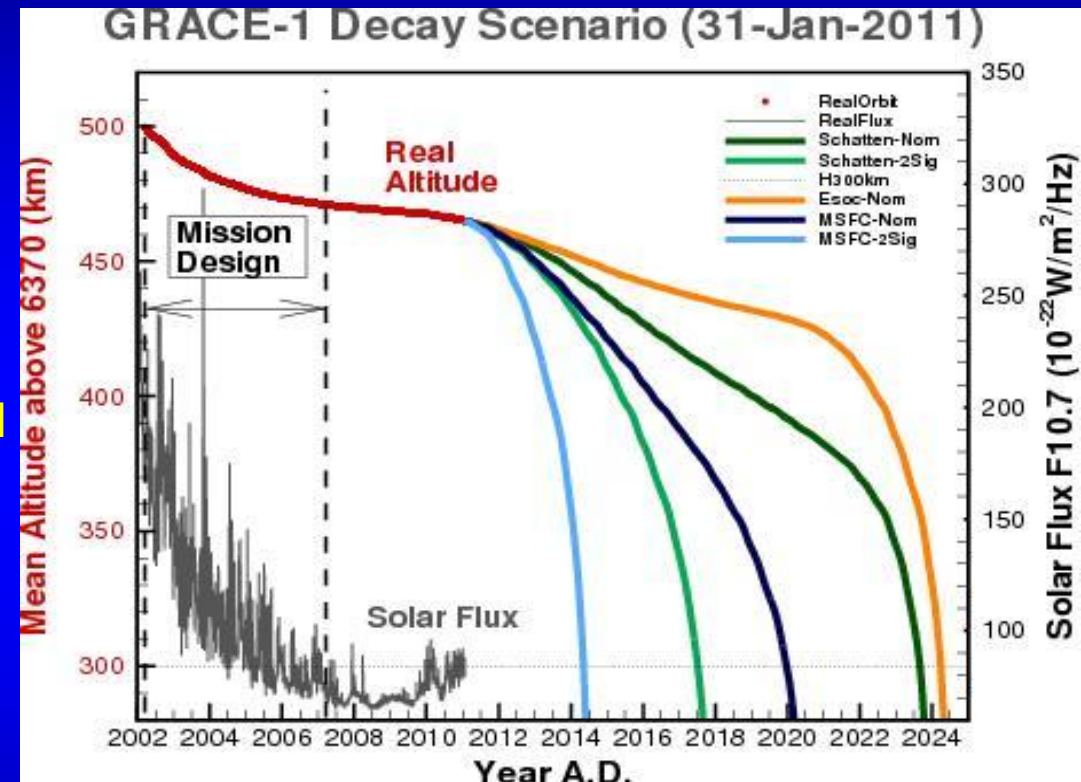
## Data coverage over Europe:



Source: NASA

# Current Status of GRACE Satellites and Instruments

- All science instruments in good health
- A major star-camera software upload was completed 4 weeks ago, significantly reducing the outages in the attitude measurement data.
- Battery Status
  - The battery capacity is diminished compared to the time of launch
  - Individual cells (2<sup>nd</sup> out of 20) show signs of “weakness”
- Operate the satellite by
  - preventing overcharging of the battery
  - limiting the power consumption during adverse conditions (e.g. switch off the GRACE-B accelerometer in January 2011 = no gravity field could be obtained!)



- The life time of GRACE is limited!



# GOCE

Mission of European Space Agency ESA within the *Living Planet Programme*  
as one of the first *Earth Explorer Missions*

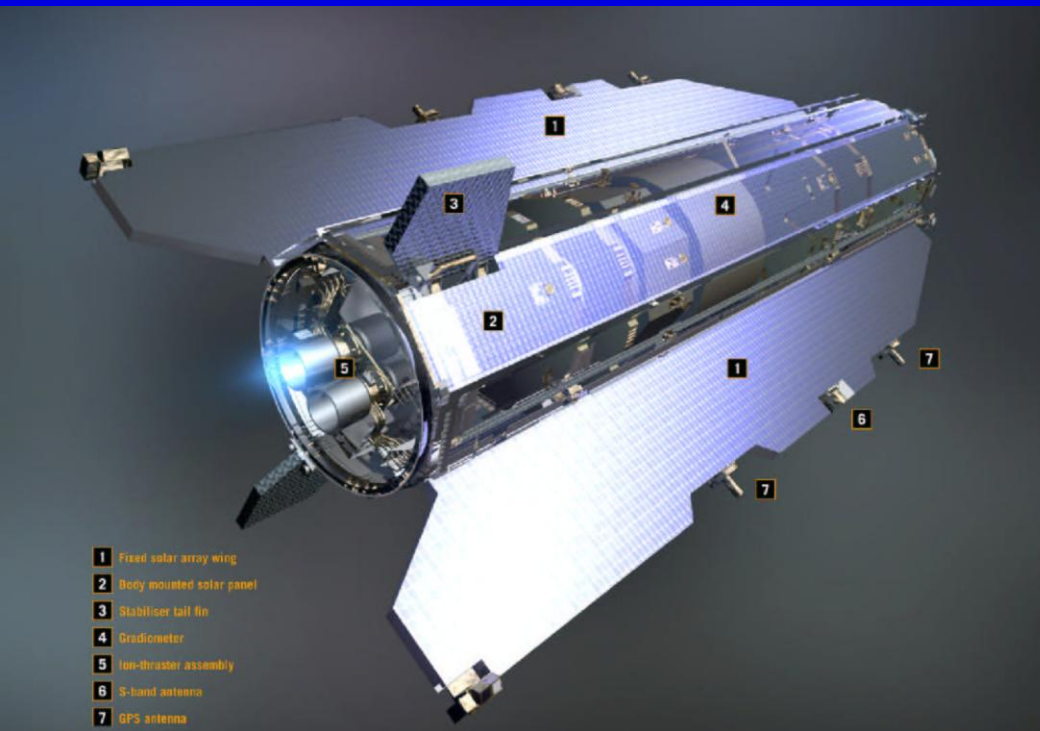
**Launch:** ~2009 **Planned Mission duration:** 20 months  
**Orbit:** nearly circular, sun-synchronous orbit (96°), 250 km altitude

## Mission goal:

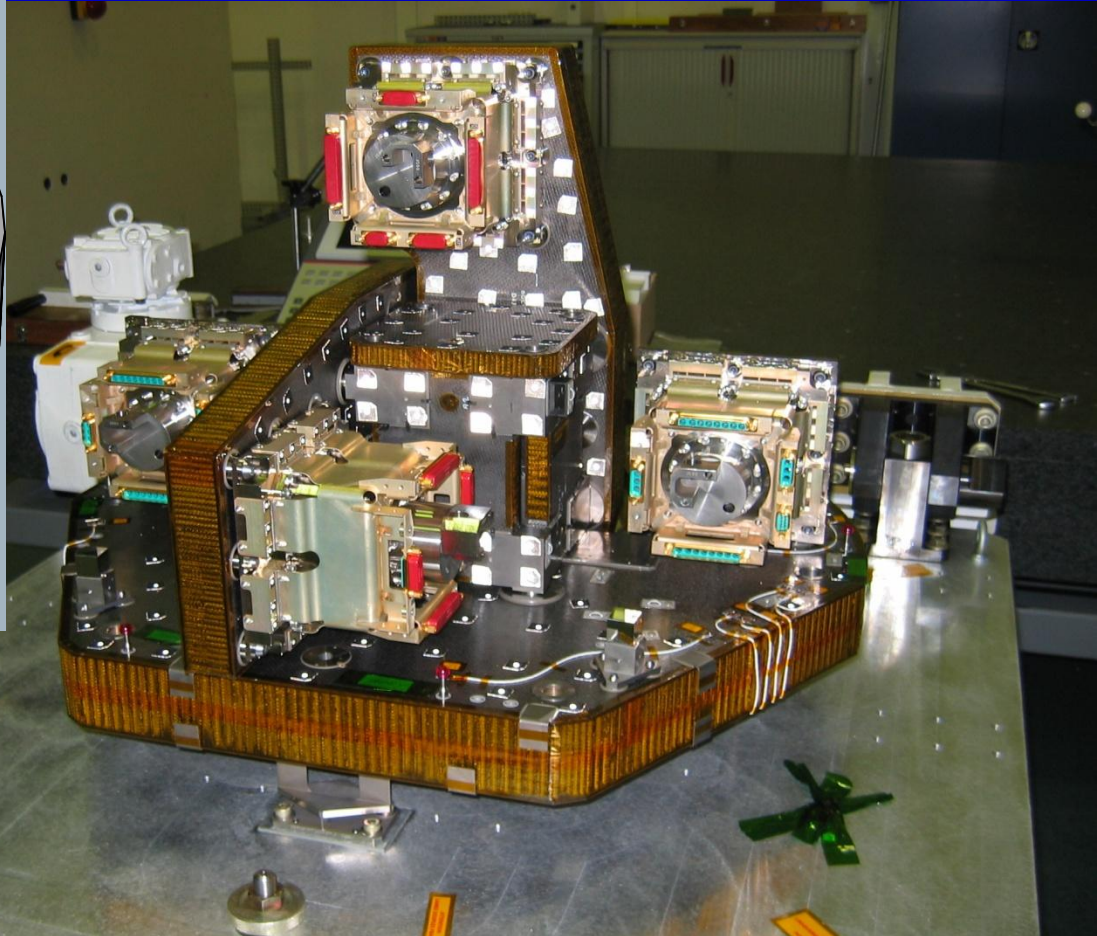
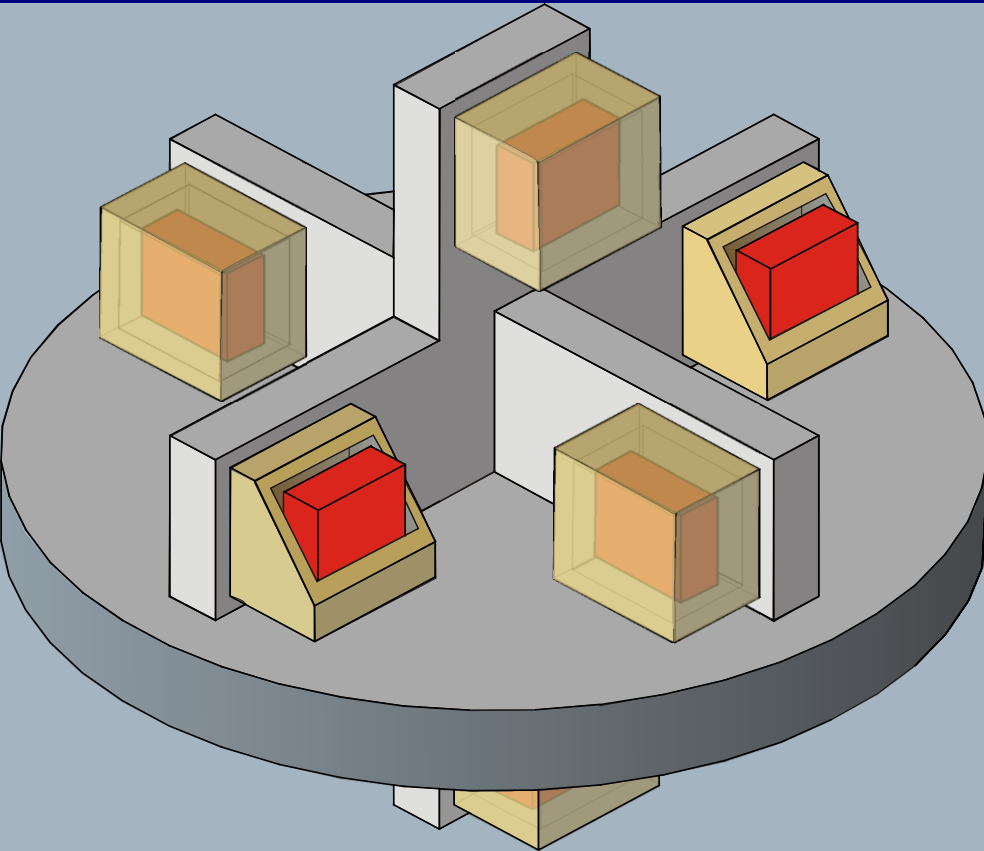
Determination of the geoid with 1 cm accuracy and 100 km spatial resolution

## Payload:

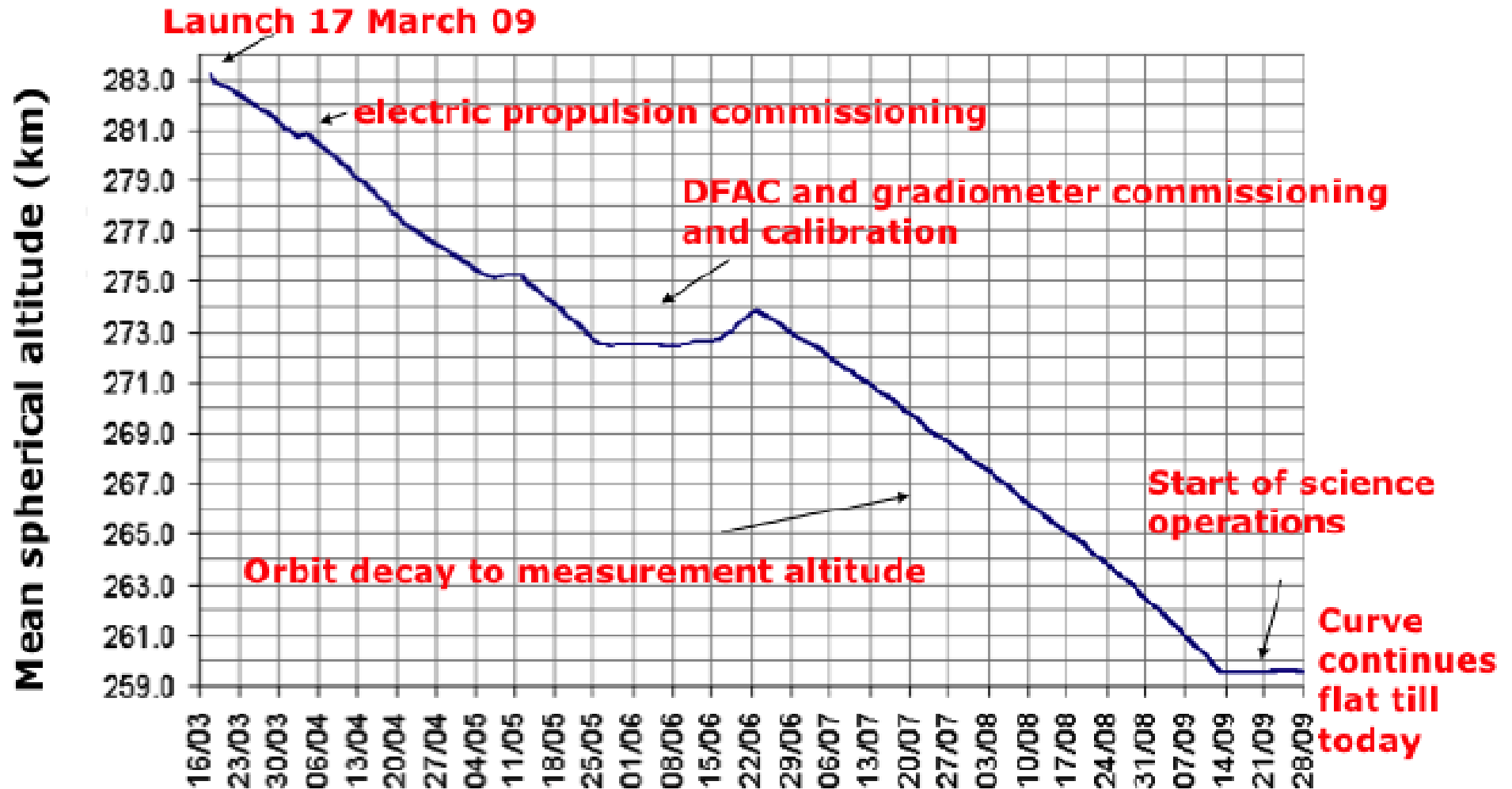
- ◆ Gradiometer consisting of 3 pairs of 3-axis accelerometers ( $3 \times 10^{-12} \text{ s}^{-2}$ )
- ◆ Drag-free compensation system
- ◆ 12-channel GPS receiver (high-low SST)
- ◆ Laser-retro-reflector
- ◆ SREM: Standard Radiation Environment Monitor
- ◆ Star sensors



# *Electrostatic Gravity Gradiometer*

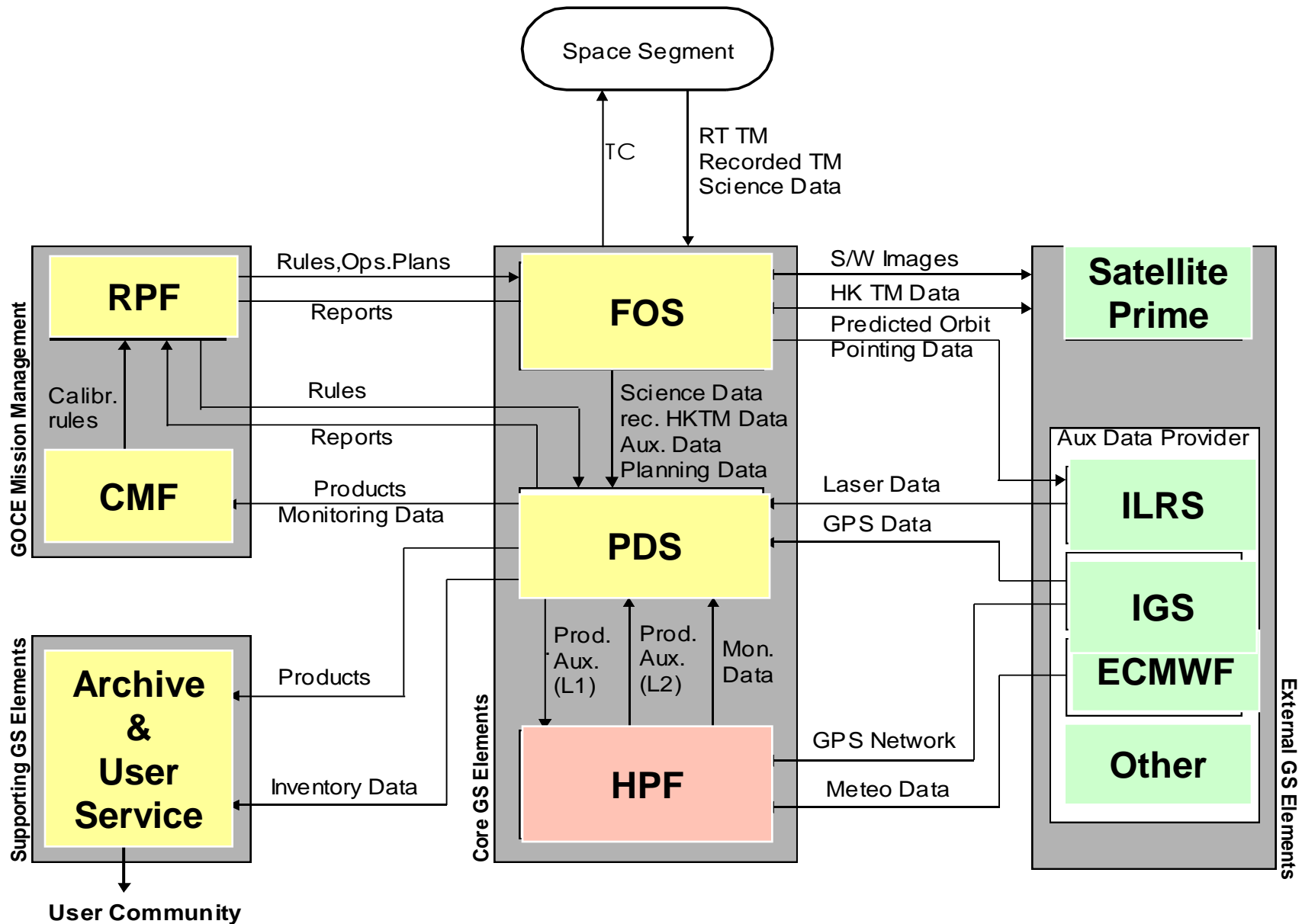


# GOCE Orbit Profile

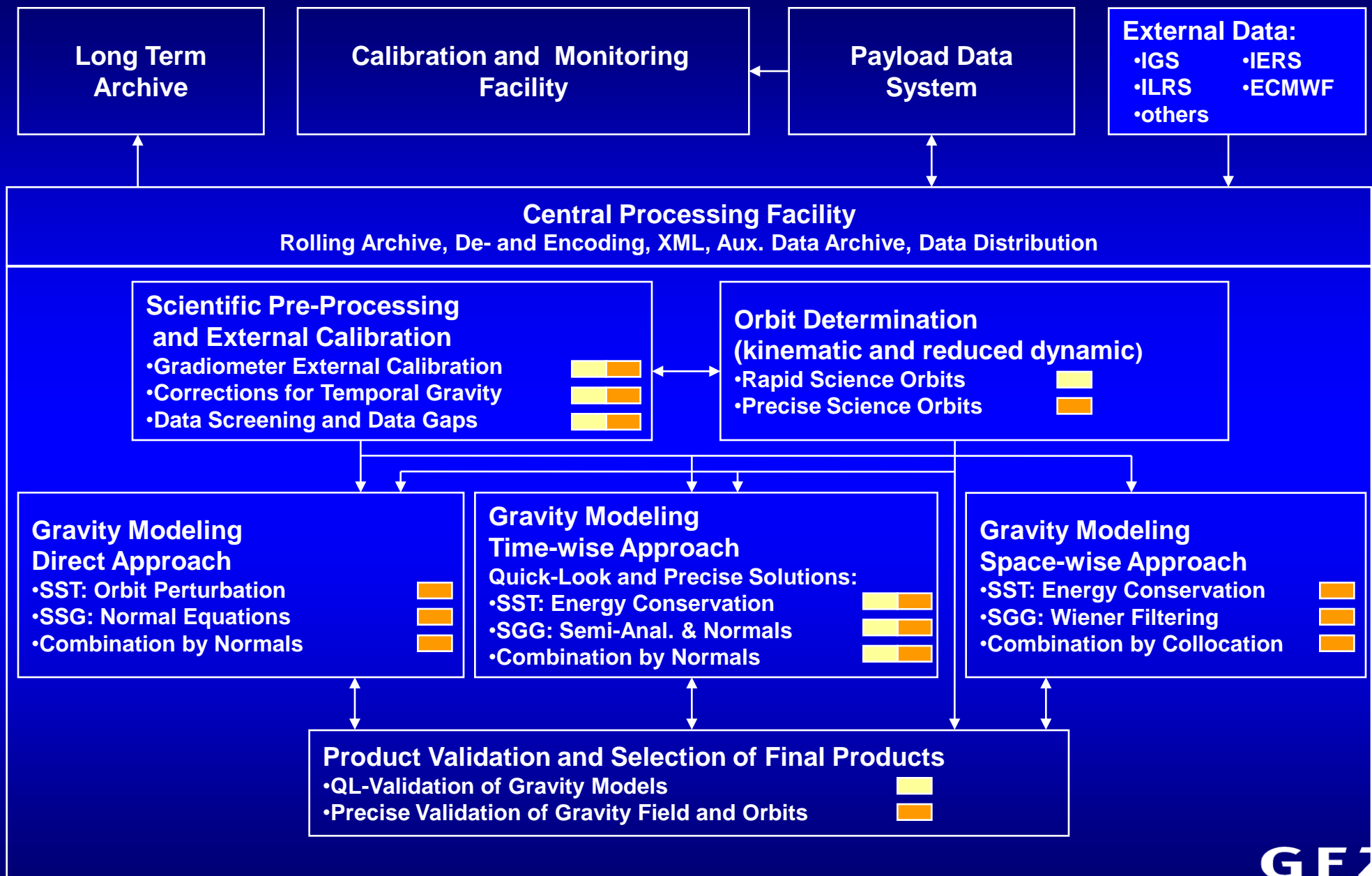


Flohberghagen et al., 2011

# GOCE Ground Segment

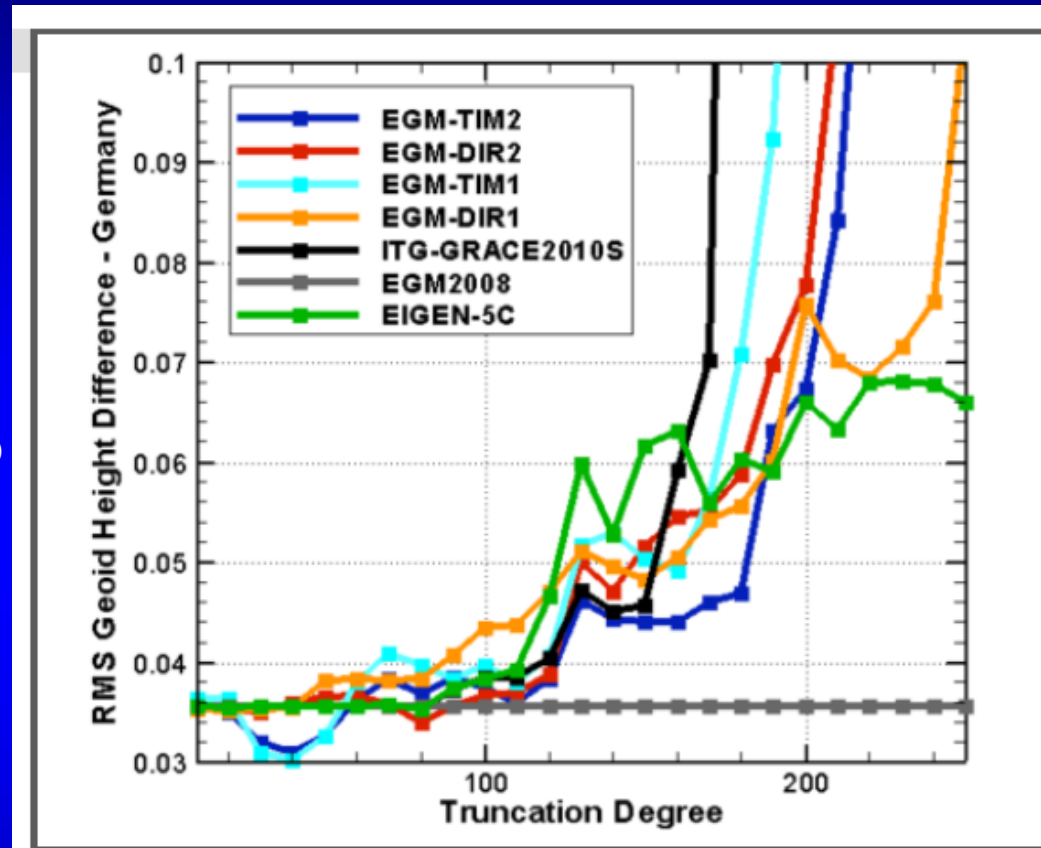




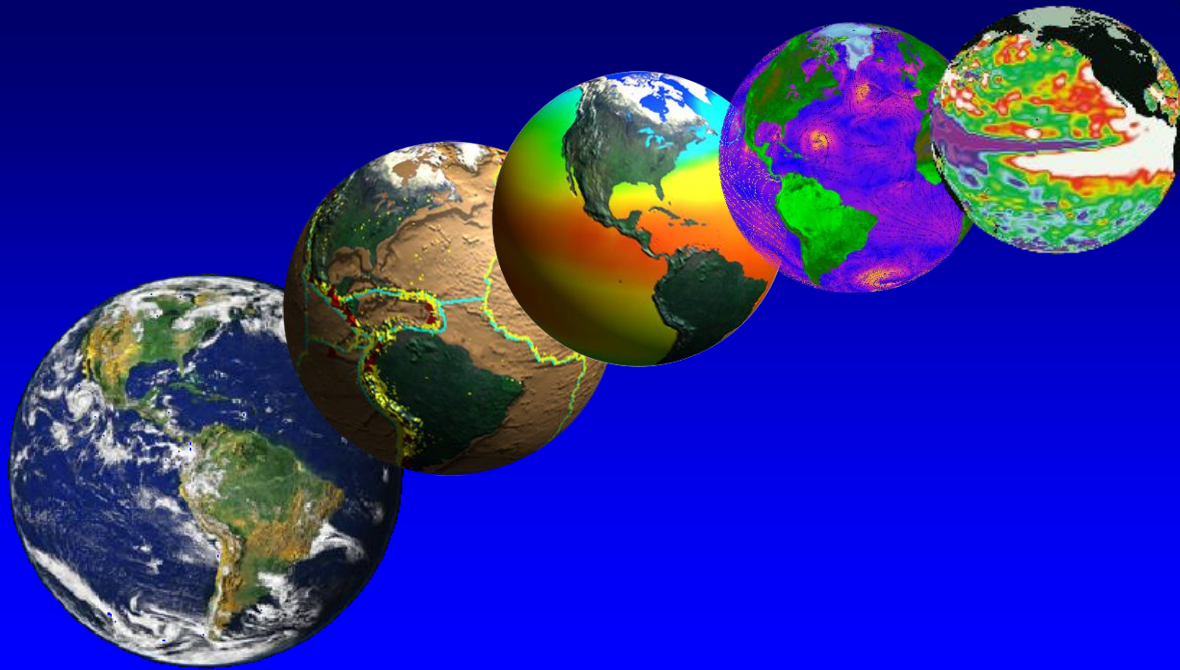


# Current Status of GOCE Spacecraft and Instruments

- The GOCE satellite and its payload are in good health and deliver top science data (SGPS and gradiometer).
- The persistent low solar activity in 2010/11, the good health of the power subsystem and the excellent thermo-elastic stability of the satellite and the gradiometer have allowed to continue science measurements also during eclipse phases. More data than originally planned will be available.
- Two satellite anomalies were encountered.
  - First an anomaly in the data communication between the main computer and the telemetry module
  - Secondly a software problem with the GPS payload
  - Both anomalies are solved and no impact on successful mission continuation

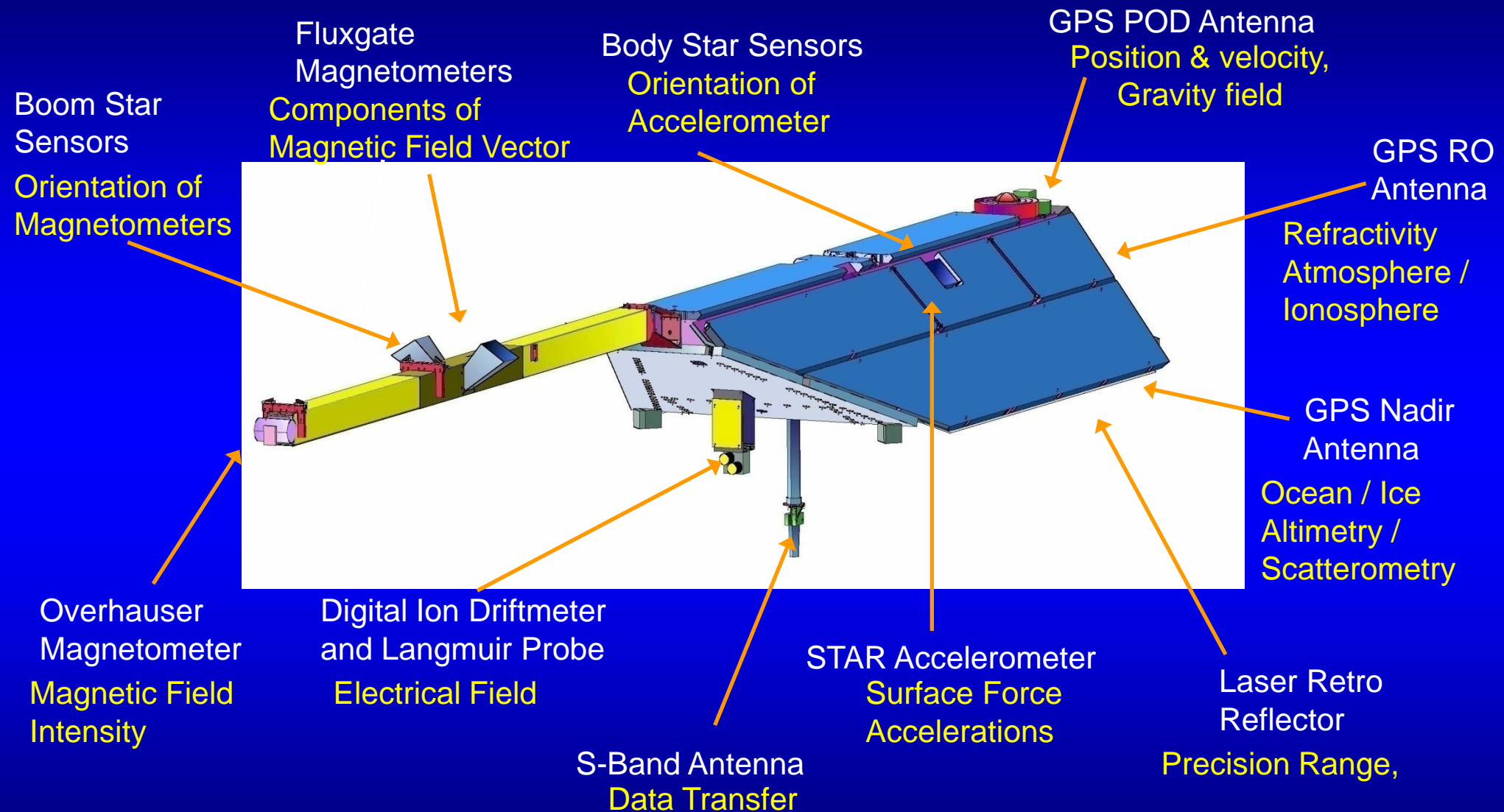


- Operation will be extended until end of 2012 !



# CHAMP/GRACE /GOCE and the Atmosphere

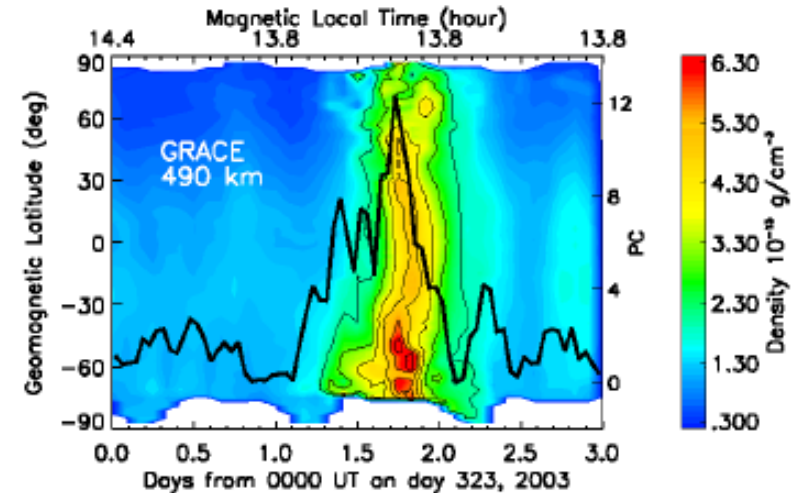
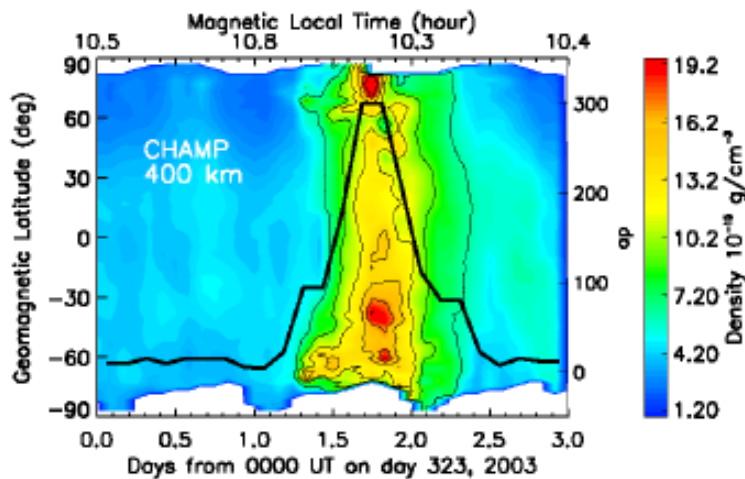
# What do we measure?



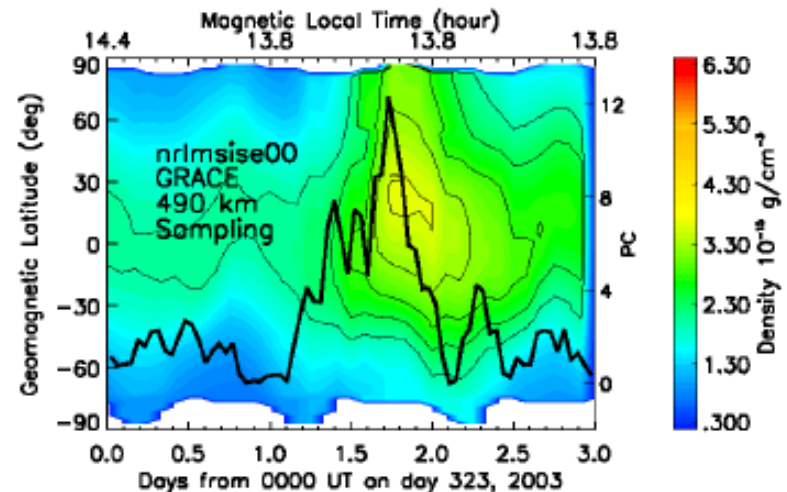
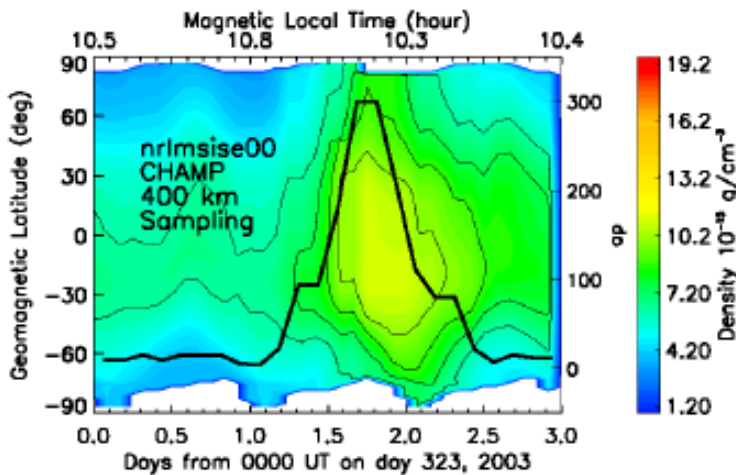


# Atmospheric Density Observations

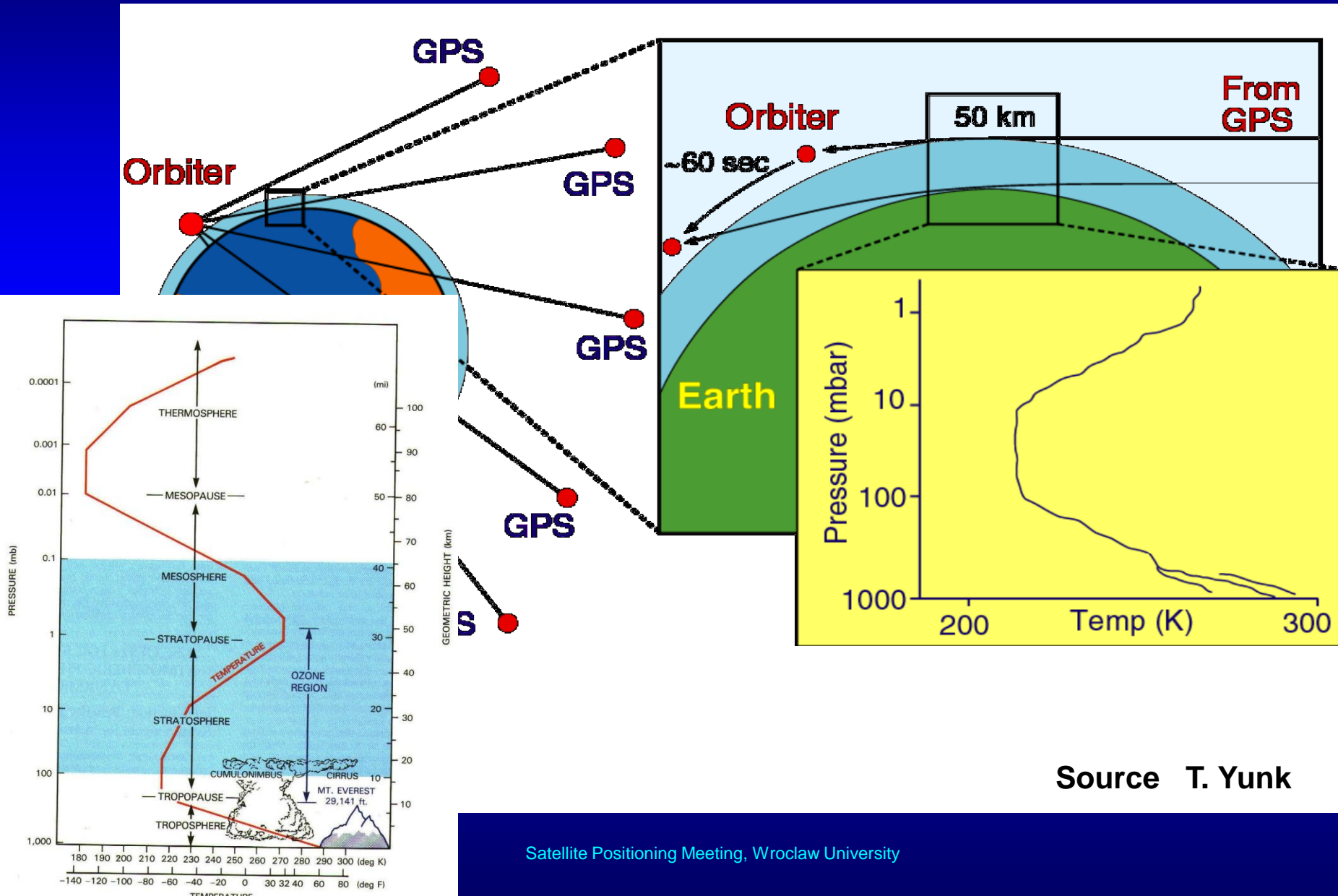
## Variabilité atmosphérique observée par accélérométrie



## Variabilité atmosphérique modélisée (450-500km d'altitude)

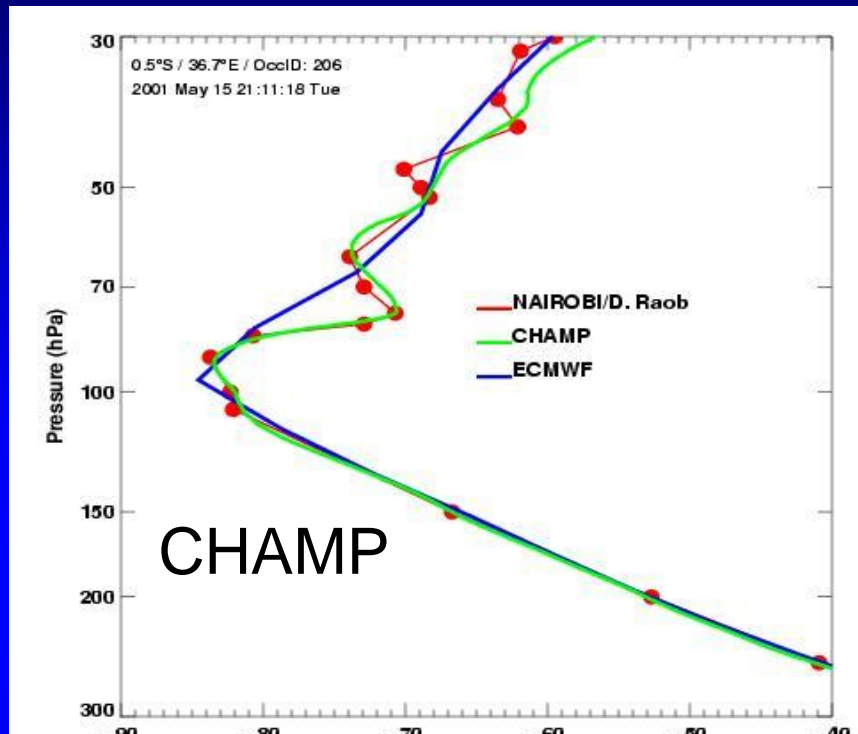


# GPS-Atmosphere-Limb-Sounding

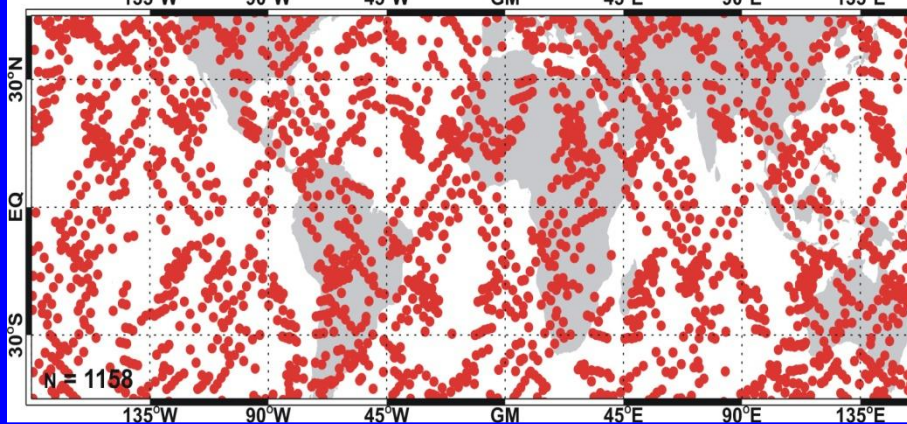


Source T. Yunk

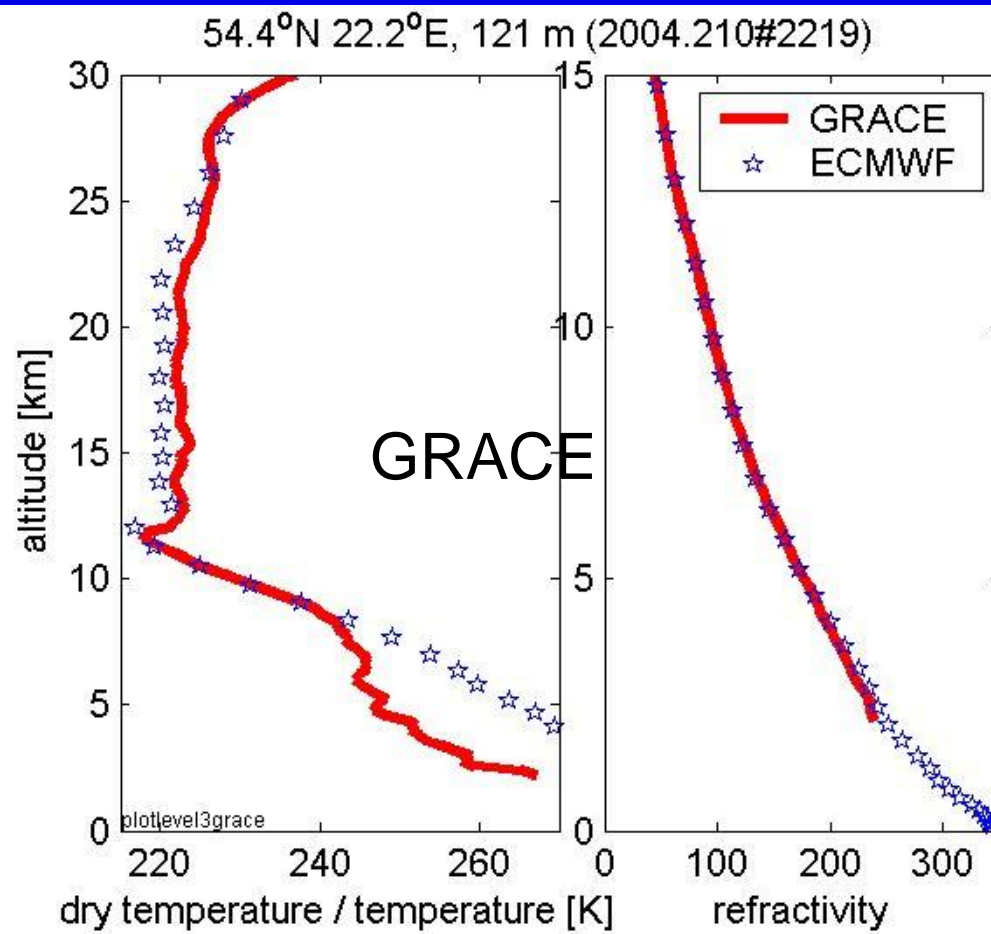
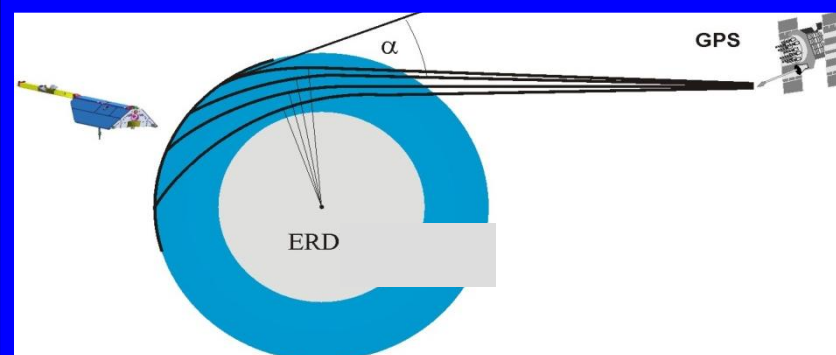
# CHAMP/GRACE First RO Profiles



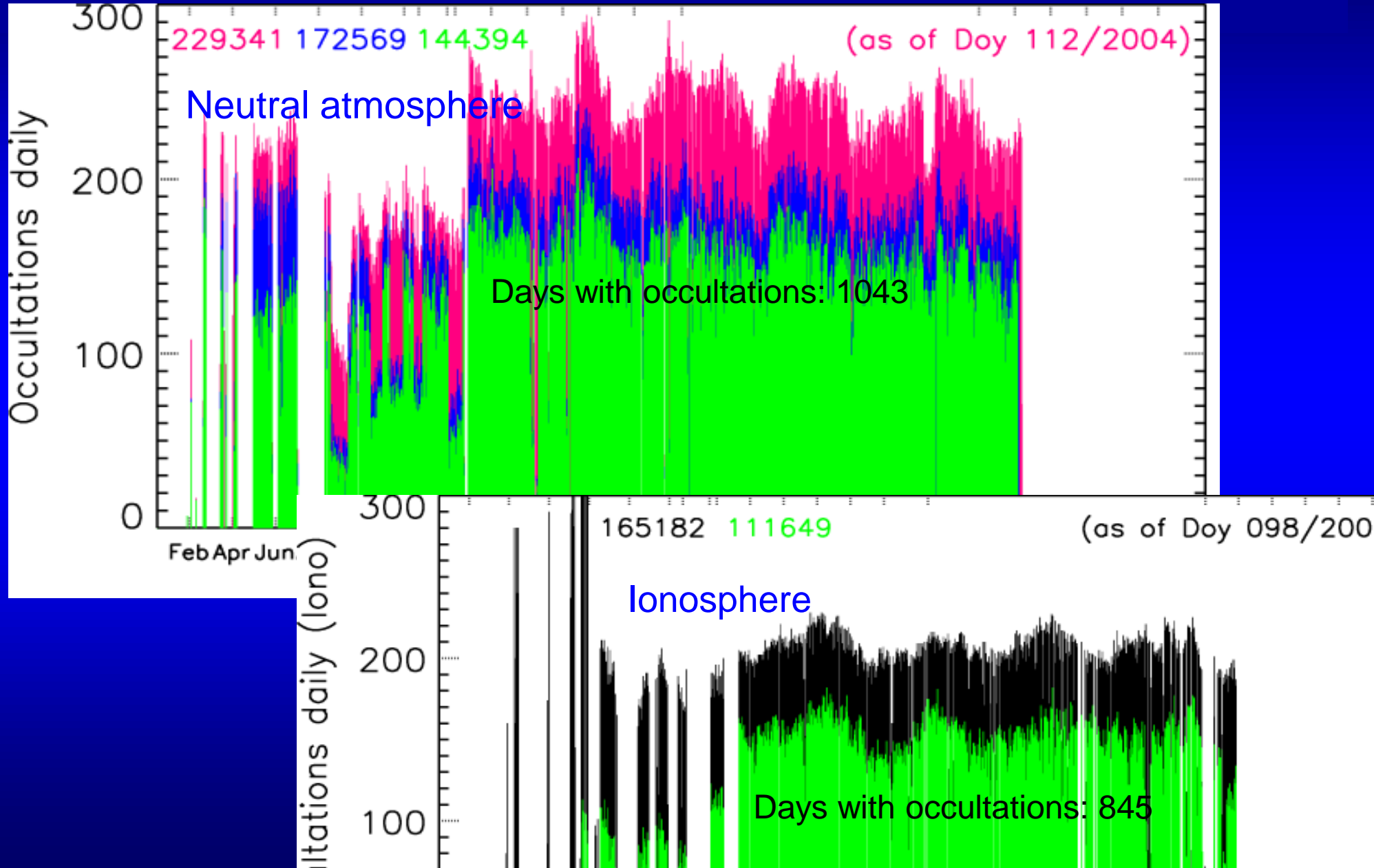
CHAMP Occultations between May 14 and Jun 10, 2001



Satellite Positioning



# Occultation statistics: Atmosphere/ionosphere



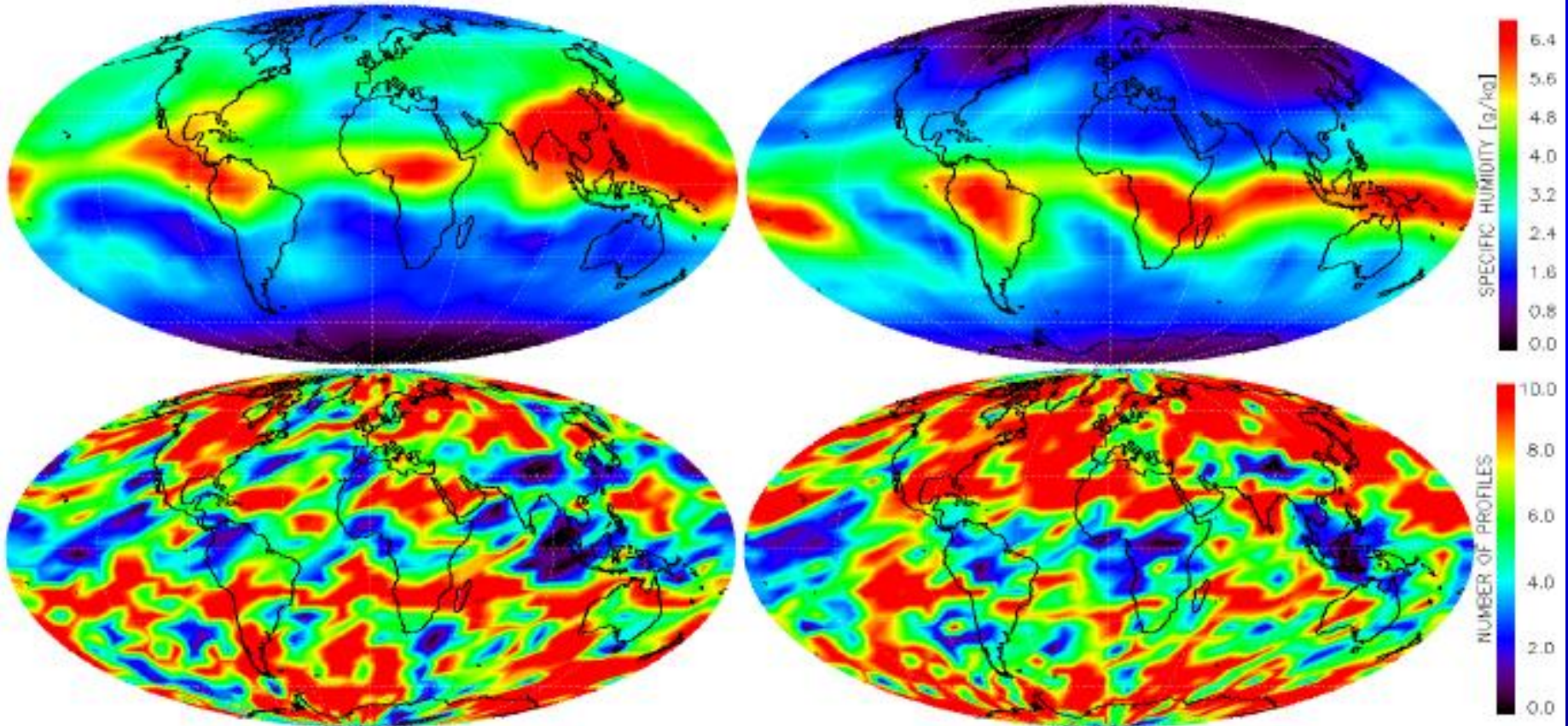


# CHAMP Water Vapour- Global Application

mean global water vapour distribution at 600 hPa

June to August 2004

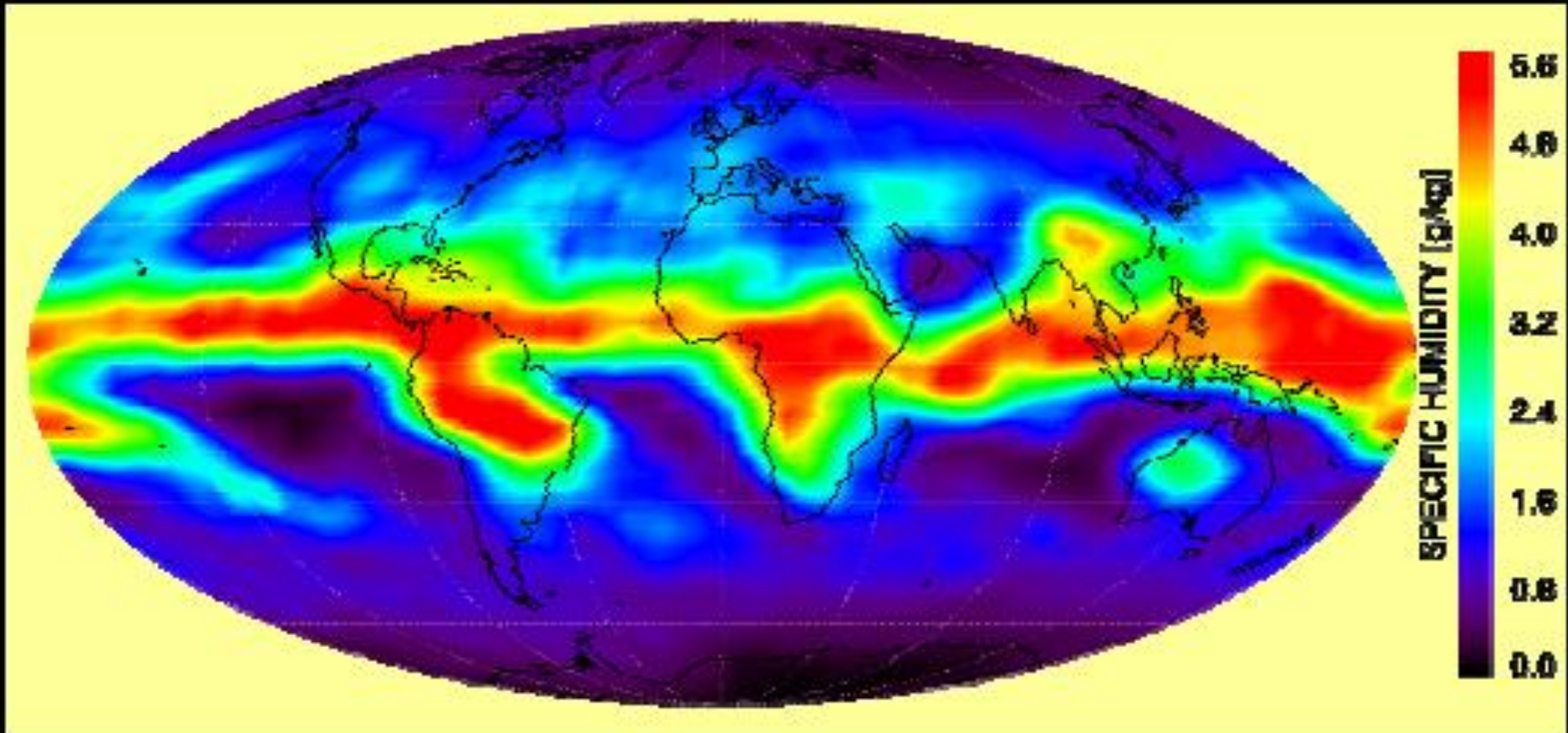
December 2004 to February 2005



data coverage according to a  $5^\circ\text{lat} \times 10^\circ\text{lon}$  grid  
(each ~12,000 data points)

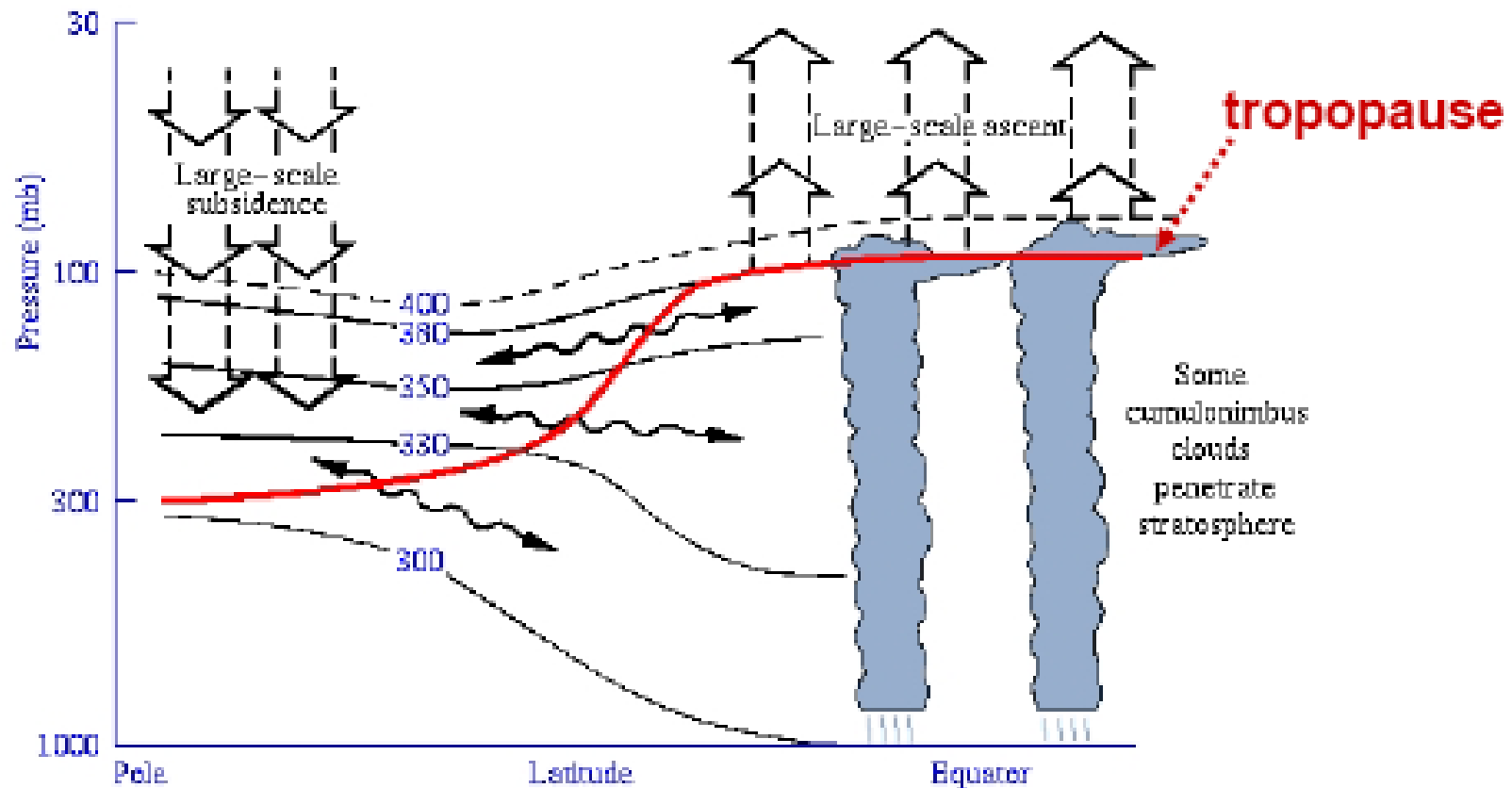
# ***Global Water Vapour (Oct.2006)***

**CHAMP + GRACE + COSMIC (2.5°lat, 5°lon)**





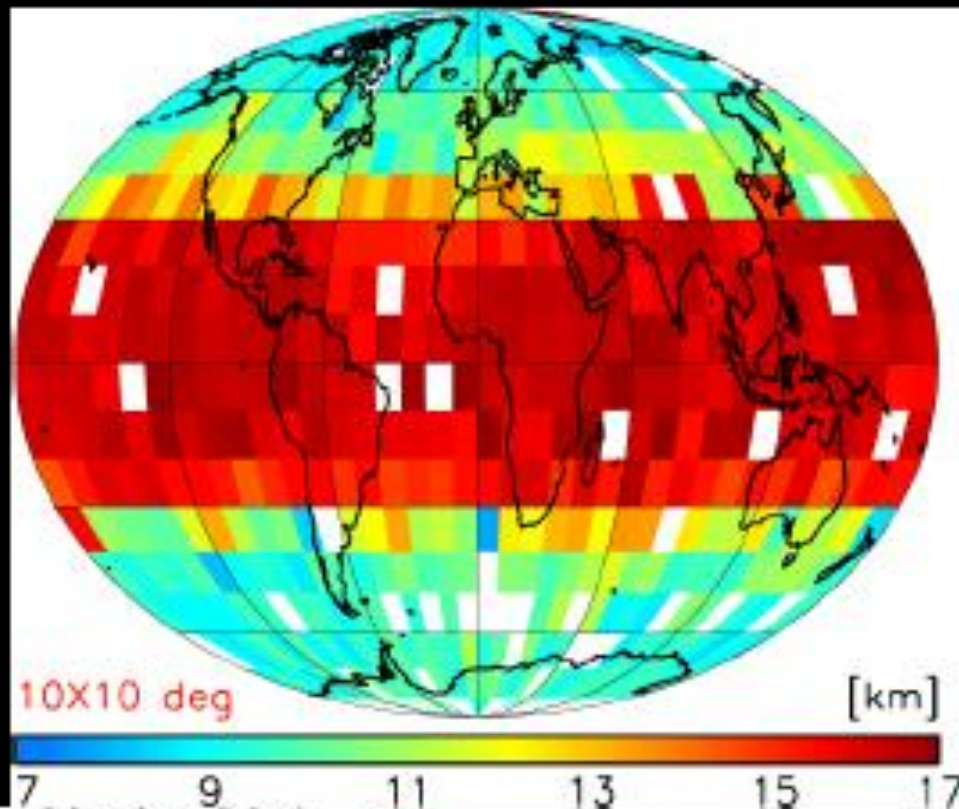
# Troposphere/Stratosphere Boundary Region



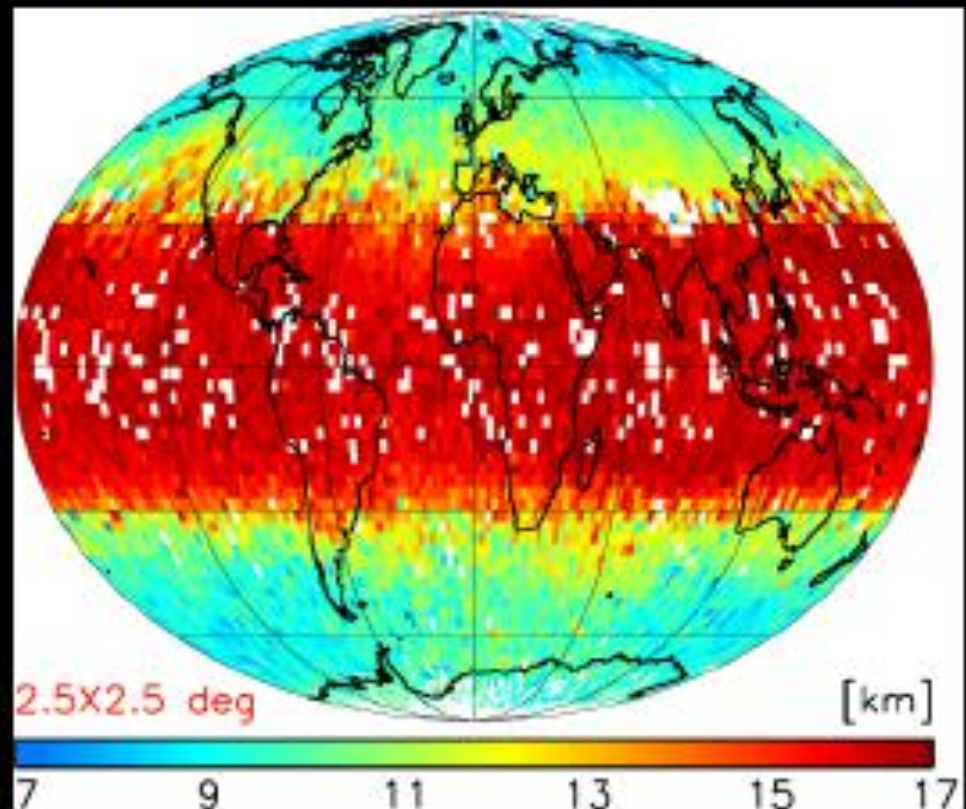
Holton et al., 1995

# Tropopause Altitude Oct. 2006

**CHAMP**



**CHAMP+GRACE+COSMIC**



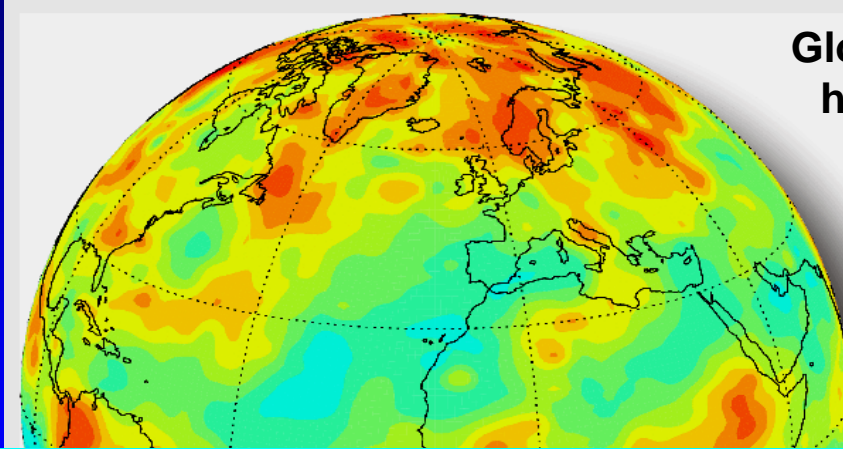
**Tropopause**

**46.235 profiles**

**Much higher resolution in time and space with new missions,  
But: CHAMP long term data set is unique**



# Key Properties of Atmospheric Occultation Data



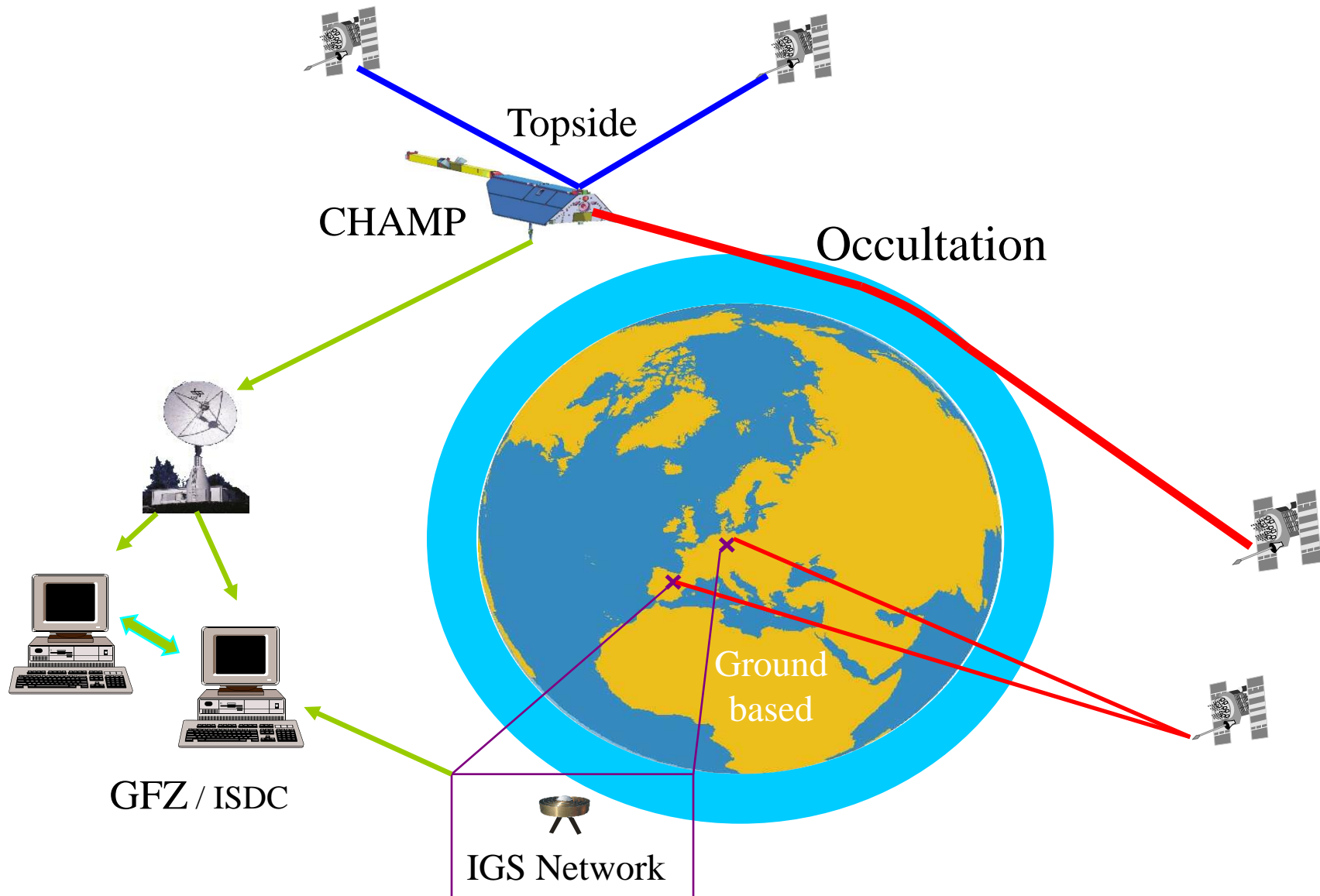
Global distribution of relative humidity (500 hPa) between mid May and June 2001, derived from CHAMP measurements

- **Exceptional accuracy: 0.1 % refractivity : ~0.5 K Temp.**
- **Exceptional vertical resolution: few hundred meters**
- **All-weather sensing: insensitive to clouds, precipitation, aerosols**
- **Each measurement is self-calibrating**  
→ All measurements directly comparable for all times
- **Independent height & pressure/temperature data**  
→ yields geopotential heights and wind fields
- **Global coverage**

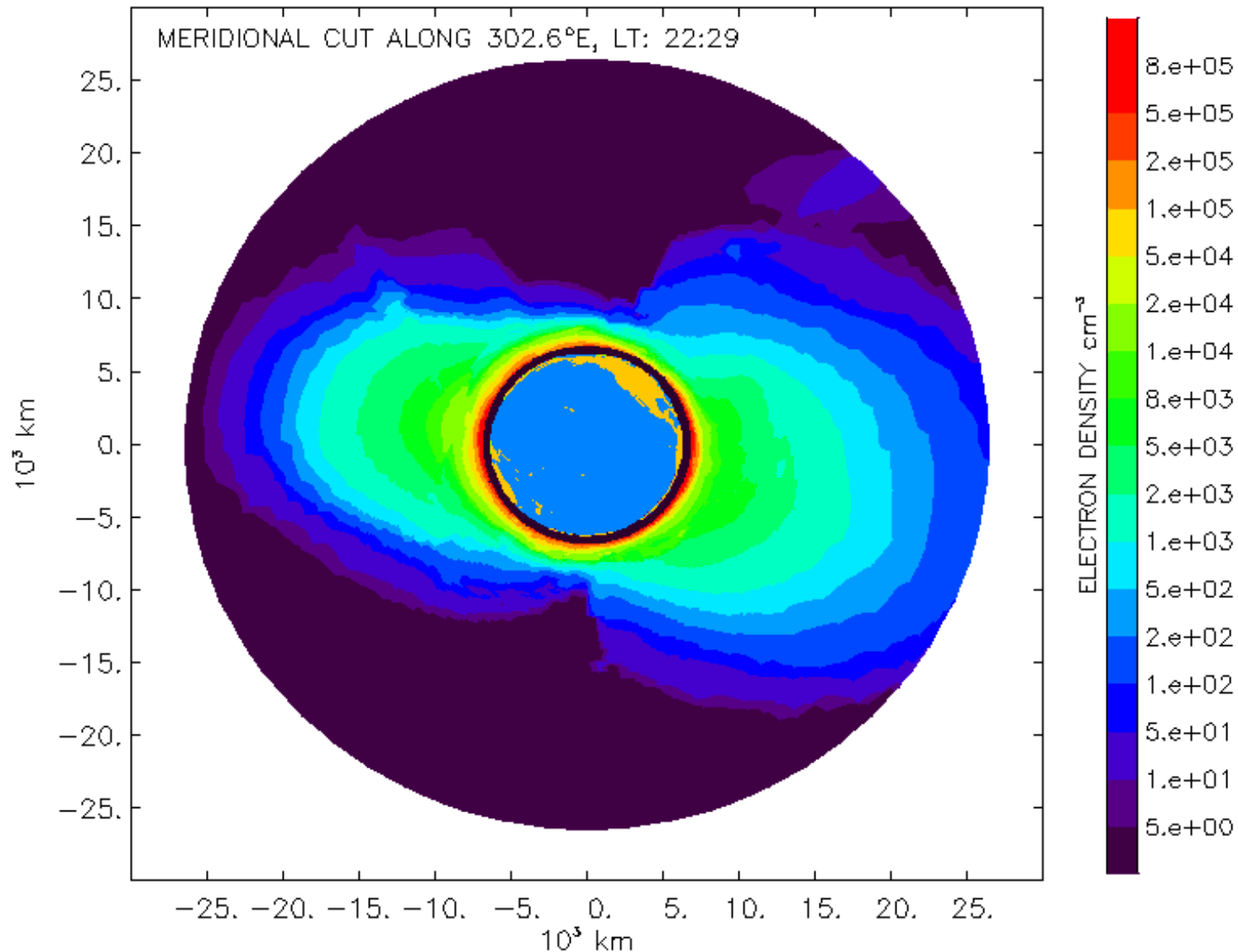


Relative Humidity

# CHAMP/GRACE GPS Ionosphere Sounding Principle



# Top-side ionosphere/plasmasphere



Electron density  
reconstruction of  
the days:

27 - 28 Oct 2002

Time resolution:

$T \approx 93$  min

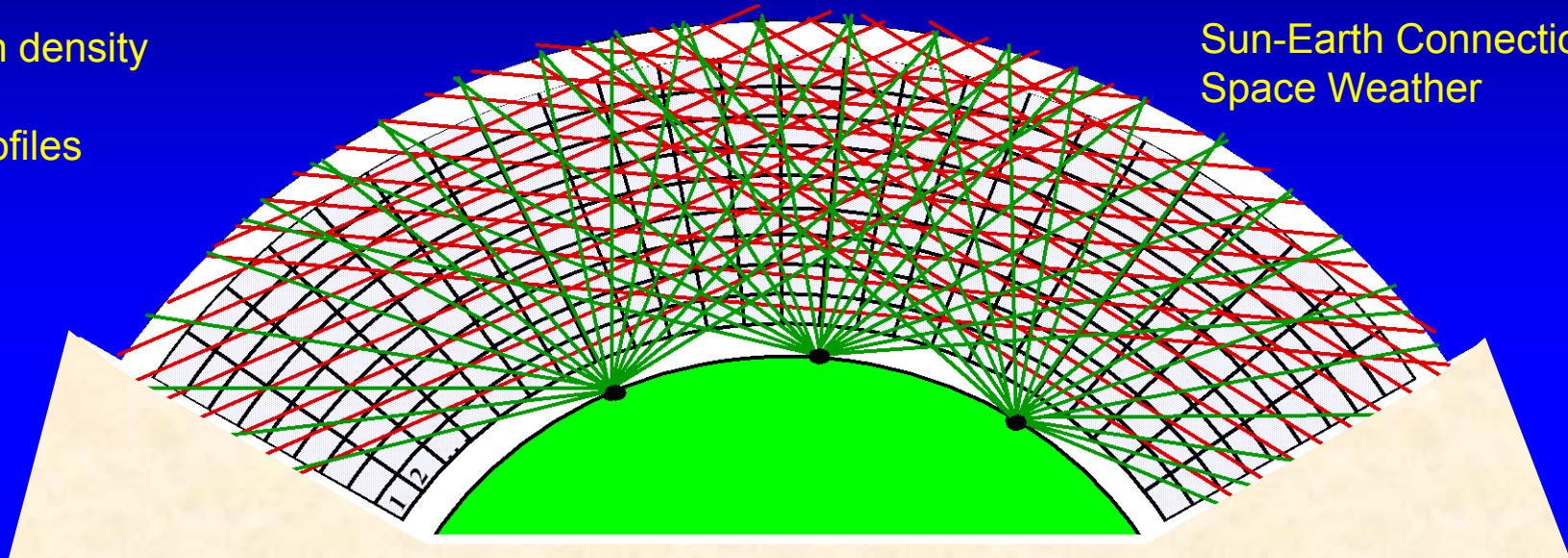
Source: Heise et al., 2004

# Key Applications of Ionospheric Data

## Snapshot 3D Ionospheric Imaging

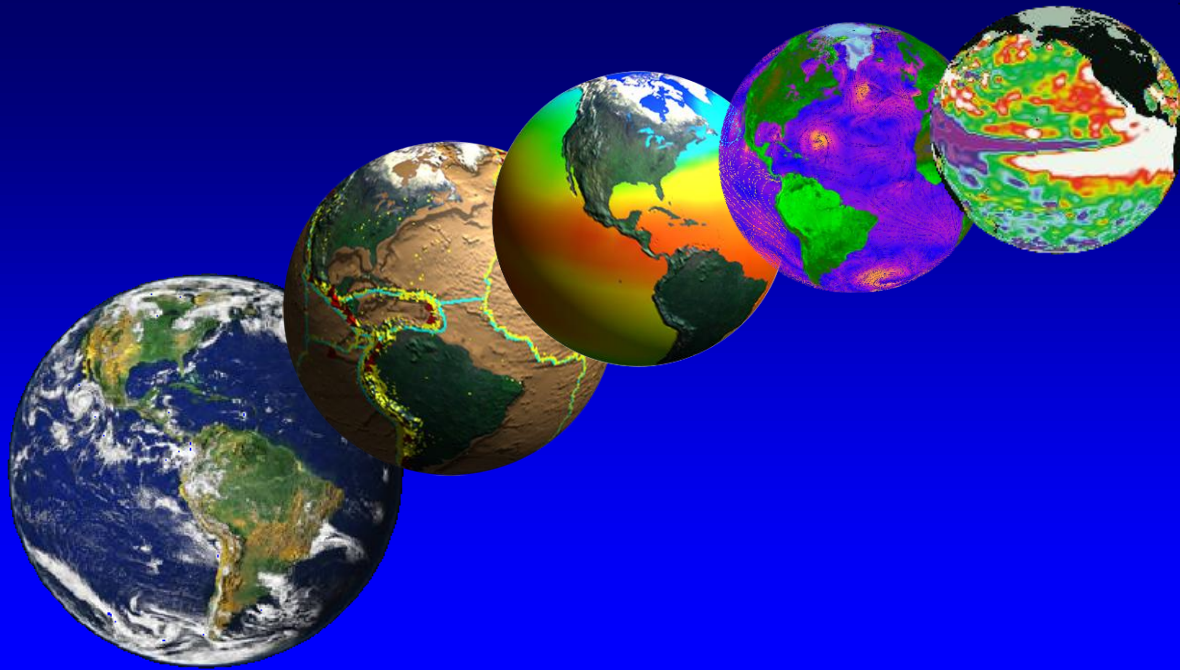
Electron density  
and  
TEC profiles

Sun-Earth Connections  
Space Weather



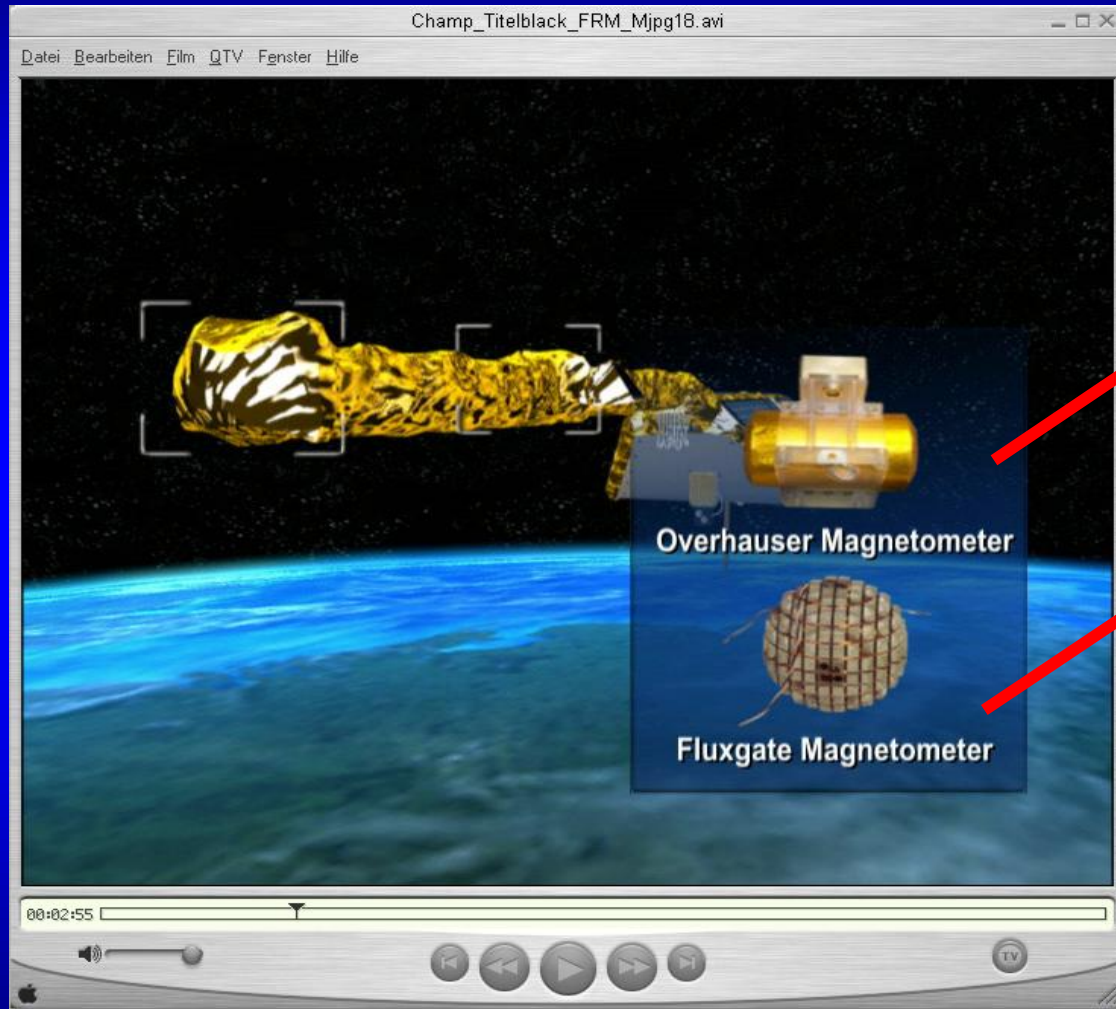
- Observe ionospheric dynamics
- 3D reconstruction of the ionosphere
- Near term predictions of space weather
- Chart full course and evolution of space storms





# CHAMP/GRACE/GOCE and Geodesy/Geophysics

# ***Magnetic field measurements with CHAMP***

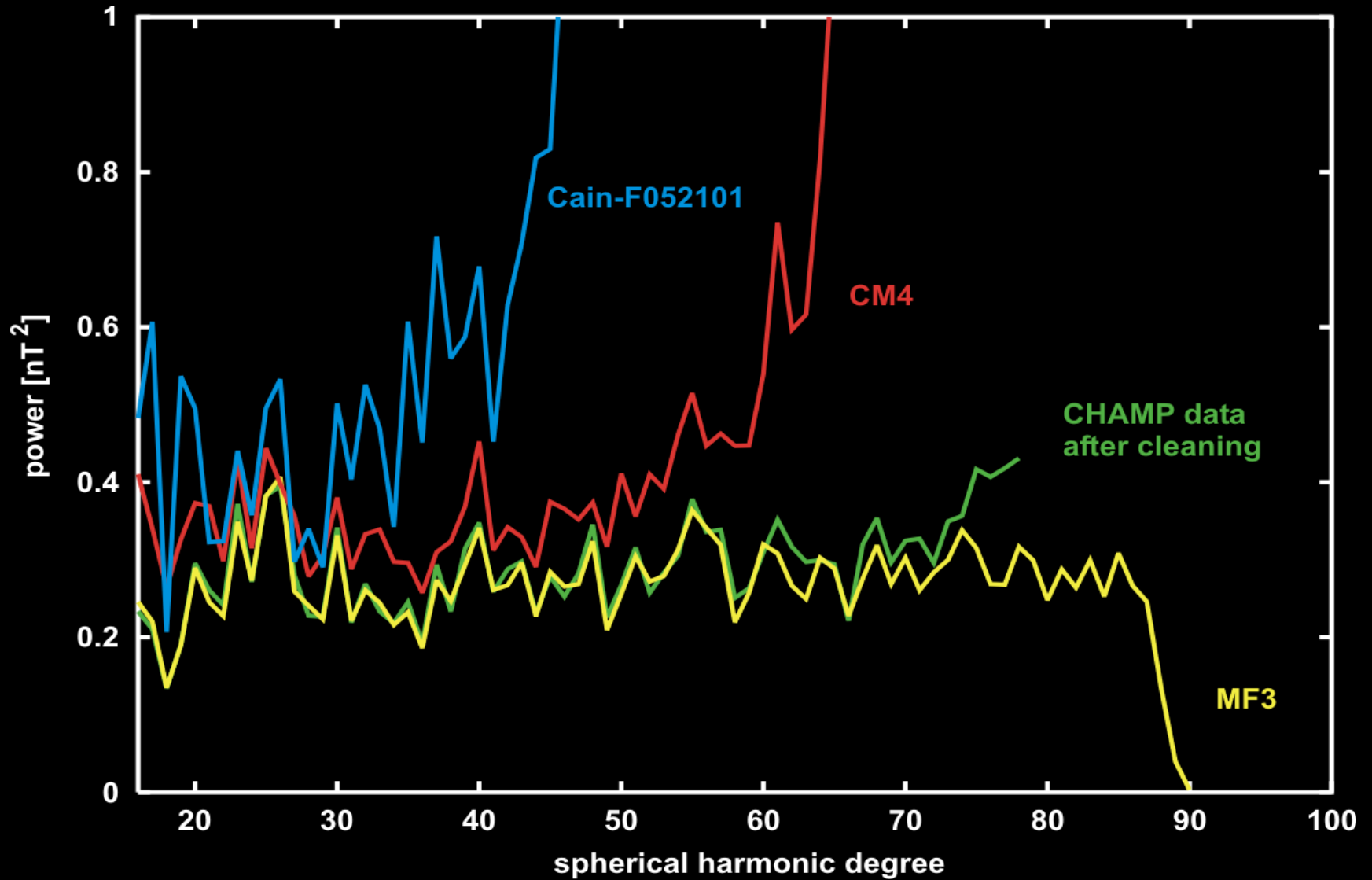


Magnetic field intensity

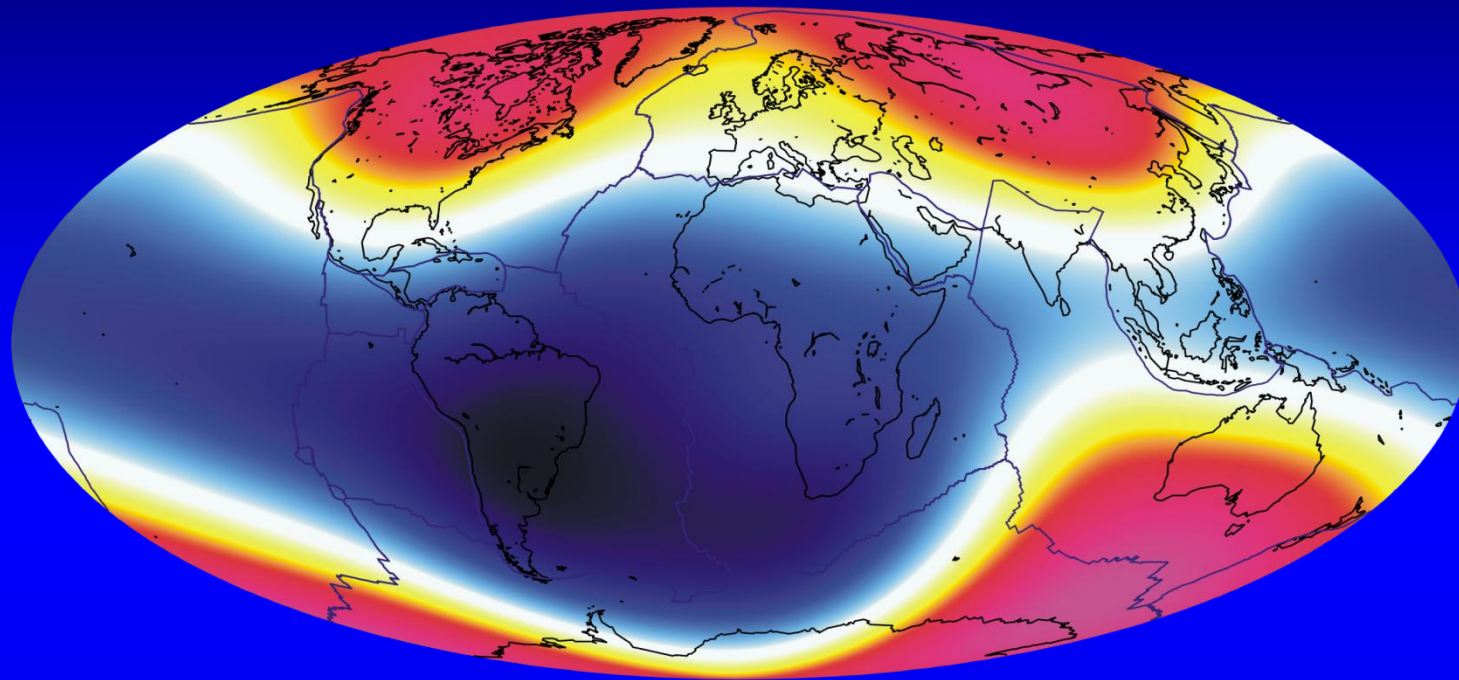
+

Components of magnetic field vector

# *Improvement since Magsat*



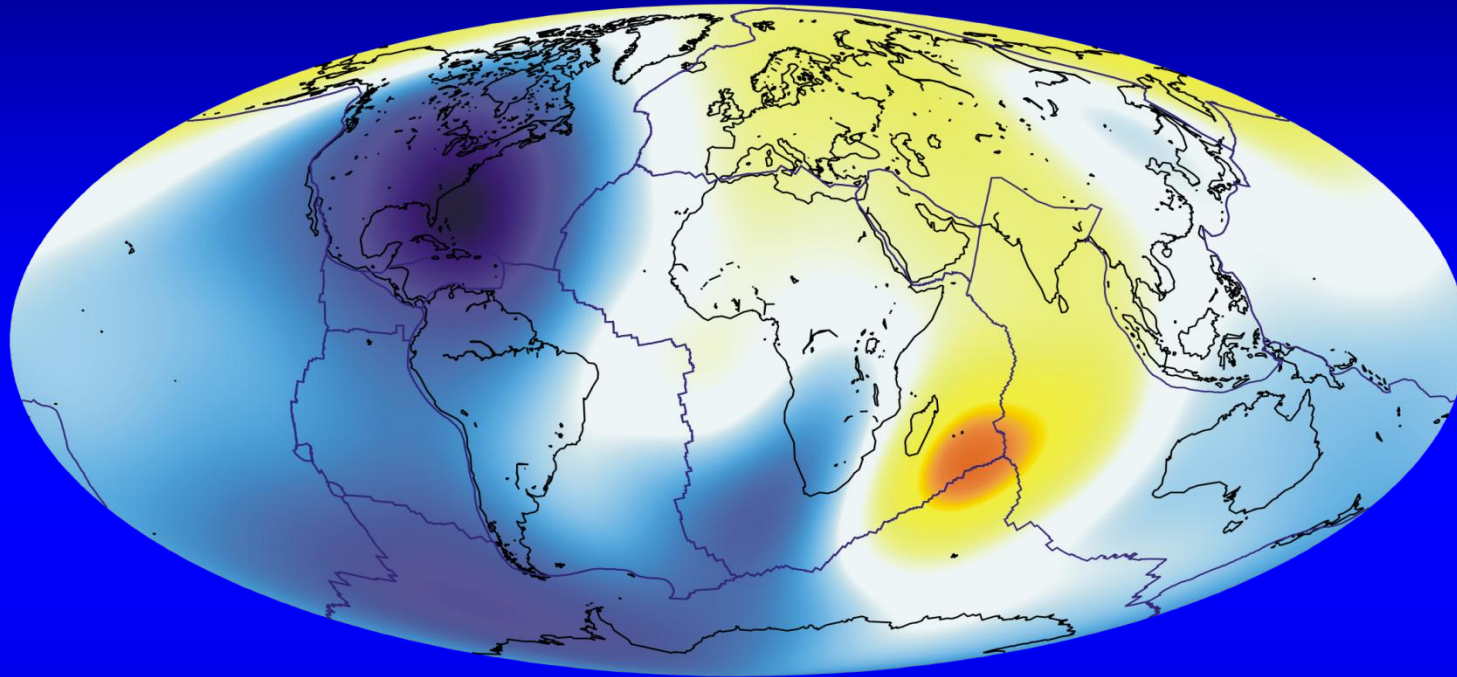
# ***CHAMP: Main Magnetic Field Strength in 2002***



Total intensity on Earth surface [nT]

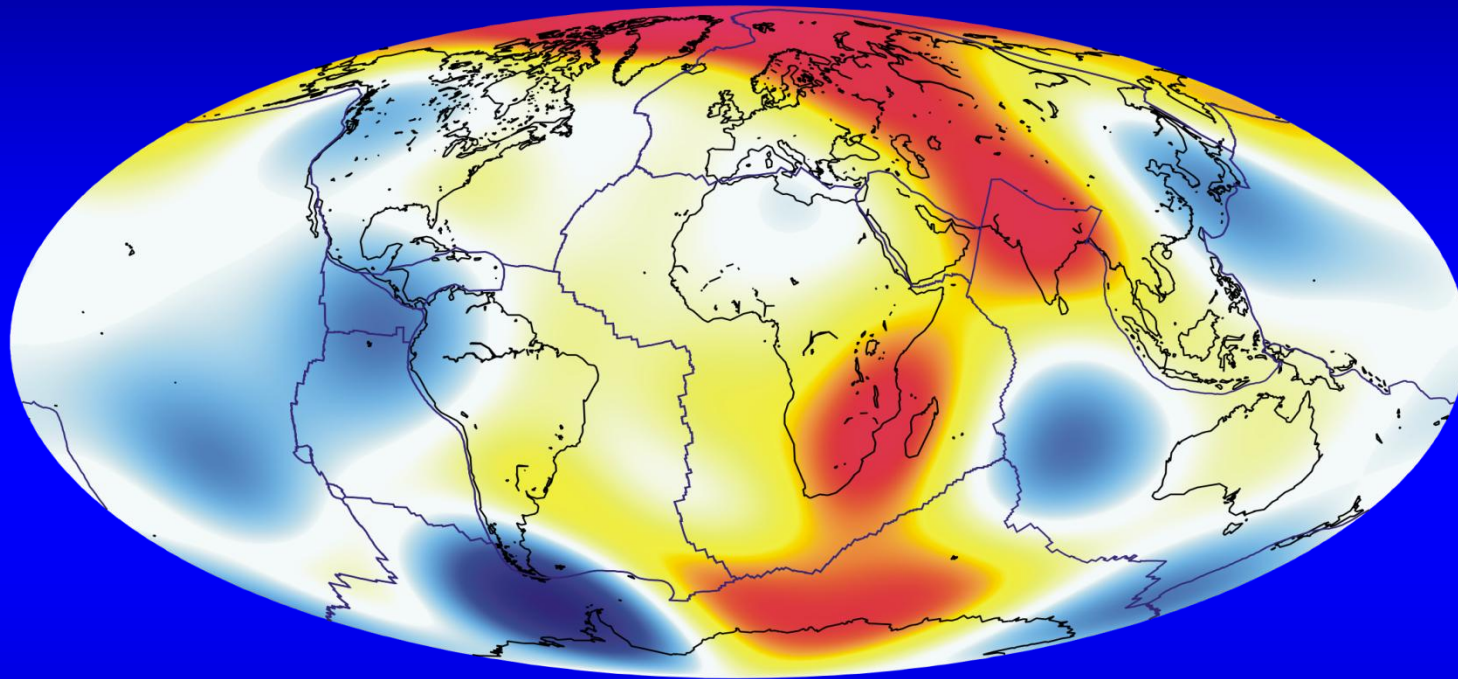


# ***CHAMP: Secular change in strength in 2002***



Change in total intensity [nT/a]

# ***CHAMP: Change in secular variation, 2002***

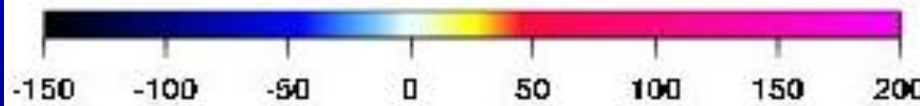
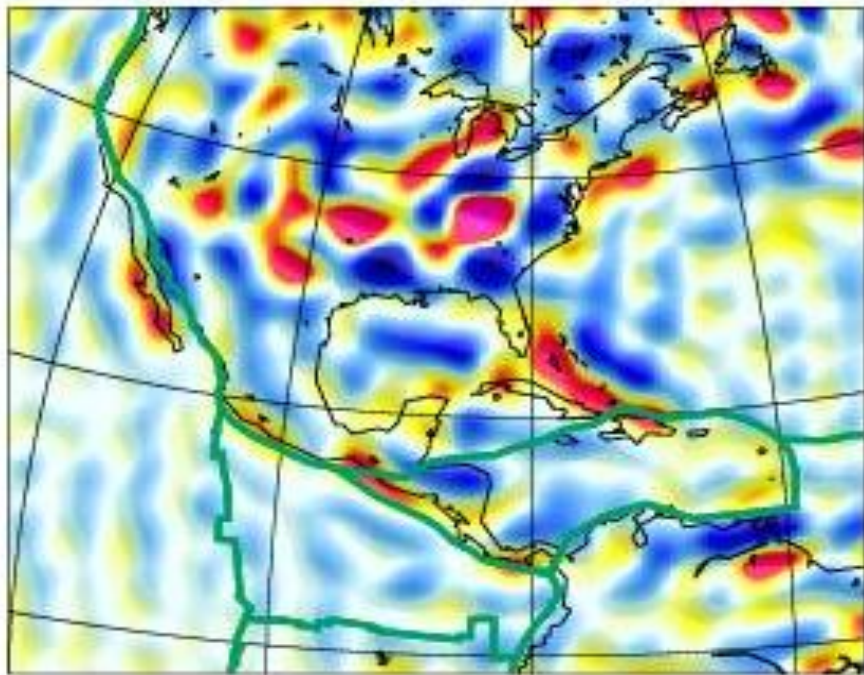


Rate of change of total intensity [nT/aa]

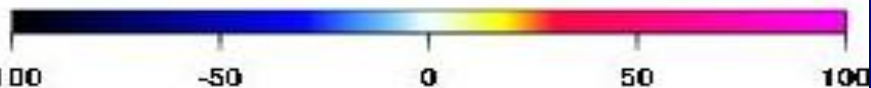
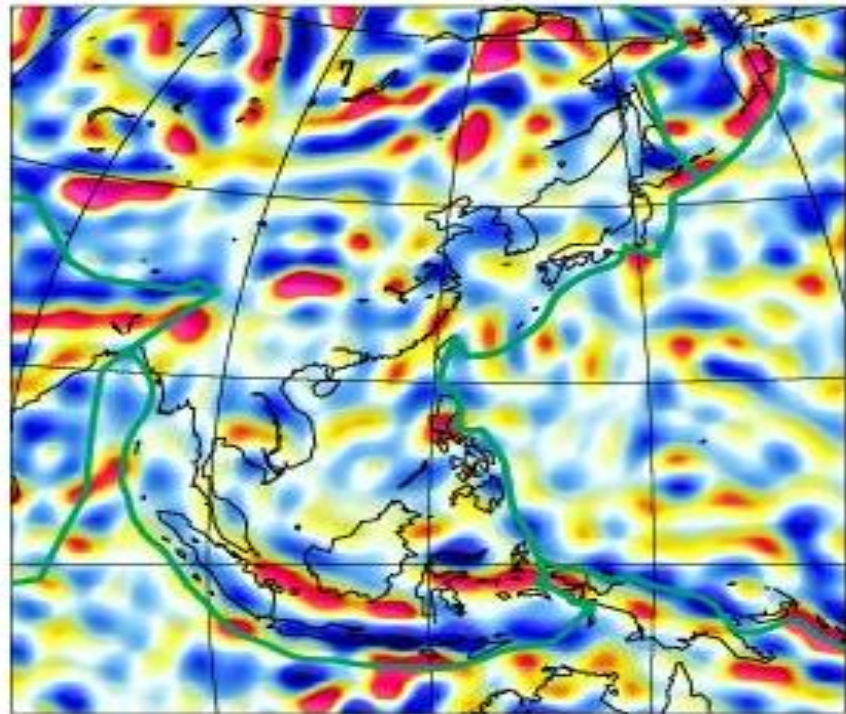


# *Model anomaly maps of $B_z$ at 50 km altitude above the reference sphere*

Mittelamerika



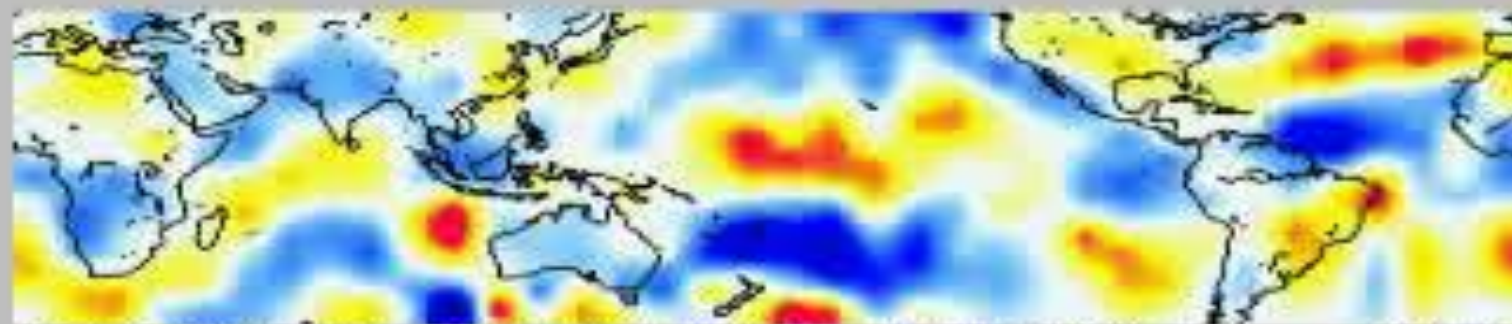
Südostasien



$B_z$  in 50 km Höhe [nT]

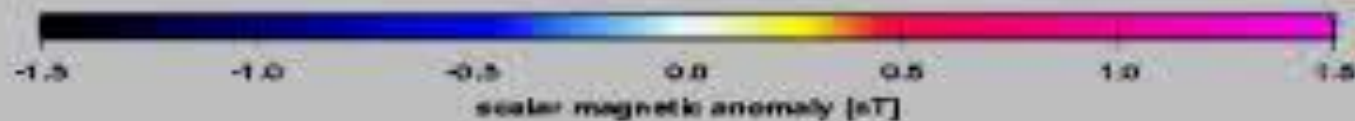
Satellite Positioning Meeting, Wrocław University

# *M2 Tide induced Magnetic Field Variations Observed by CHAMP Satellite*

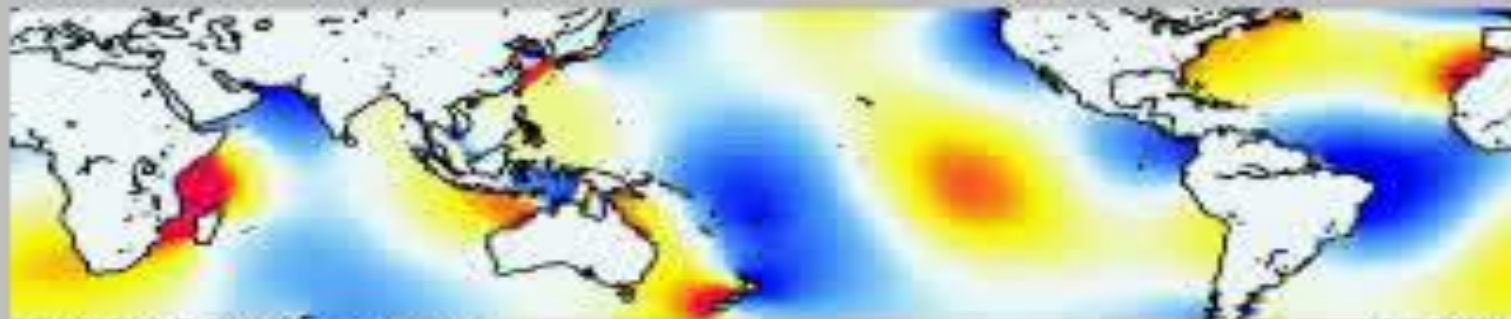


Satellite: CHAMP data period: 1-Aug-00 to 24-Jun-02 mean altitude: 400 km

S.M., GFZ

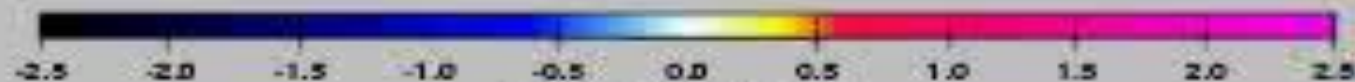


scalar magnetic anomaly [nT]



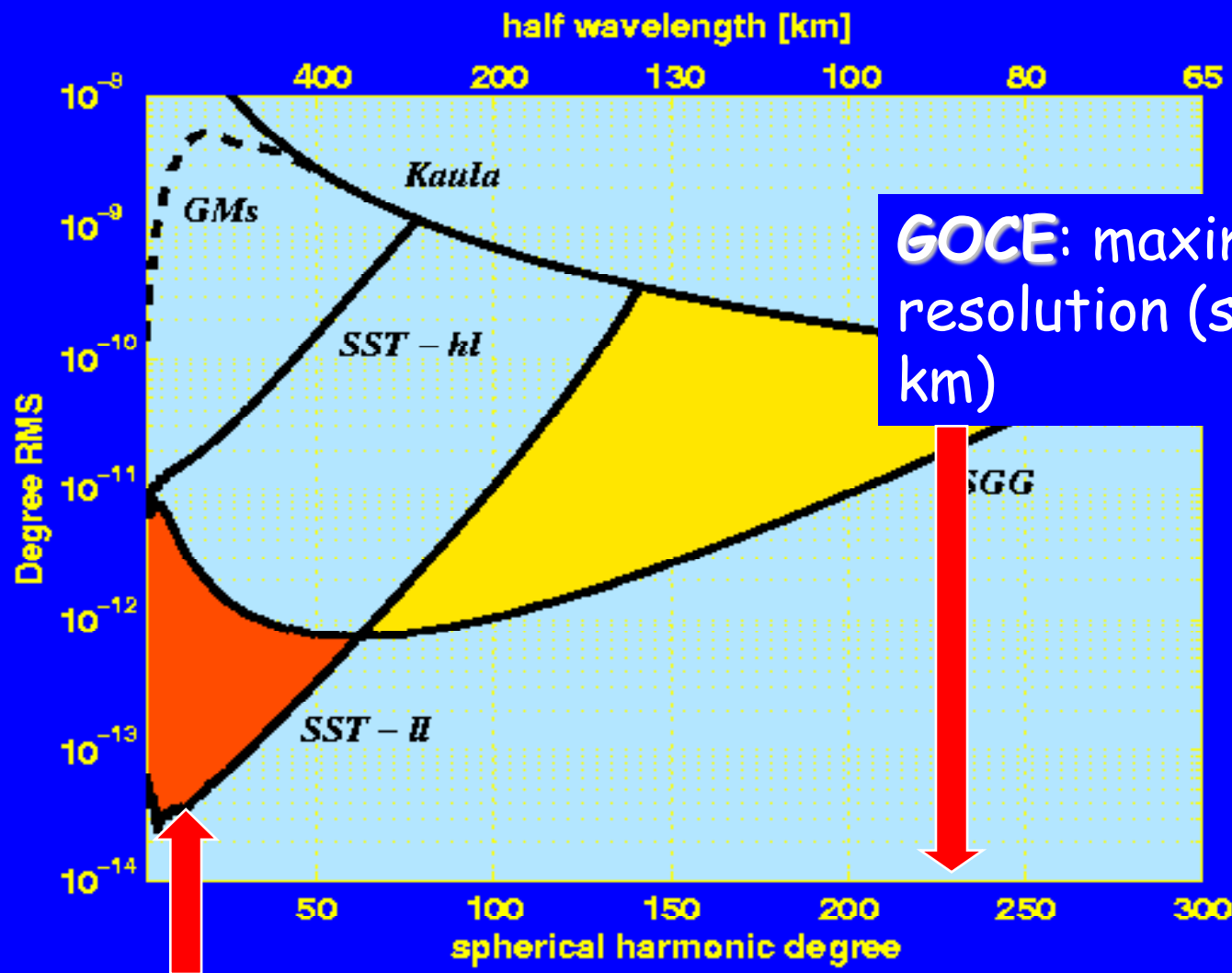
M2 ocean tide, model TPXO.5.2

G.E., OSU



sea surface height [m]

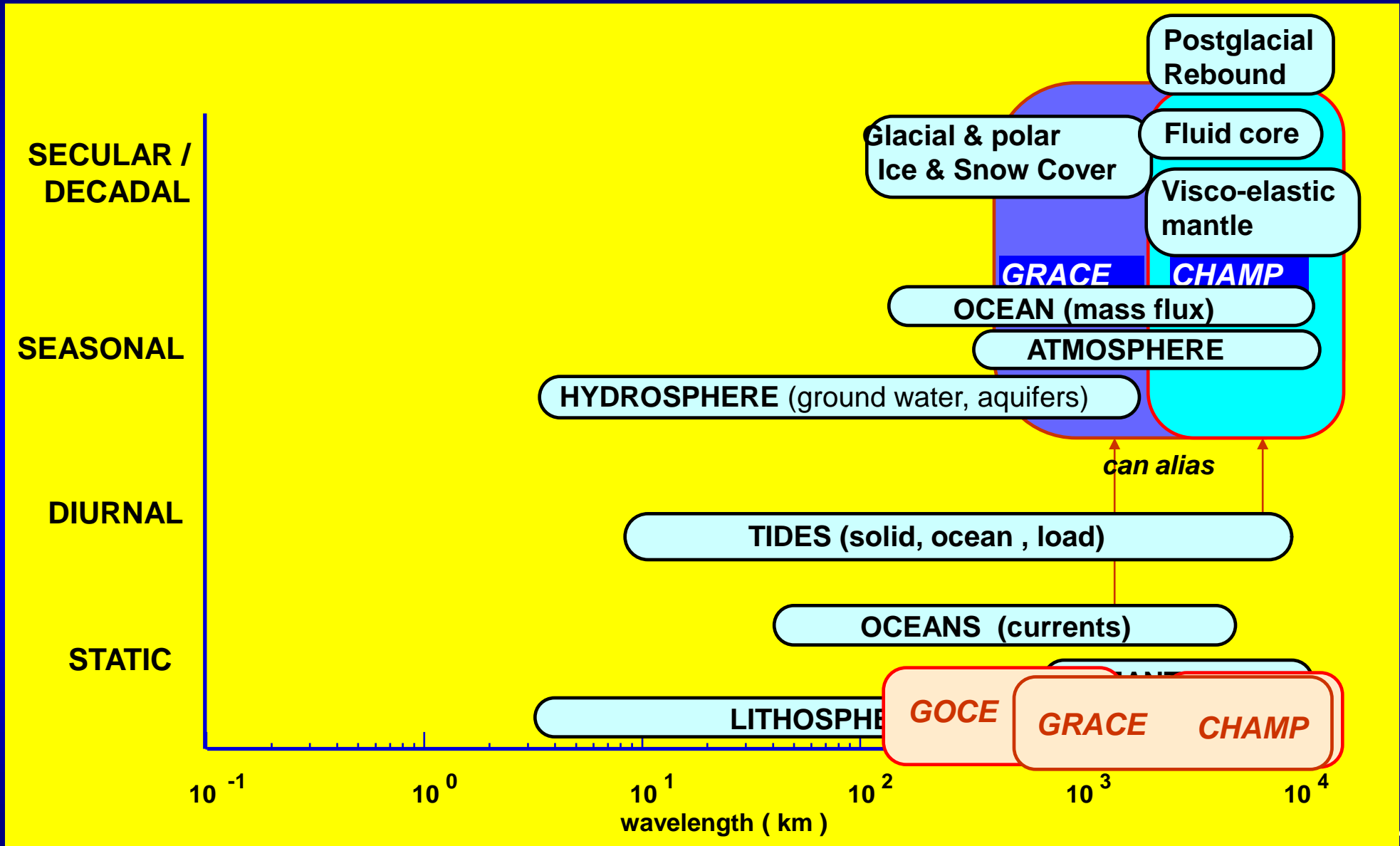




GOCE: maximum resolution ( $s = 80$  km)

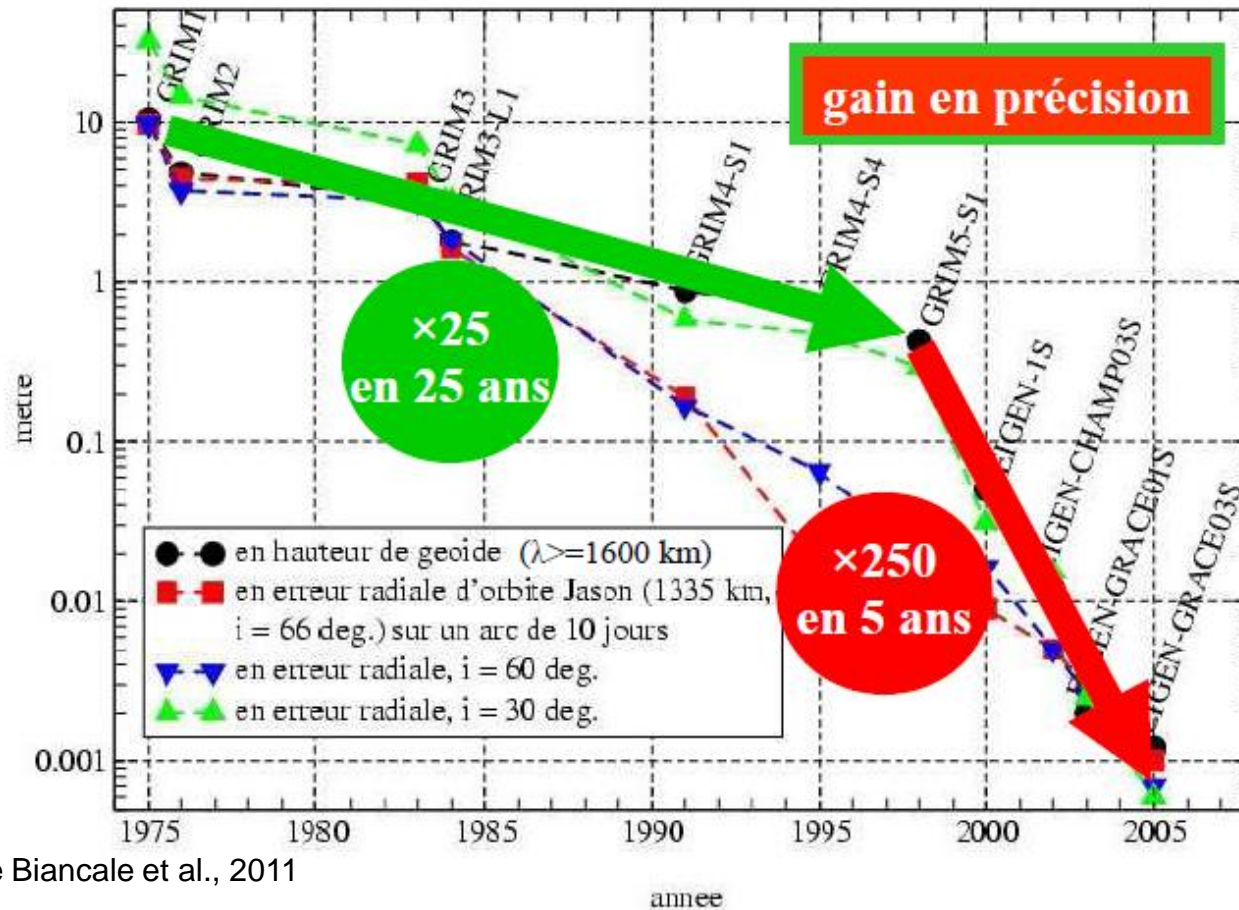
GRACE: maximum precision (geoid  $< \mu\text{m}$ )

# Signals Producing Temporal Gravity Changes



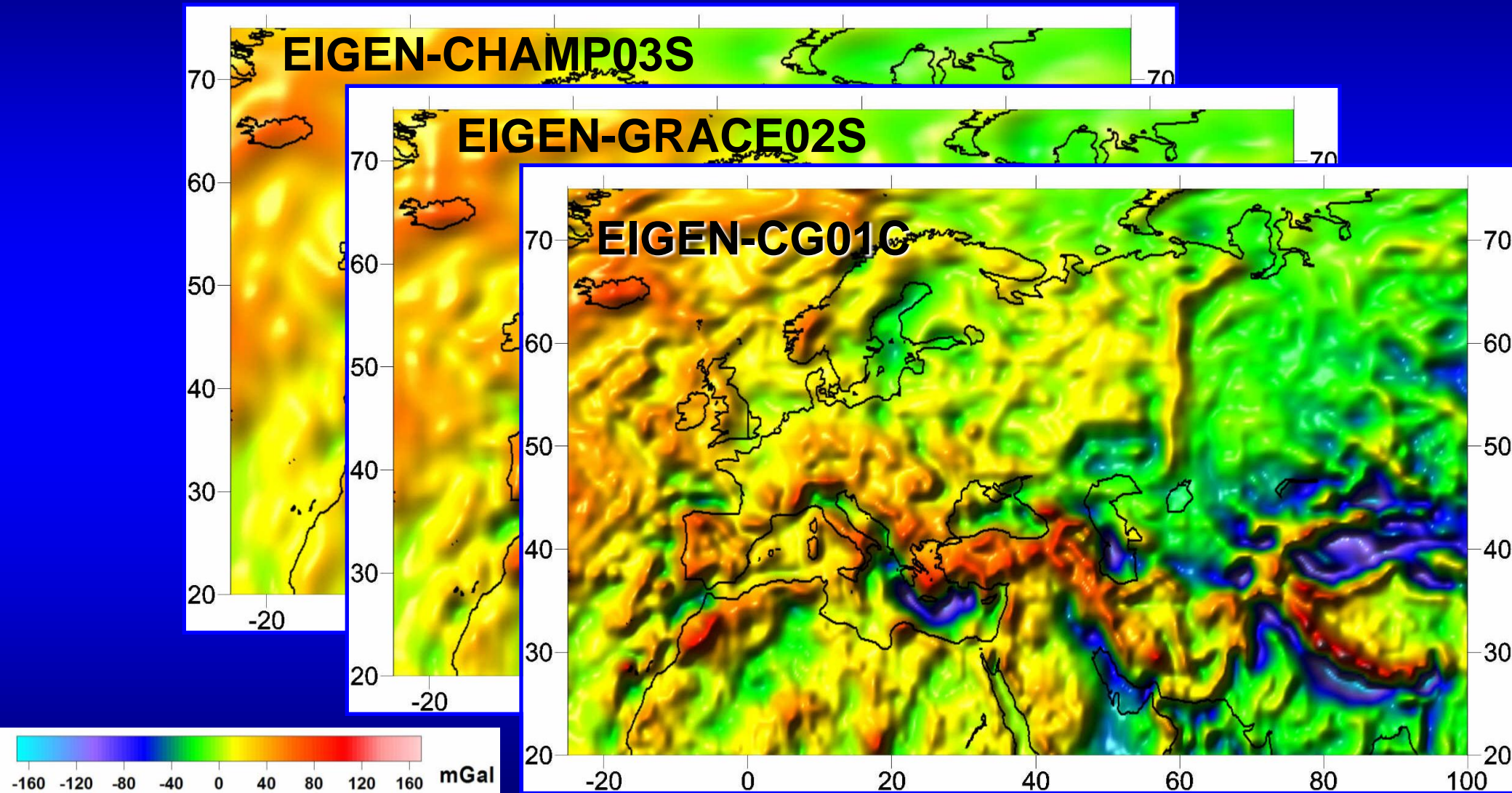
# GFZ/GRGS gravity model improvements over time (GRIM and EIGEN models)

(par différence au modèle EIGEN-GL04S)



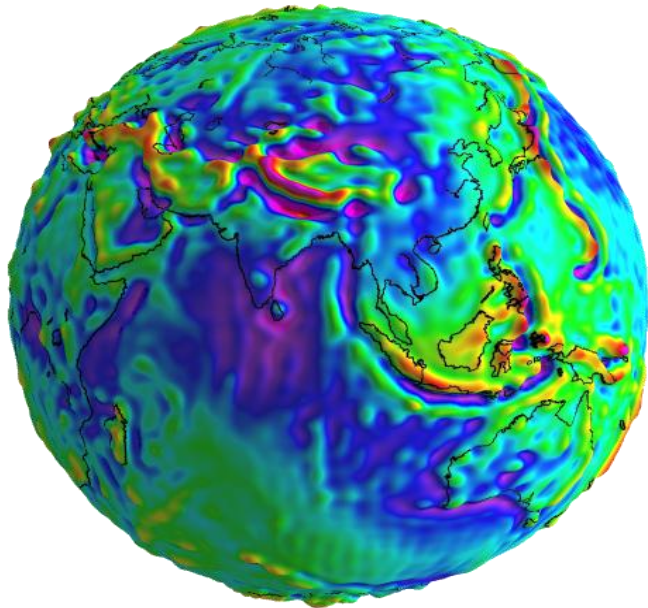
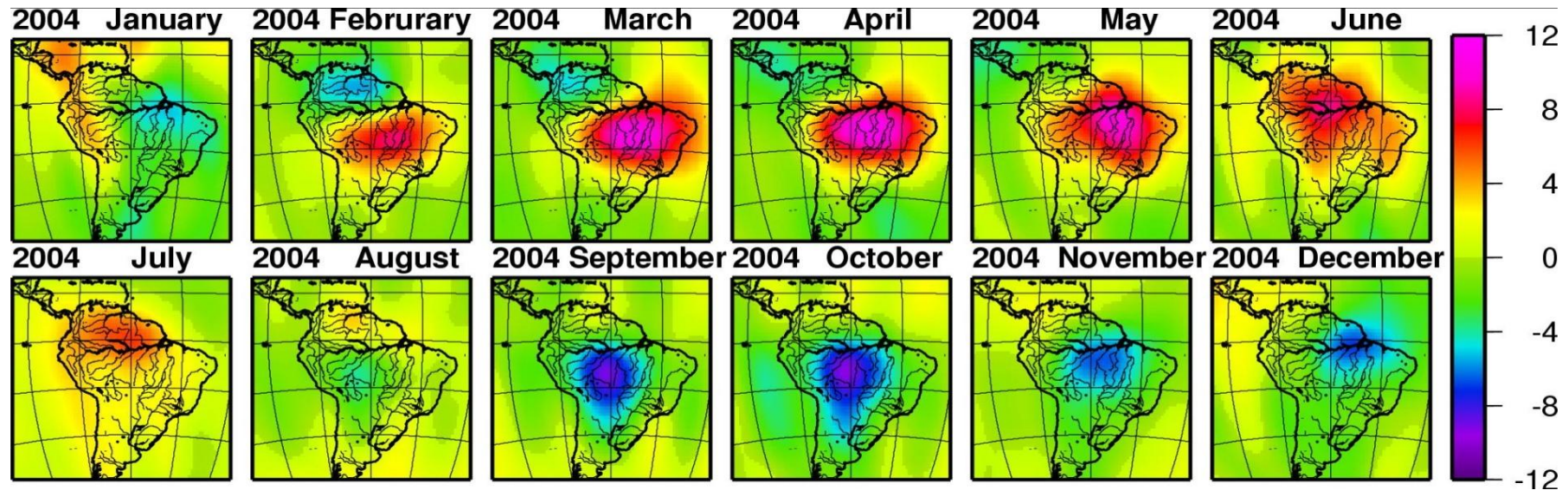
Source Biancale et al., 2011

# Gravity over Europe as Seen by Different Missions





# GRACE Gravity Field Models



- 100+ monthly gravity field models between April 2002 and December 2010 with 167/333 km resolution
- Static (mean) satellite-only and combined (with terrestrial data) gravity models with 55km resolution
- GRACE gravity fields provide information on mass transport and mass distribution in the system Earth → breakthroughs in the understanding of changes in Earth system components

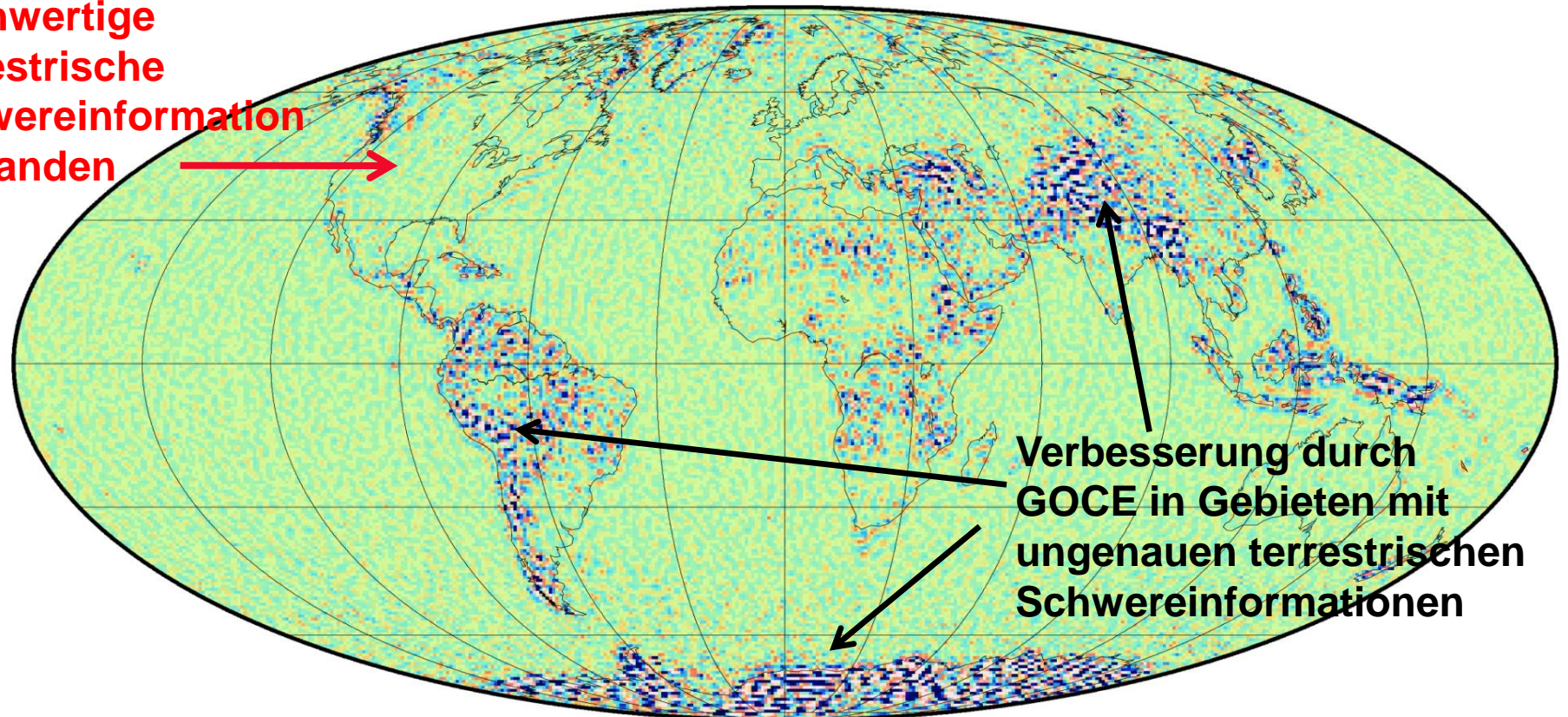


# First GOCE Results (Bergen, ESA EO Symposium)

Förste et al., 2010

Geoid height differences EIGEN-GOCE14p und EIGEN-5C (GRACE)

Hochwertige  
Terrestrische  
Schwereinformation  
vorhanden

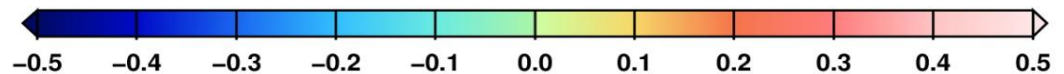


Verbesserung durch  
GOCE in Gebieten mit  
ungenauen terrestrischen  
Schwereinformationen

EIGEN-5C vs. EIGEN-GOCE-14p

$\zeta$ ,  $0.75^\circ \times 0.75^\circ$

wrms about mean / min / max = 0.2043 / -3.328 / 4.226 meter



# Combination scheme of EIGEN-6C

Accumulation of a full normal matrix up to d/o 370:

~200.000 parameters, ~ 250 GByte

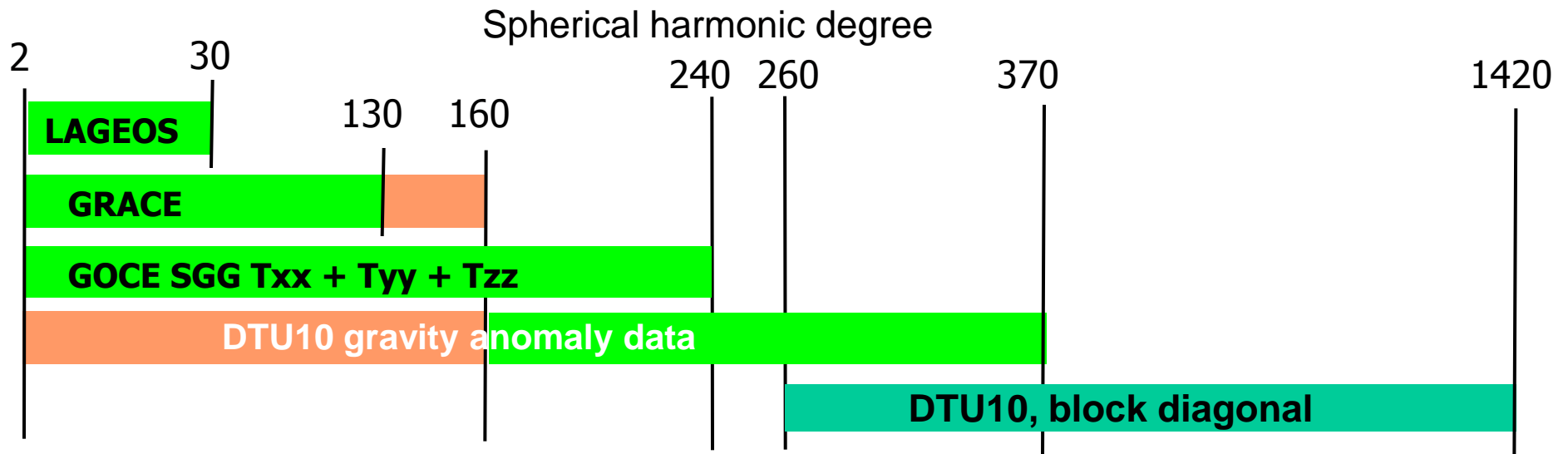
contribution to the solution:



kept separately:

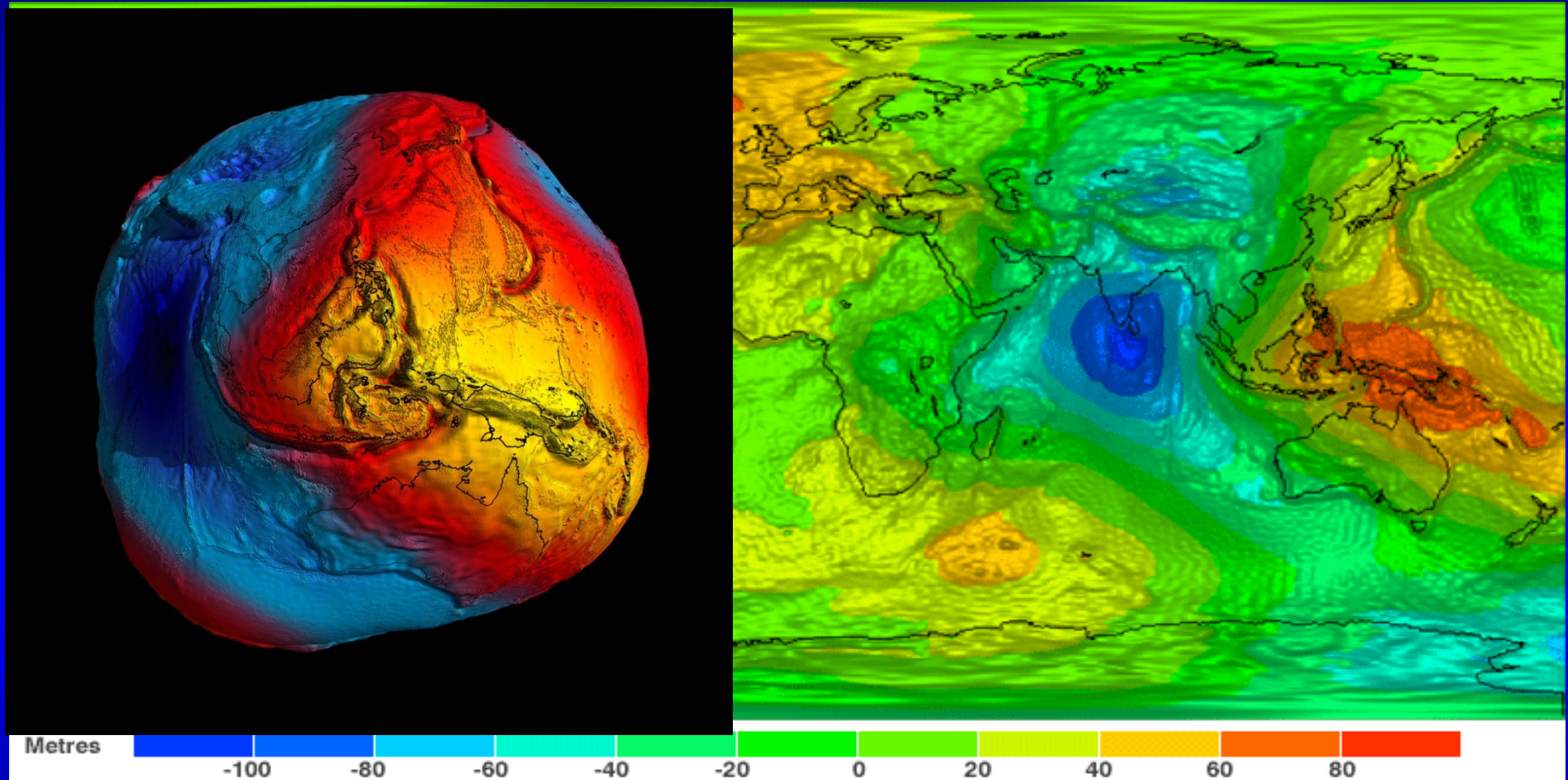


Separate block diagonal solution:





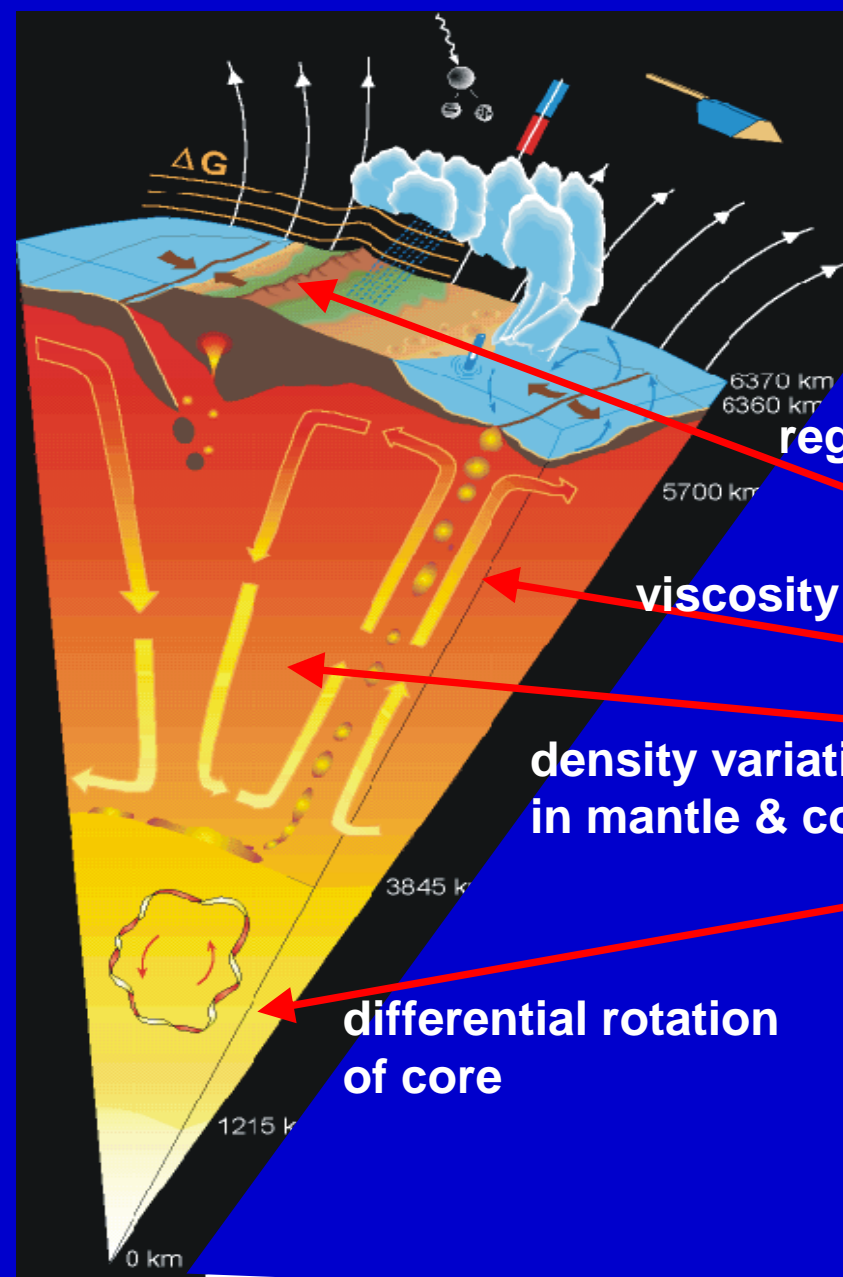
# *Recent Results from GOCE Symposium/Workshop*



GOCE Geoid



# ***CHAMP-GRACE-GOCE & Solid Earth Physics***



**regional structures & post-glacial rebound**

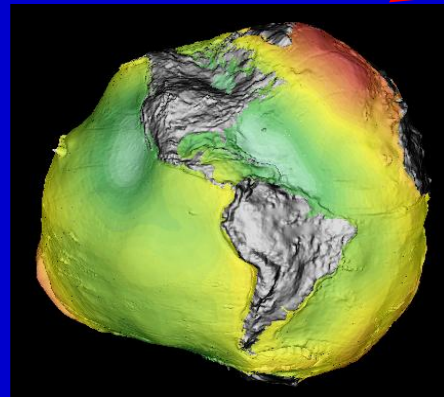
**viscosity distribution & mantle convection**

**density variations  
in mantle & core**

**differential rotation  
of core**

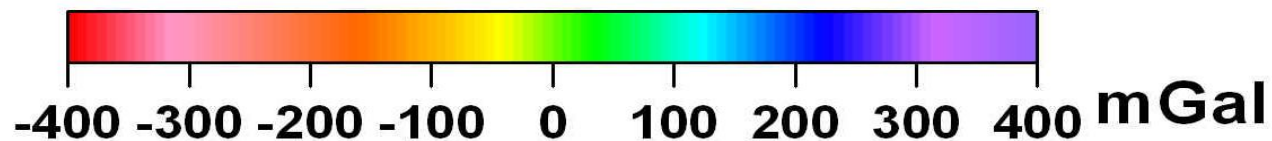
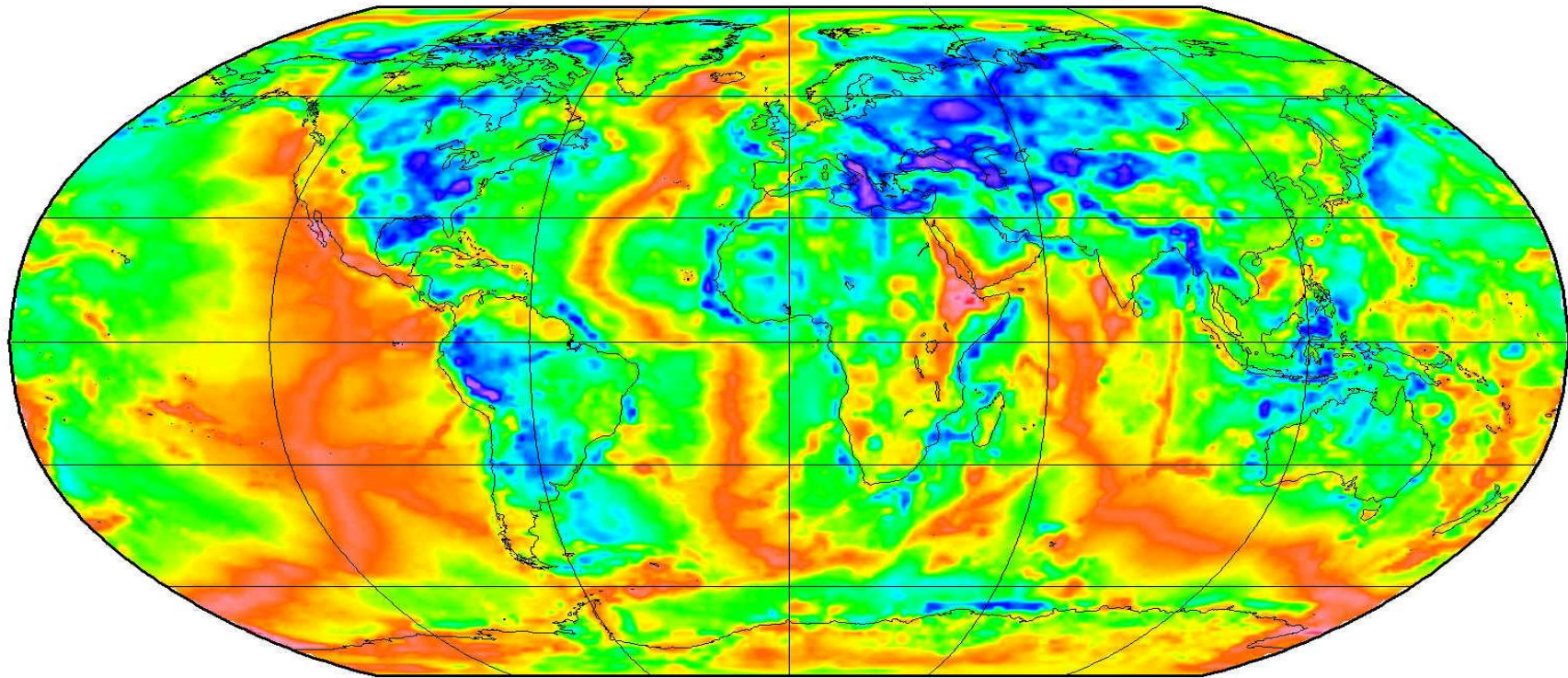
**GRACE/GOCE  
mission impact**

**precision geoid**



***Mantle Gravity Anomalies Derived from GRACE Combination Solution  
A Contribution to Modeling the Earth Interior***

**Mantle Gravity Anomalies  
(crust effect is removed)**



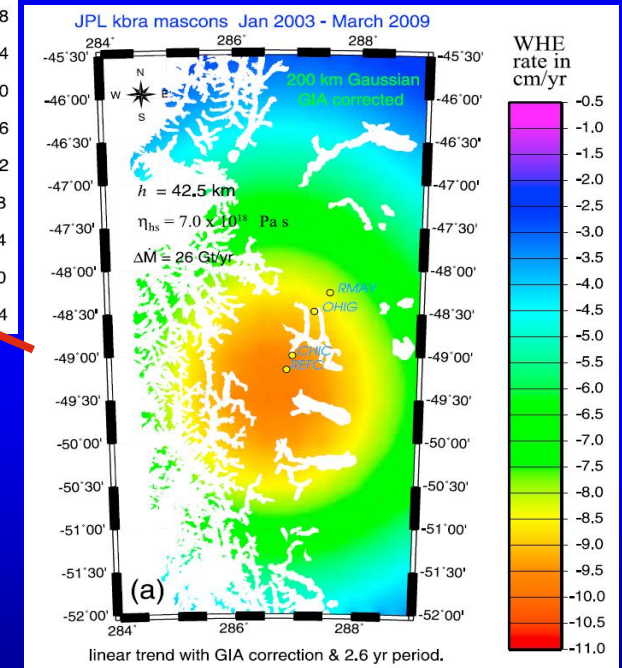
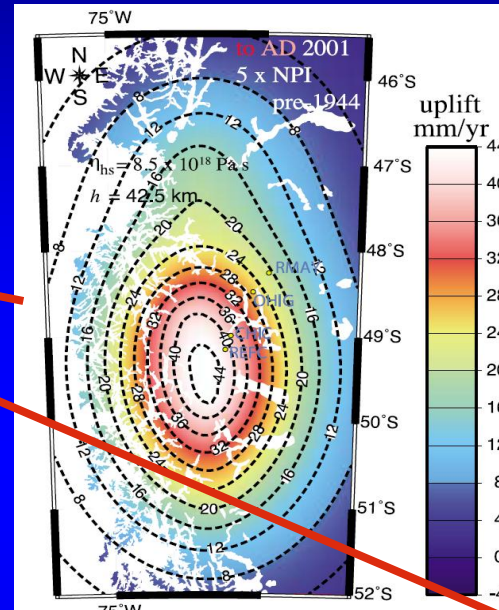
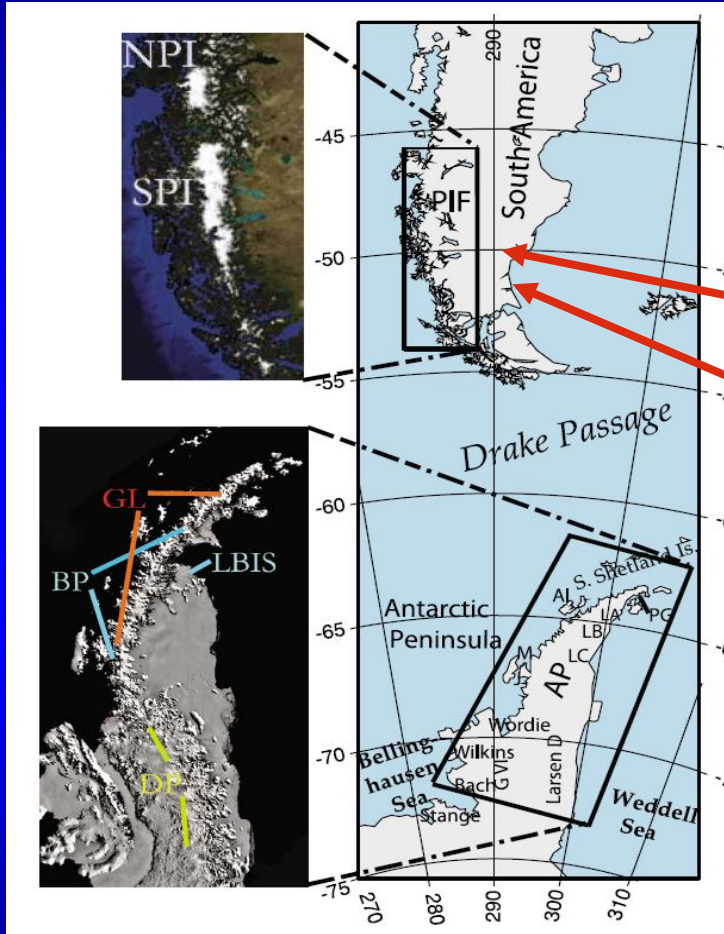


# Recent results from the GRACE mission

## - solid Earth -

### Glacial isostatic adjustment (response to past melting) from GRACE

ICE5G/IJ05/VM2

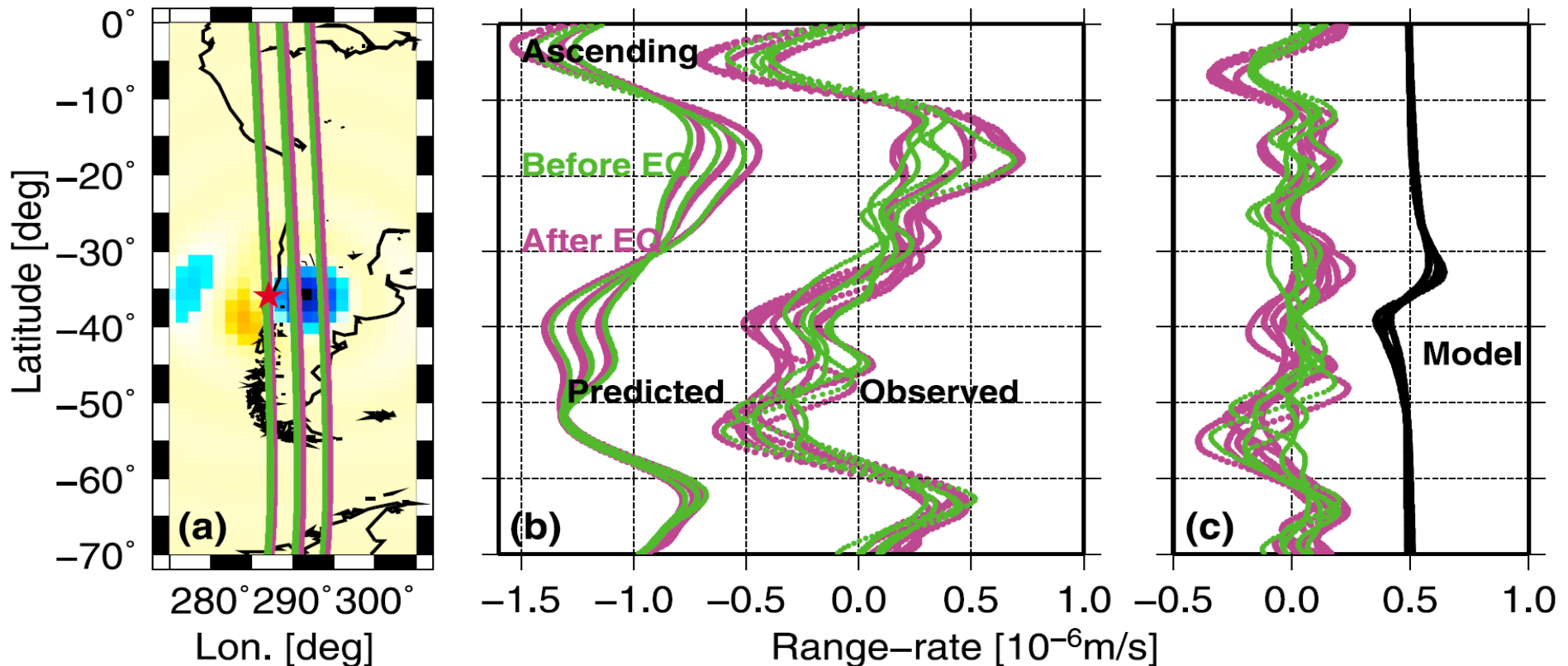


Ivins et al. in JGR (2011) reconstruct glacier ice mass loss and GIA effect by adjusting regional Earth parameters (lithosphere thickness, viscosity) to GRACE and GPS uplift.

# Recent results from the GRACE mission

- solid Earth -

## Mass dislocation by large Earthquakes (Chile M8.8 2010)

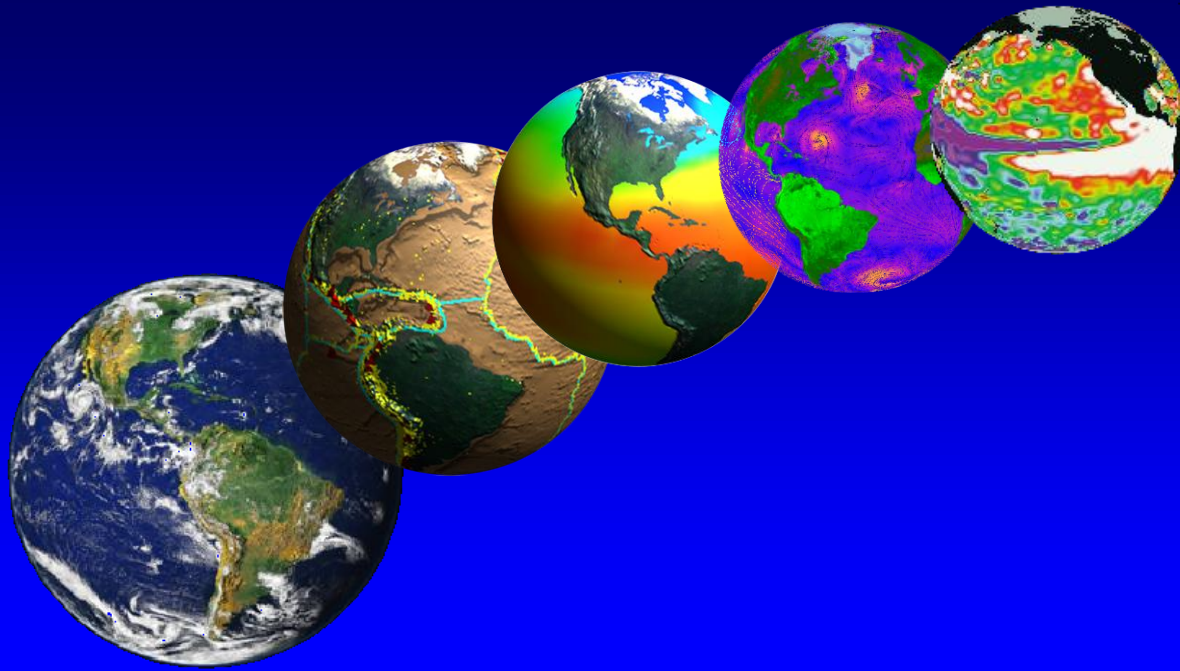


GRACE observes coseismic gravity effect of the M8.8 2010 Chile Earthquake.

According to **Han et al (2010, GRL)**, different fault models may be testable

→ Recent papers Panet and Einarsson on postseismic gravity change Sumatra EQ

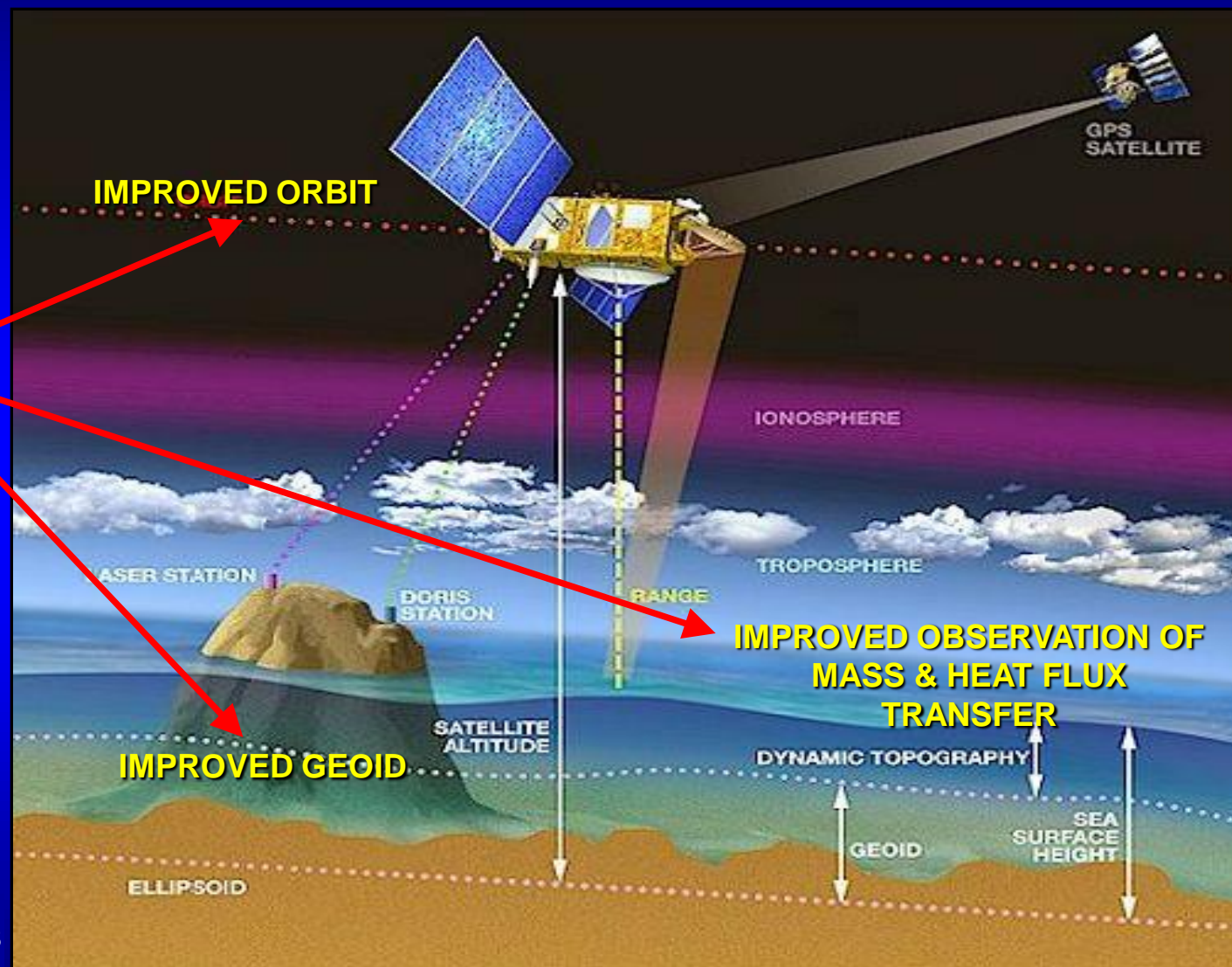




# GRACE/GOCE and Oceanography

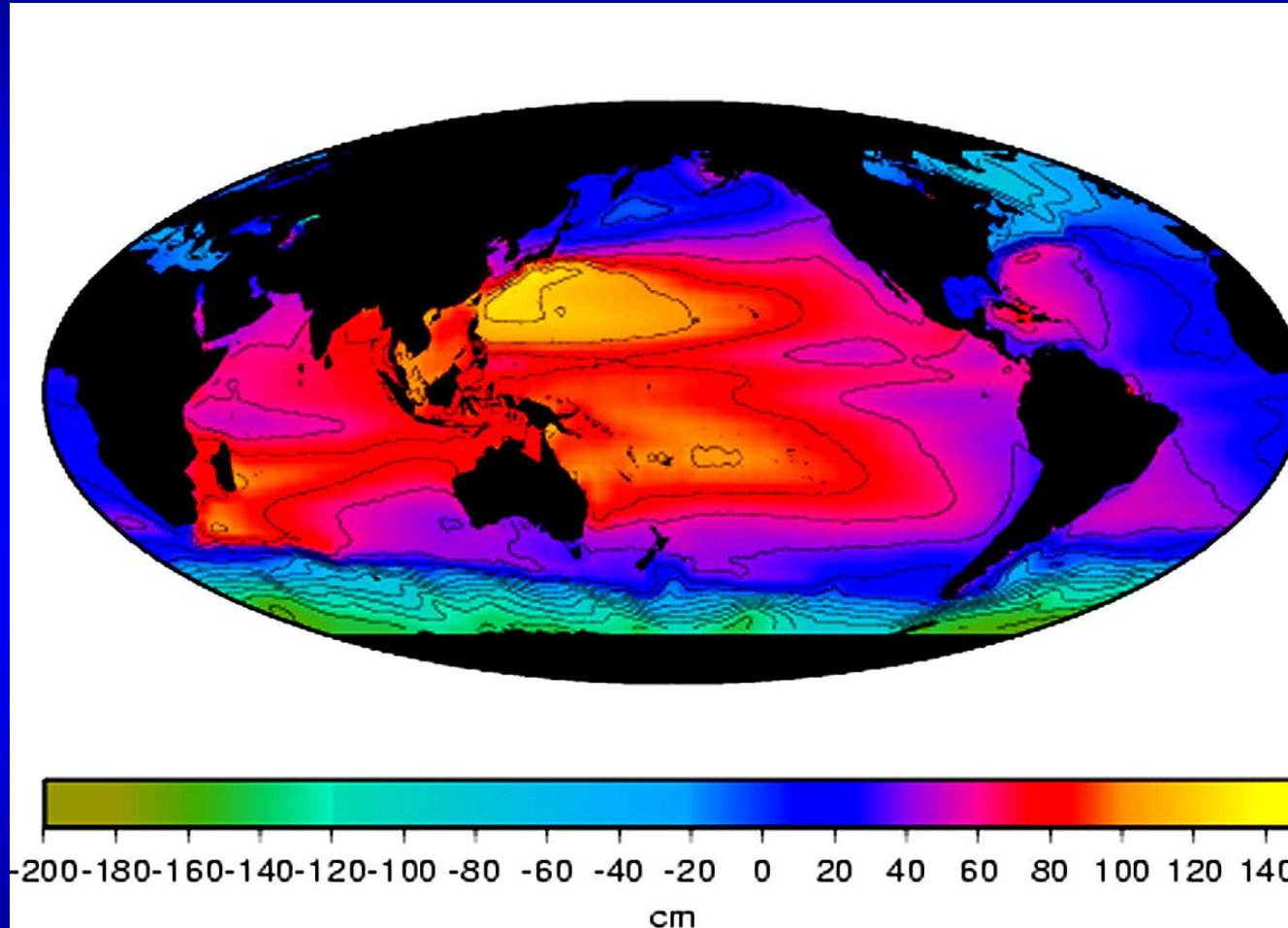
# Gravity Missions and Ocean Dynamics

**GRACE&GOCE  
missions impact**



Source: CNES

# *Dynamic Ocean Topography from GRACE Gravity Model*



The dynamic ocean topography is the difference between the mean sea surface (observed from altimeter data) and the geoid.

This difference is caused by the ocean currents.

With no currents, the ocean surface would coincide with the geoid.

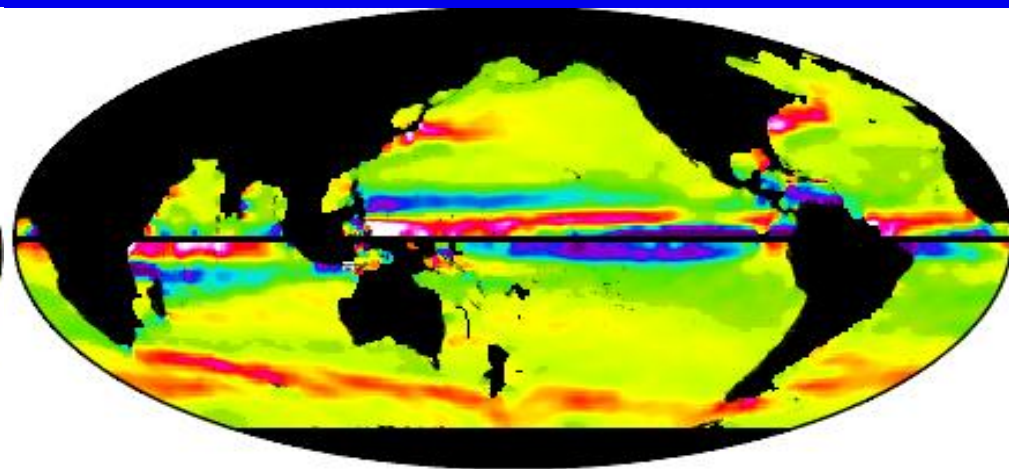
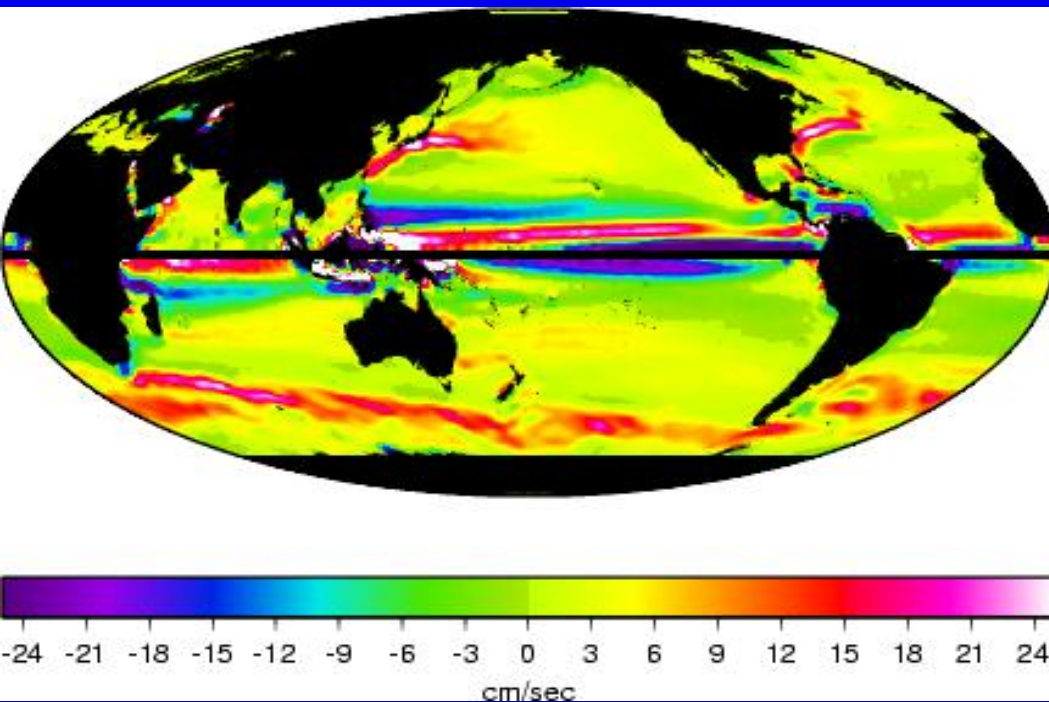


# *Zonal Circulation Estimates from GRACE and GOCE*

East-west currents are much more clearly seen with a GRACE gravity model than with previous models and even better with GOCE models.

Altimeter MSS with GRACE gravity model

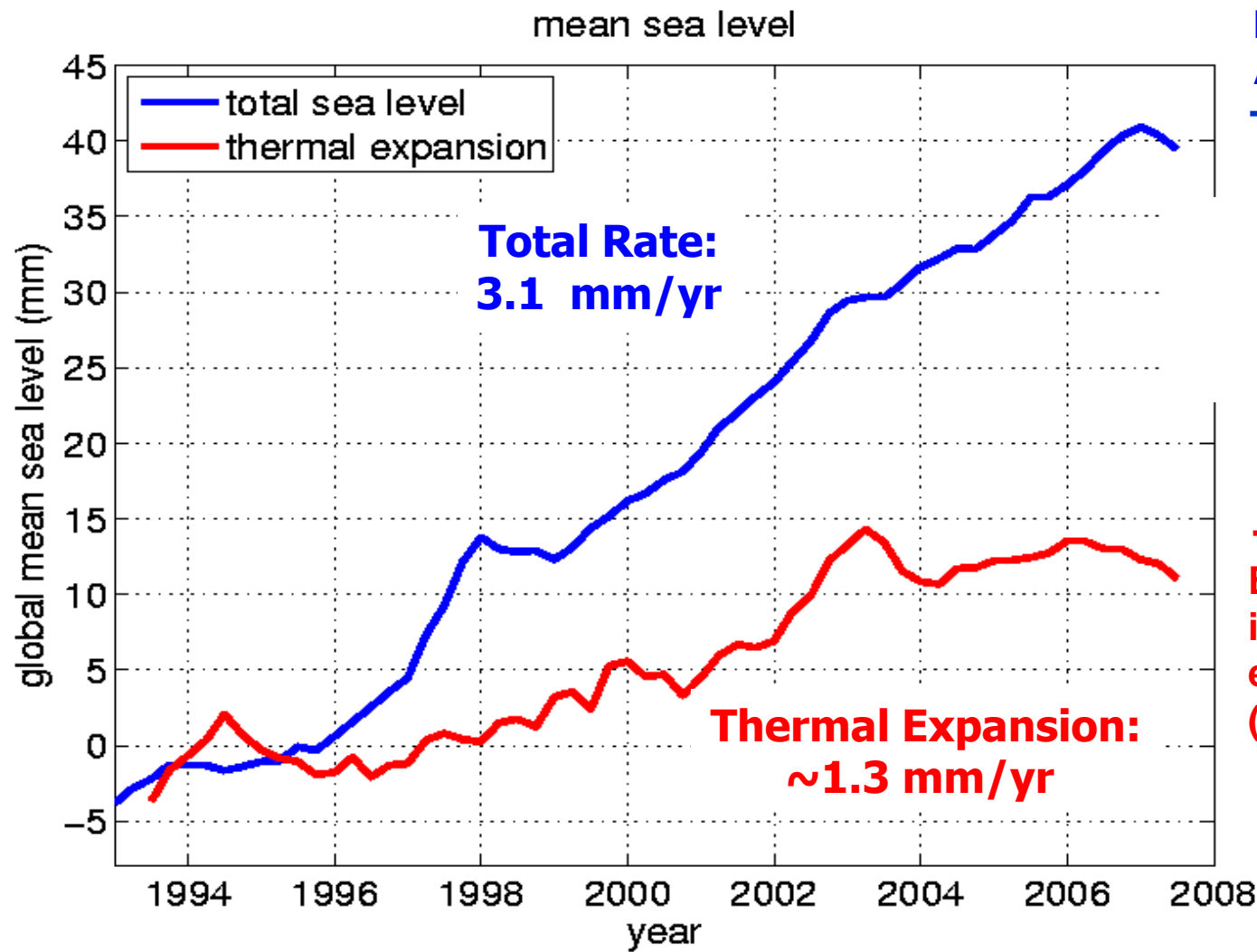
WOA01 Hydrography



Tapley, et al, GRL, 2004



# Sea level change: dM/dV components

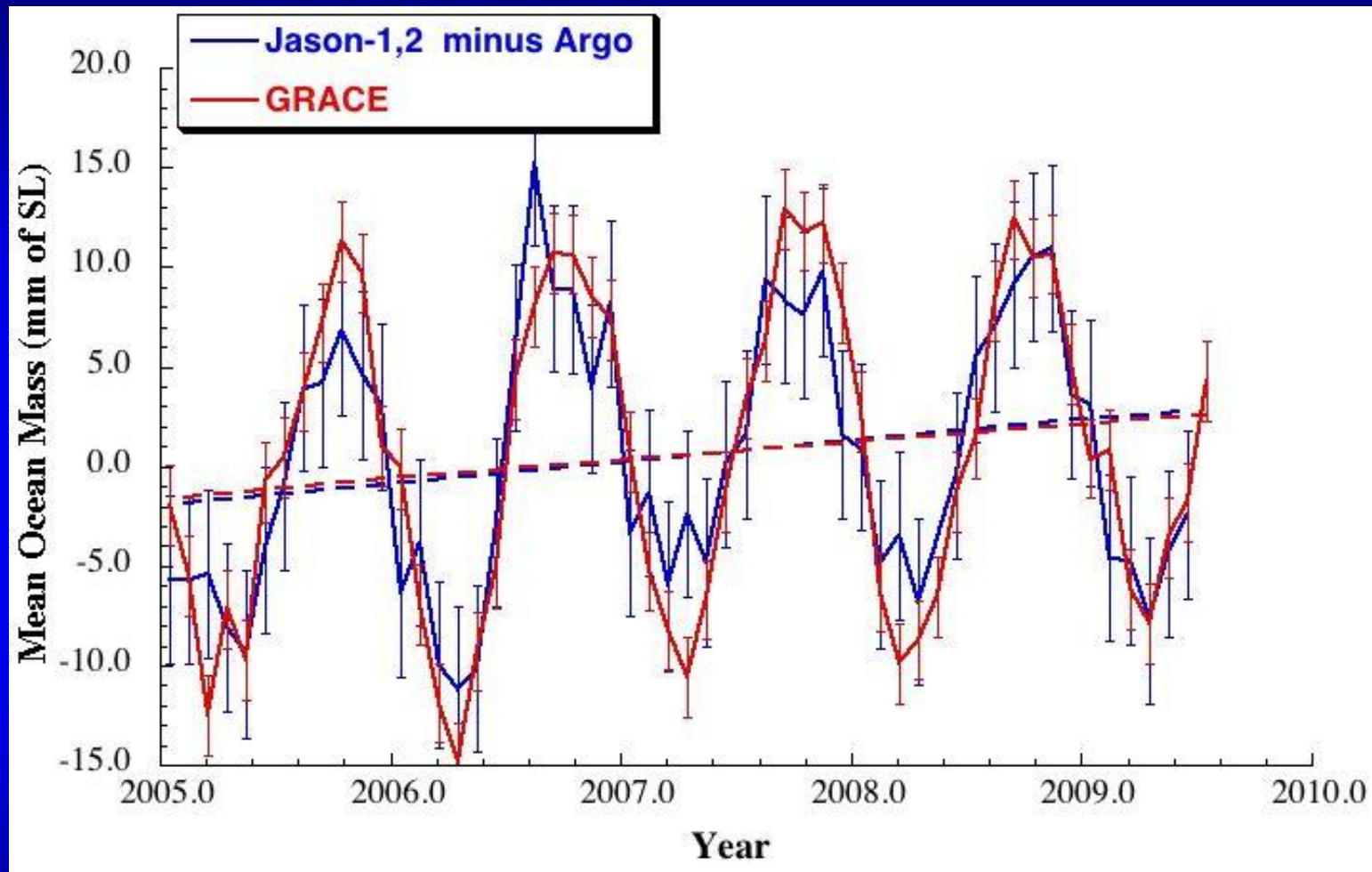


Measured by  
Altimeters

Most of this  
Difference  
is due to  
melting ice

Estimated change  
in volume due to thermal  
expansion  
(e.g. ARGO Floats)

# Sea Level Budget 2003 – 2009.5

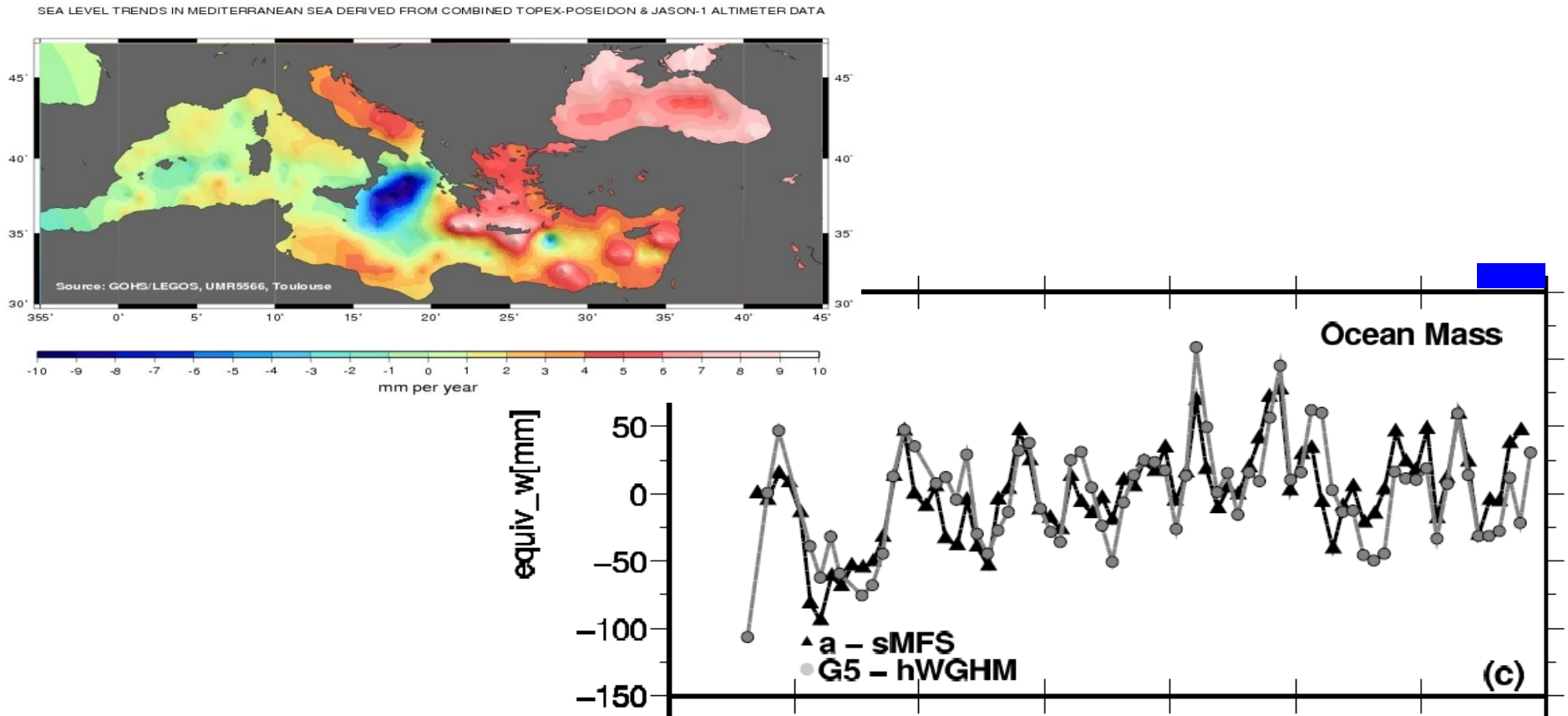


Grace Trend(2003-2009.5) =  $1.3 \pm 0.8$  mm/yr

# Recent results from the GRACE mission

## - ocean and sea level -

### GRACE enables separation of sea level contributions in Med Sea



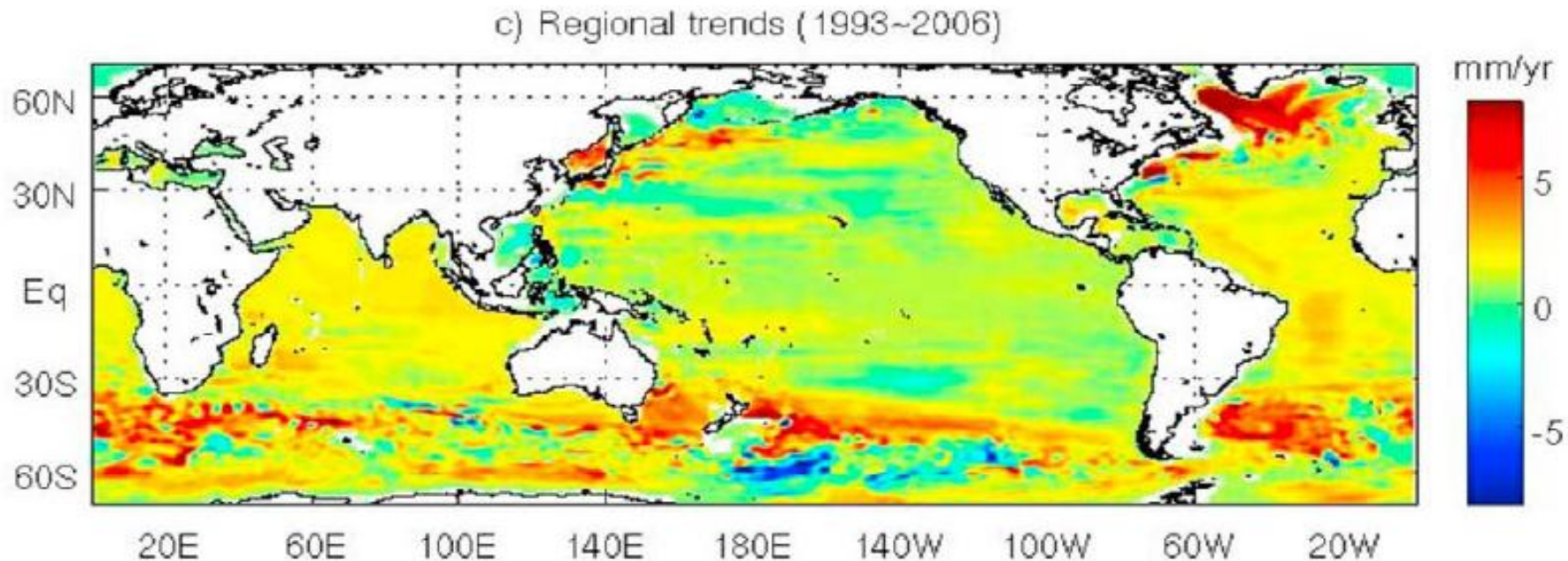
Fenoglio et al (submitted, J. Geodyn.):

- In 2002-2008, total altimetric sea level observed shows no significant trend (0.9 mm/y)
- GRACE reveals that mass is increasing by 6.3 mm/y, on top there is a warming
- Mass increase and warming is compensated by salinity increase (invisible for altimeter)

# Recent results from the GRACE mission

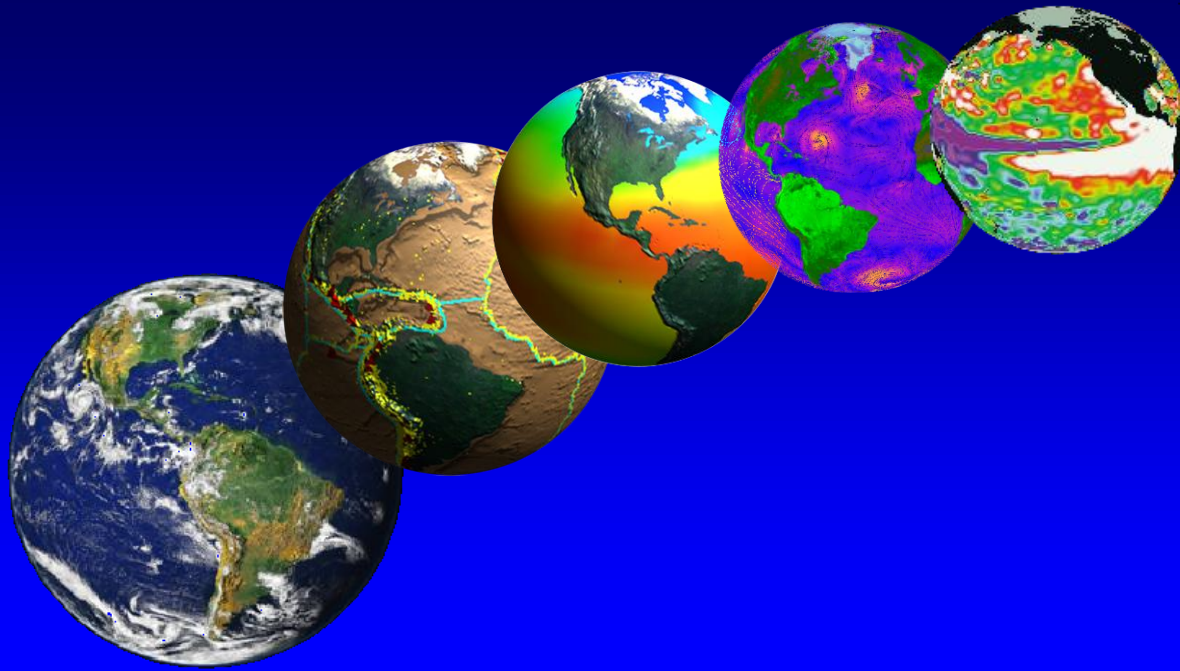
## - ocean and sea level -

Deep ocean warming cannot be assessed by in-situ observation



**Song and Colberg (2011)** combine GRACE, altimetry, observation of the warming of the upper 700m and modelling, to obtain 1.1mm/y contribution from deep ocean (=1/3 of observed sea level rise).



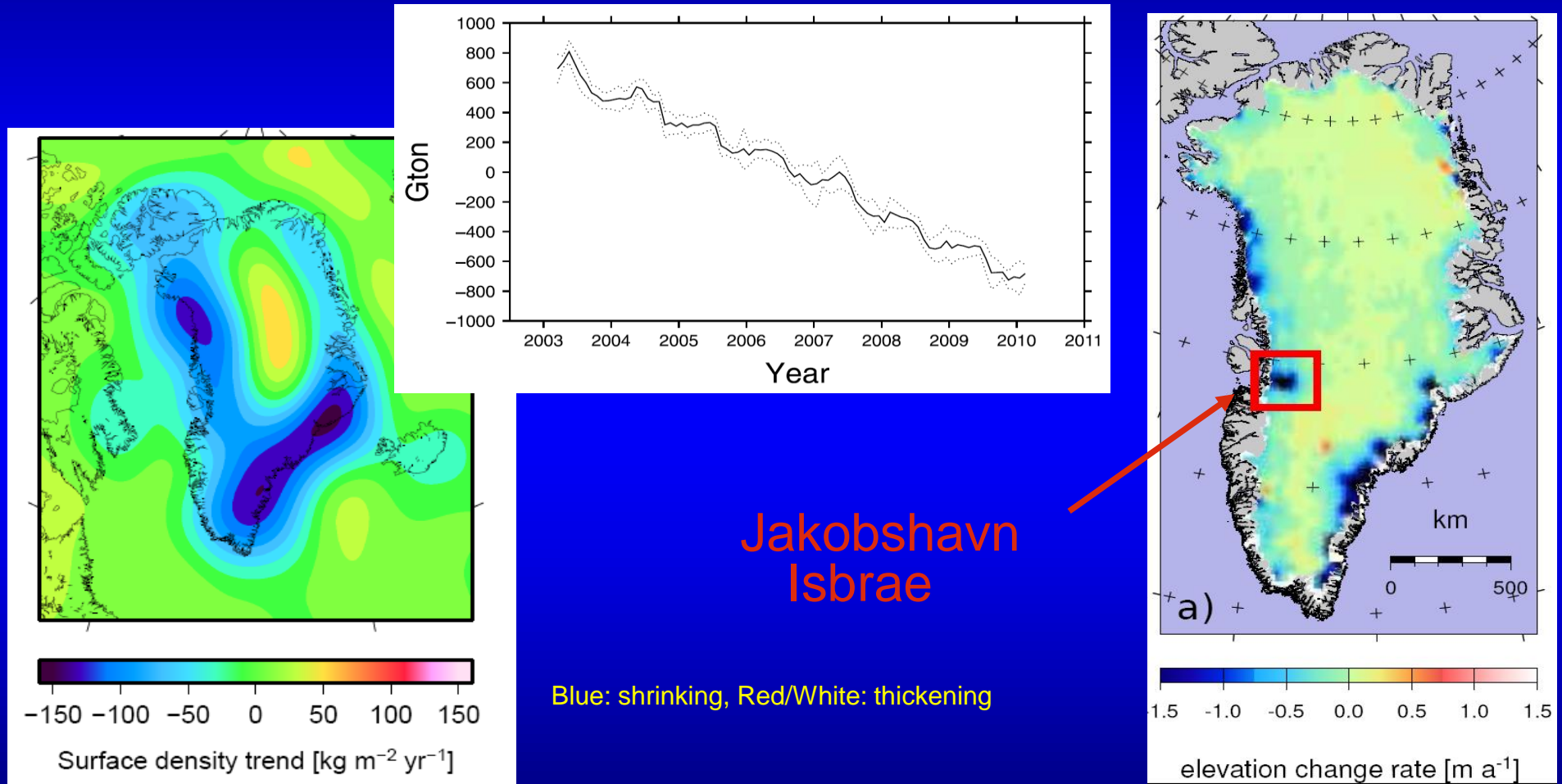


# GRACE and Cryosphere

# Recent results from the GRACE mission

## - cryosphere -

Greenland ice sheet shrinking from GRACE: estimates converge



Schrama and Wouters in JGR (2011, top) find a total rate of -201Gt/y from GRACE  
Ewert et al. (subm., J. Geodyn.): -191.2Gt/y from GRACE (left), -184.8Gt/y from ICESat.(right)

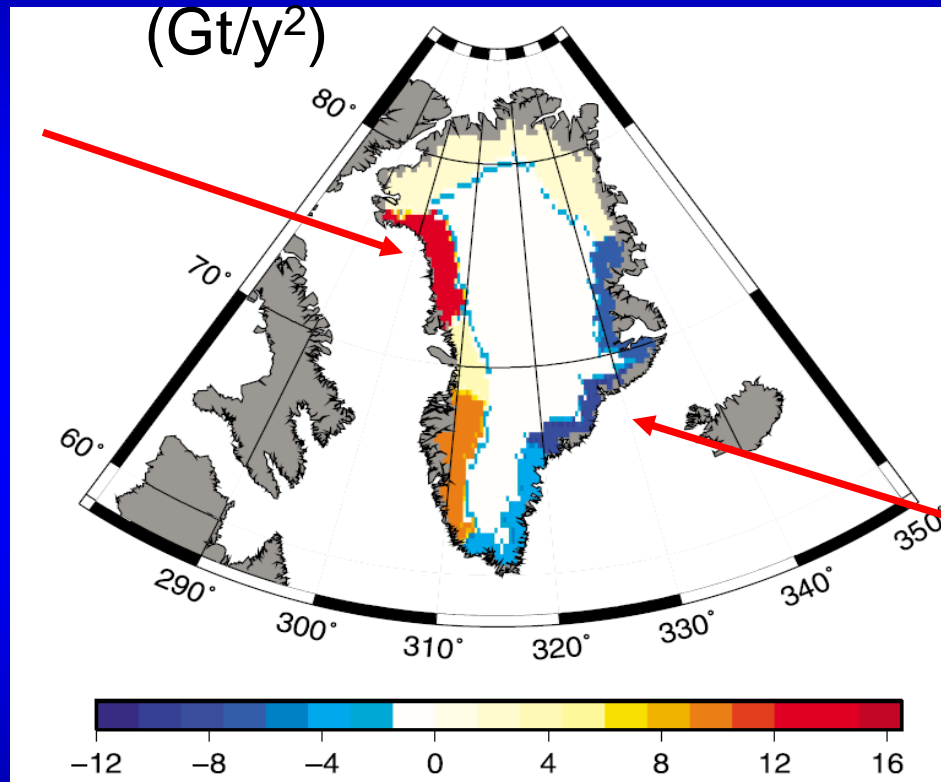
# Recent results from the GRACE mission

## - cryosphere -

Greenland ice sheet shrinking from GRACE: not uniform

Acceleration of mass loss

Shrinking at  
increasing speed



Shrinking at  
decreasing speed

Schrama and Wouters in JGR (2011) find a total acceleration of  $-8.3 \text{ Gt/y}^2$ , which is less than e.g. Velicogna (2009) ( $-30 \text{ Gt/y}^2$ ), Rignot et al. (2011) ( $-17 \text{ Gt/y}^2$ )

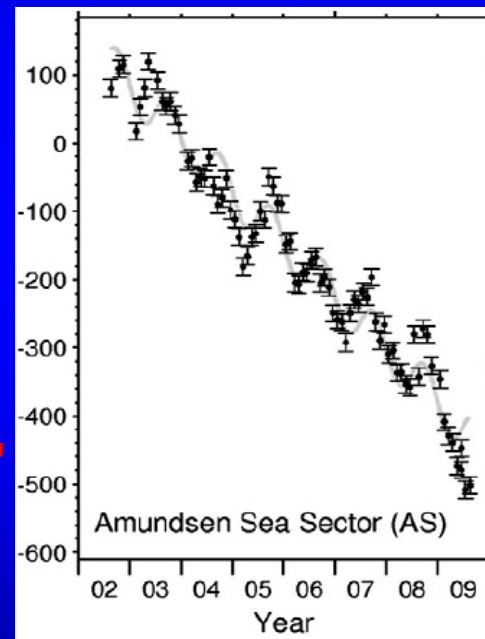


# Recent results from the GRACE mission

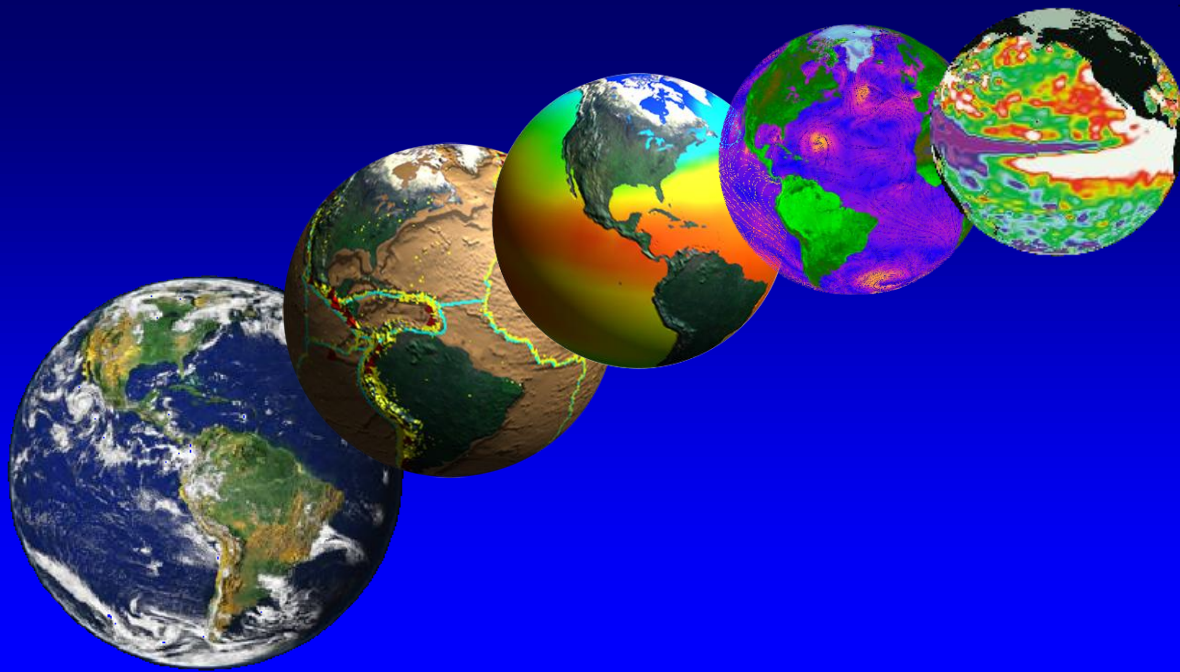
## - cryosphere -

GRACE measures interannual snow accumulation in West Antarctica

ECMWF, GRACE (trend/annual removed)



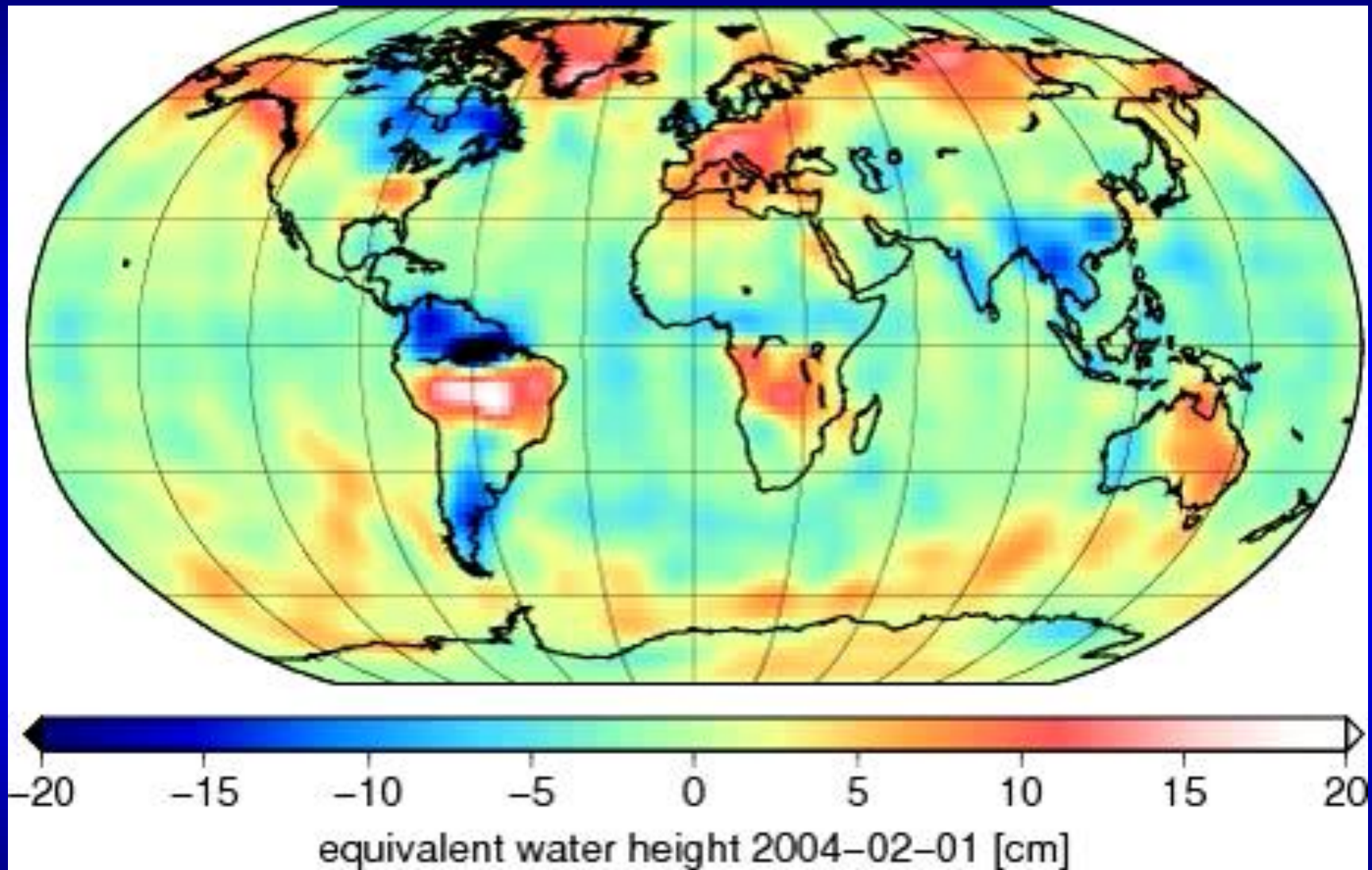
Interannual mass change in Antarctic Peninsula + Amundsen Sea Sector are governed by changes in precipitation rates. They clearly contain El-Nino Southern Oscillation signatures. **Sasgen et al. (2010, EPSL).**



# GRACE and Hydrology

# Recent results from the GRACE mission

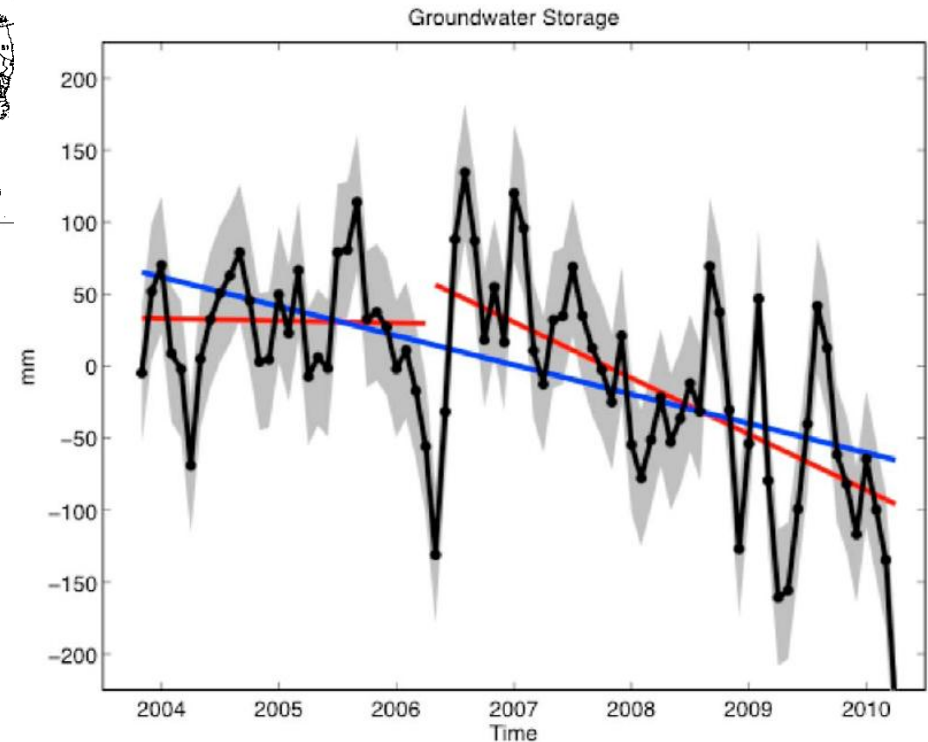
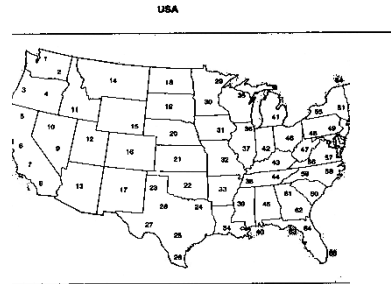
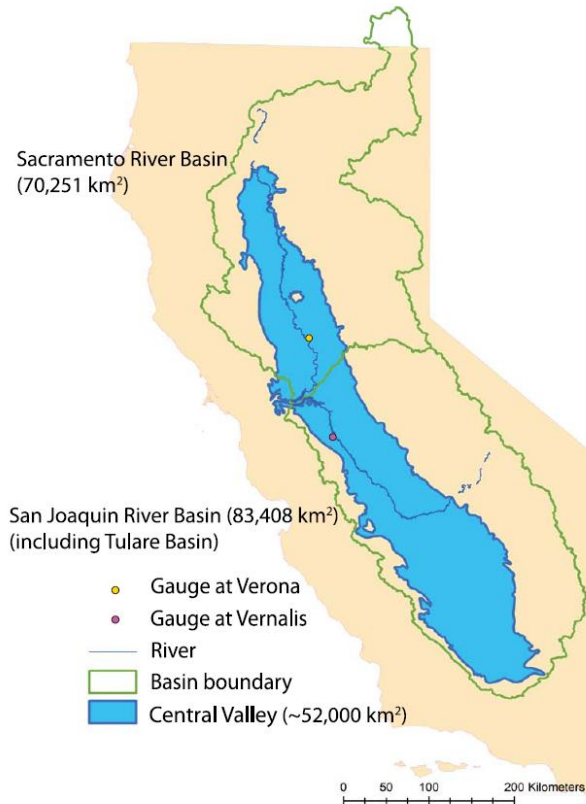
## - global mass transport -



Enrico Kurtenbach, Univ. of Bonn

# Recent results from the GRACE mission - changes in the terrestrial water cycle -

## GRACES measures changes of groundwater storage



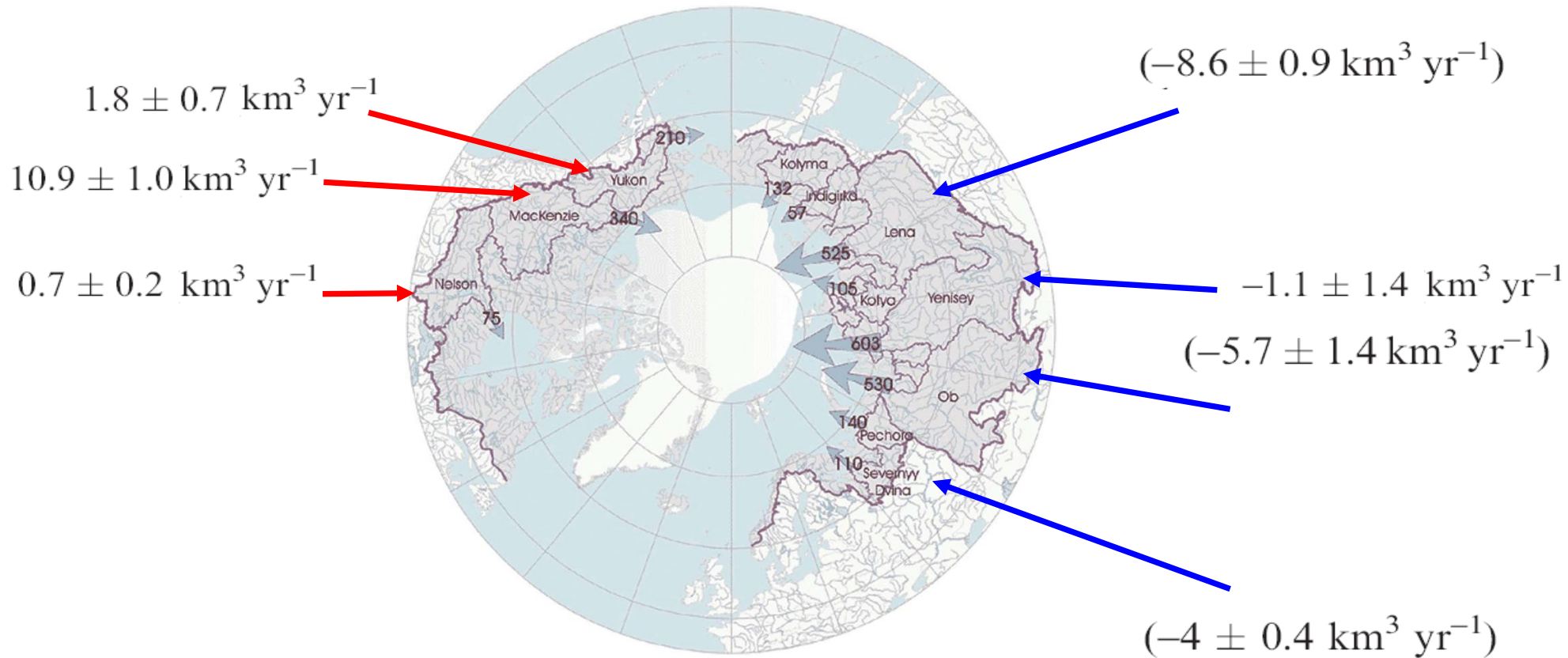
Groundwater depletion at a rate of 20.4mm/y in California central valley (52.000km<sup>2</sup>),

20.3km<sup>3</sup> within 6.5y study period, **Famiglietti et al. in GRL (2011)**



# Recent results from the GRACE mission - changes in the terrestrial water cycle -

Water balance of the Arctic drainage system – warming, snow and permafrost



**Frappart et al. in Int. J. Rem. Sens. (2011):**

- Eurasian basins loose snow mass, N-American basins gain mass.
- In accordance with large measured increase of European river discharge and small decrease of N-American river discharge.

# **The New Satellite Gravity Missions**

*have and will greatly contribute to*

- **Geodesy**
  - **Solid Earth Physics**
    - **Oceanography**
      - **Ice Physics**
        - **Meteorology**
          - **Climatology**

2000-2010	CHAMP
2002-2015	GRACE
2009-2012	GOCE

<b>2012</b>	<b>SWARM</b>
-------------	--------------

<b>2016</b>	<b>GRACE FO</b>
-------------	-----------------

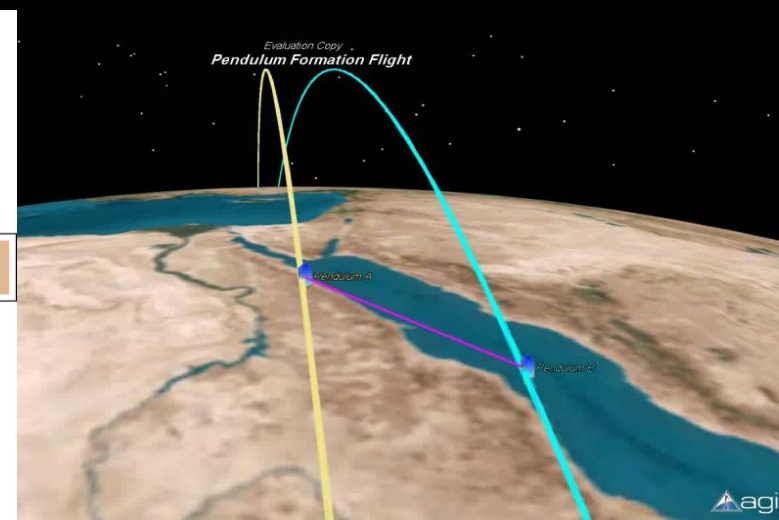
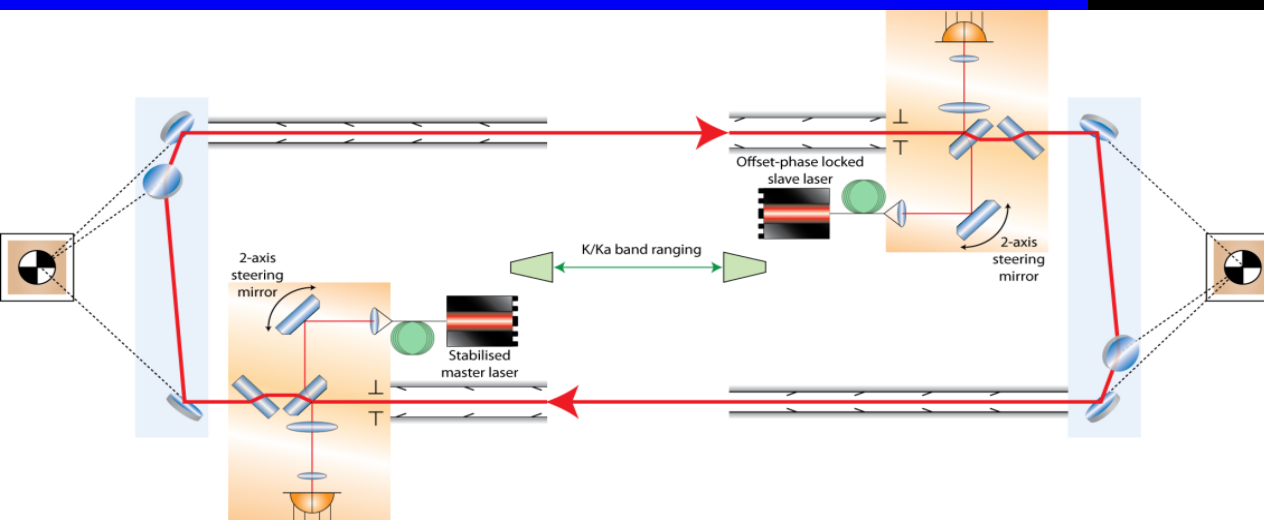
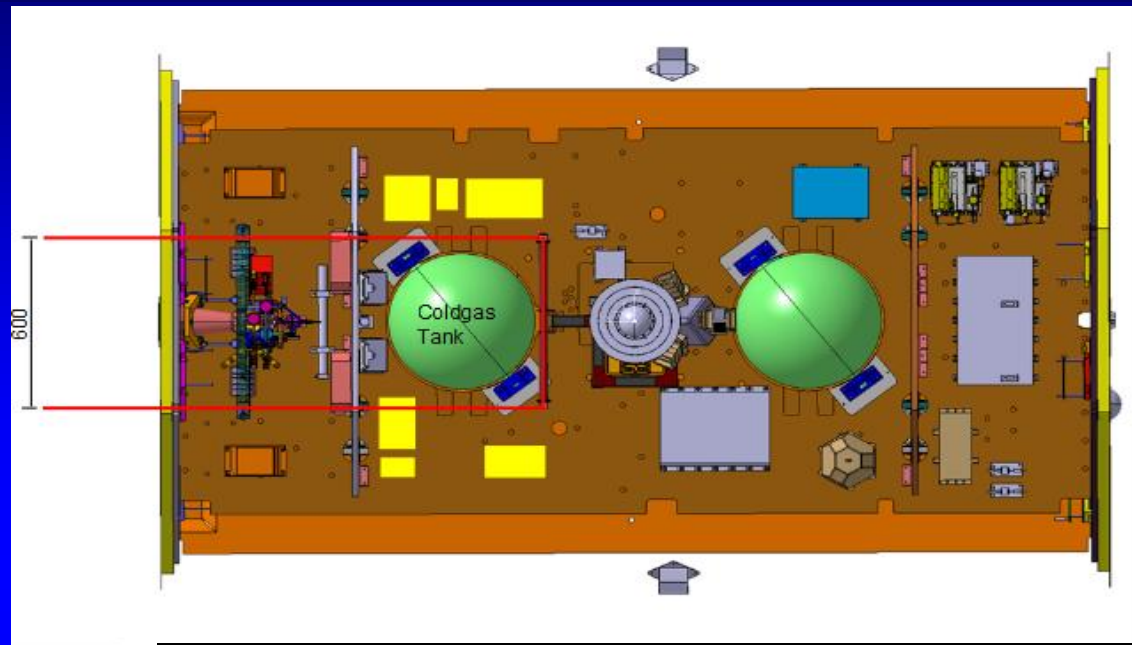
<b>2022</b>	<b>NG2/GRACE 2</b>
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## **Complementary Missions:**

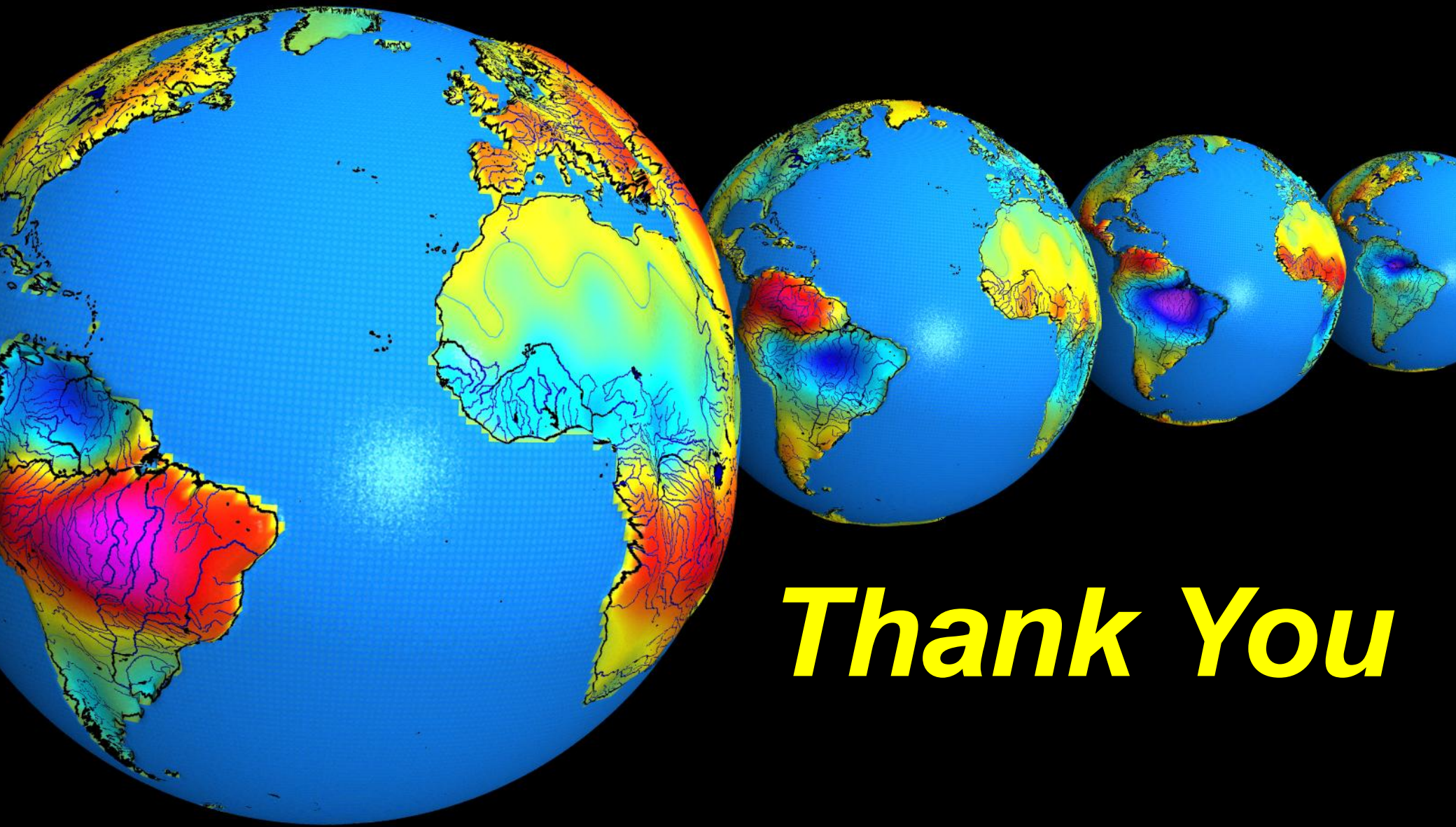
- ICESAT (2003) / CRYOSAT2 (2008), Tandem-X (2010)
- Altimetry Missions: Jason (2002), Envisat (2002)
- Aerogravimetry

# GRACE Follow-on

- Again US/D project
- Launch 2016, 7 years lifetime
- GRACE “Clone” with
  - Lessons learnt from GRACE
  - Add-on Laser Ranging Instrument (factor of 20 improved SST)
  - Improved orbit
- Funding not yet settled







***Thank You***